



The BDS/GNSS-Based Quality Control for ITS Applications

Dr. Rui Sun (Gloria) 2021-10-29



OUTLINE

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Background And Requirement

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BDS/GNSS-Based Quality Control Techniques

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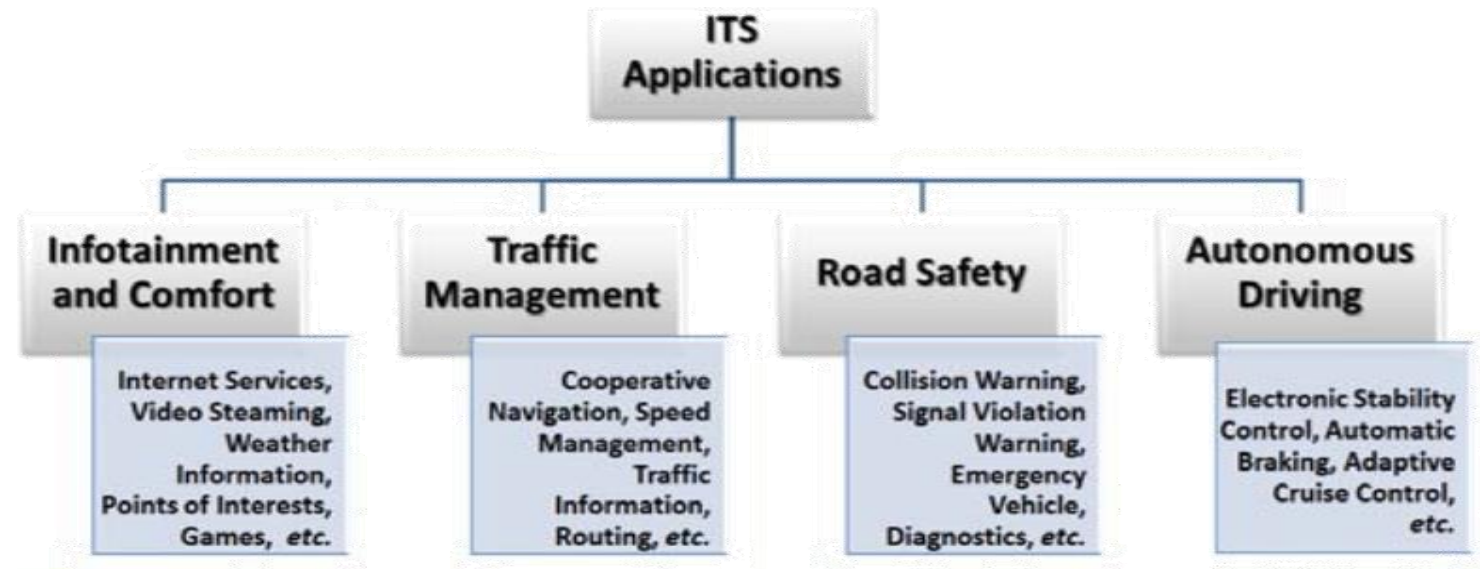
Quality Control Solutions

4

Conclusions

• Intelligent Transport Systems (ITS)

- *ITS is the integration of **information and communications** technology with transport infrastructure, vehicles and users*
- *ITS improves transport safety and mobility and enhances productivity through the use of advanced **information and communications** technologies.*



• Safety-Critical Applications

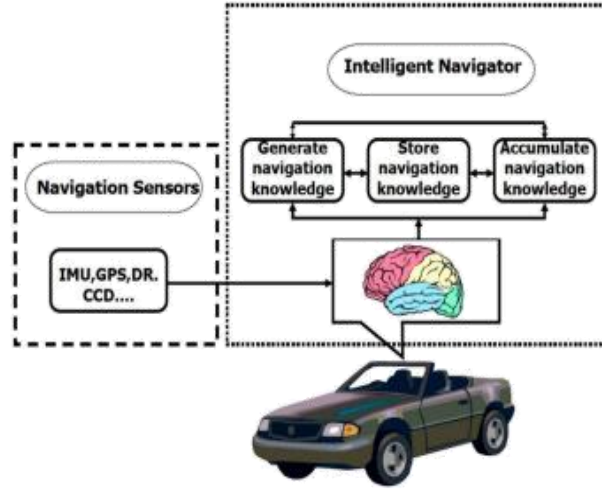
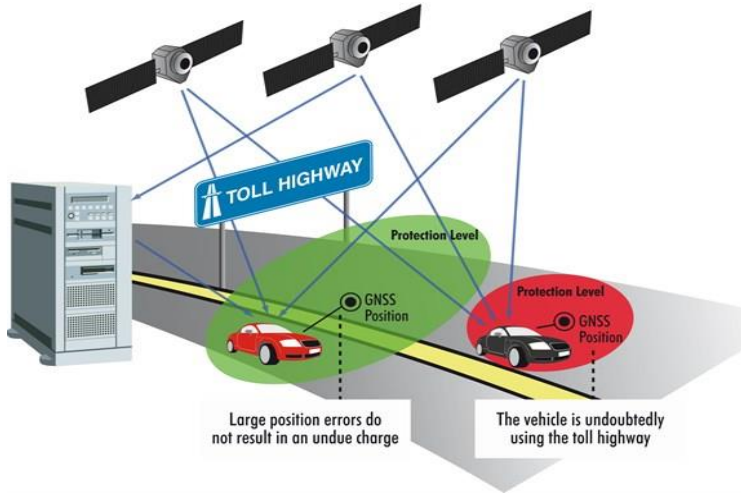
- Autonomous Driving
- ADAS (such as Intelligent speed adaptation)
- Hazardous Material Tracking

• Payment-Critical Applications

- Road User Charging (RUC)
- Pay-per-use services (PAYD, PPUI...)

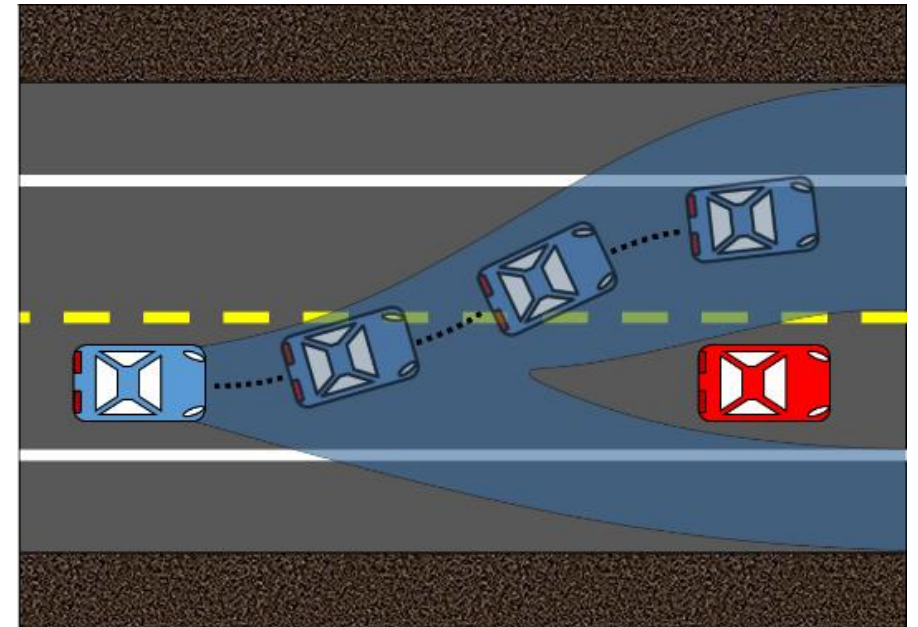
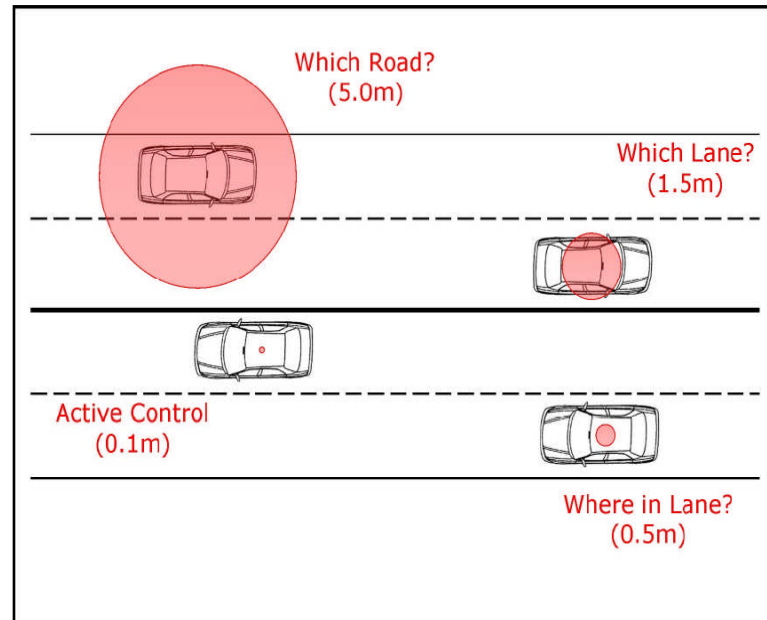
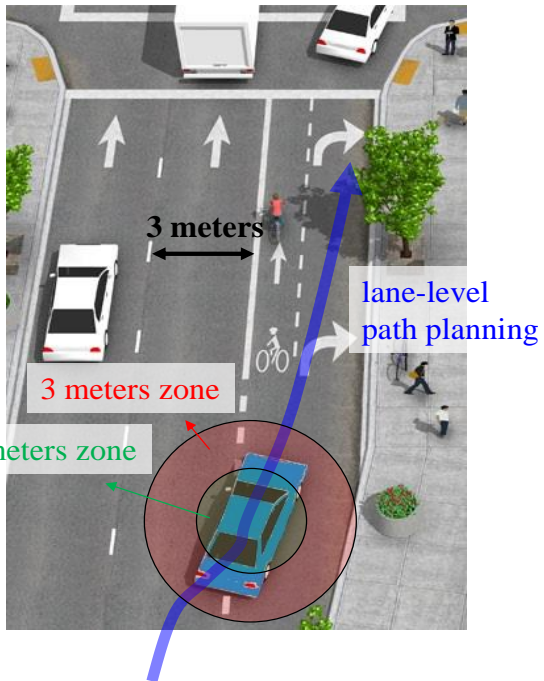
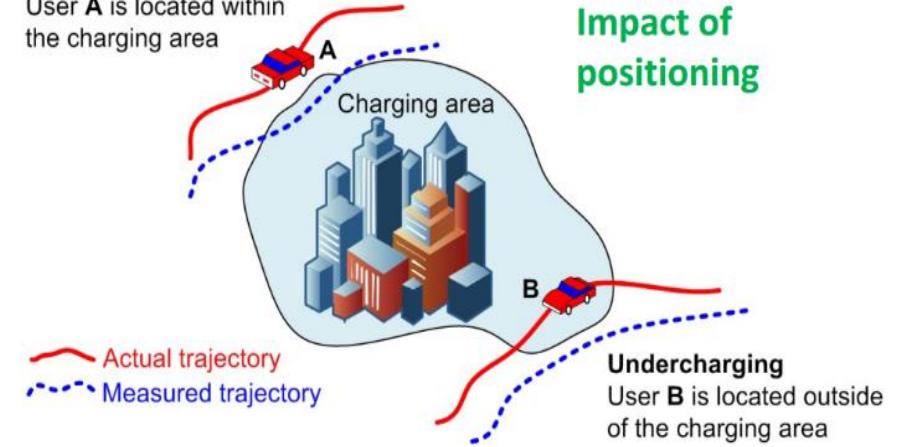
• Regulatory-Critical Applications

- Emergency services (eCall)
- Emergency vehicles navigation



Overcharging

User A is located within the charging area



• Performance Features

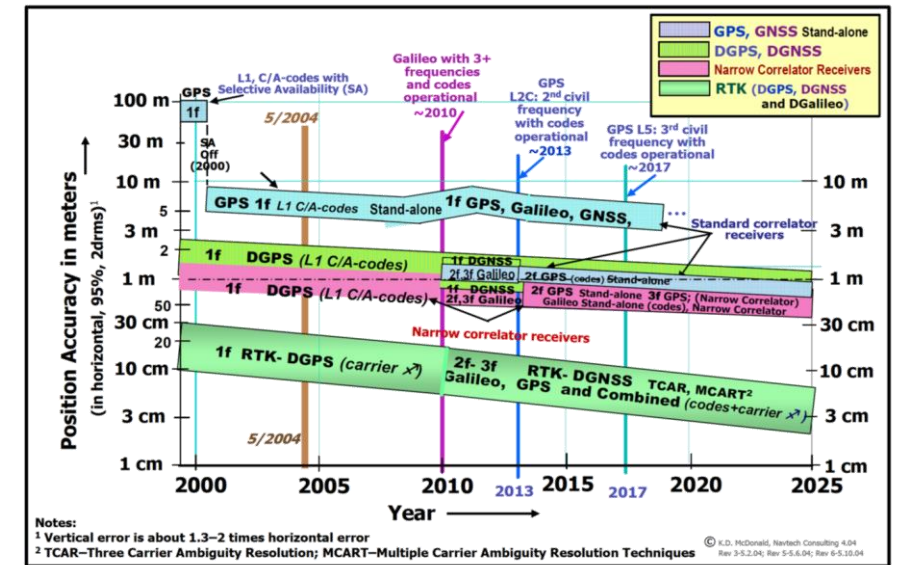
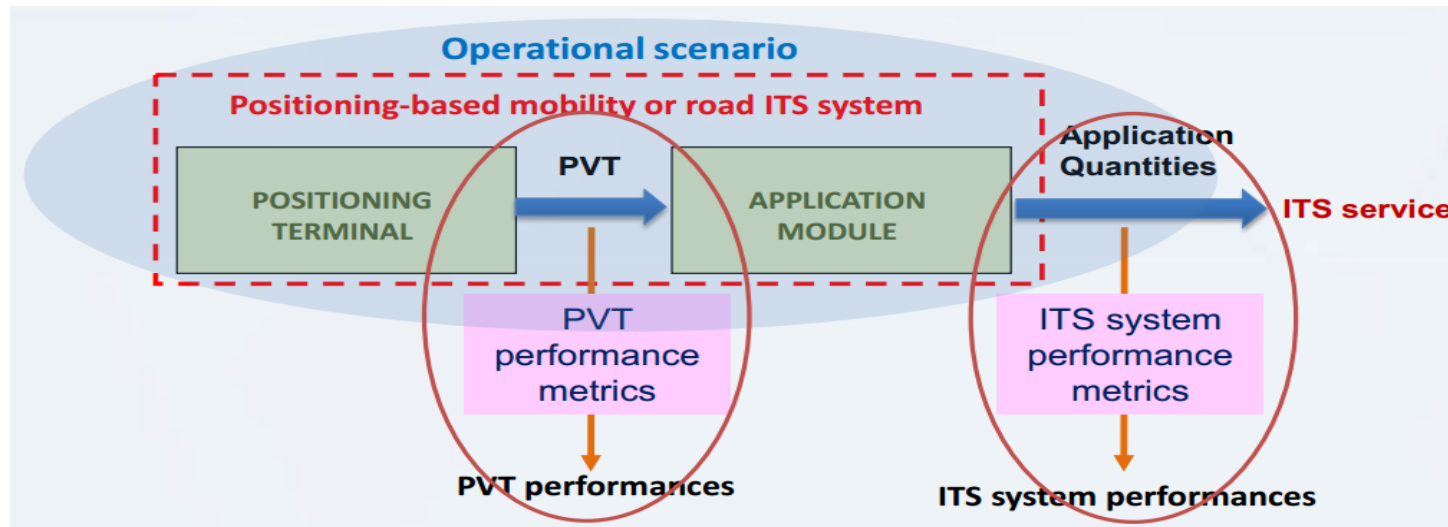
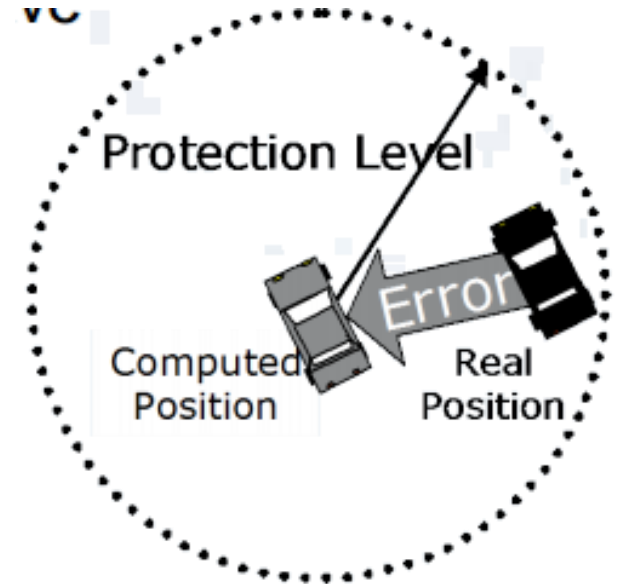
Accuracy: it refers to statistical figures of merit of position error, velocity error or speed error

Integrity: it refers to the level of **trust** a user can have in the value of a given component :

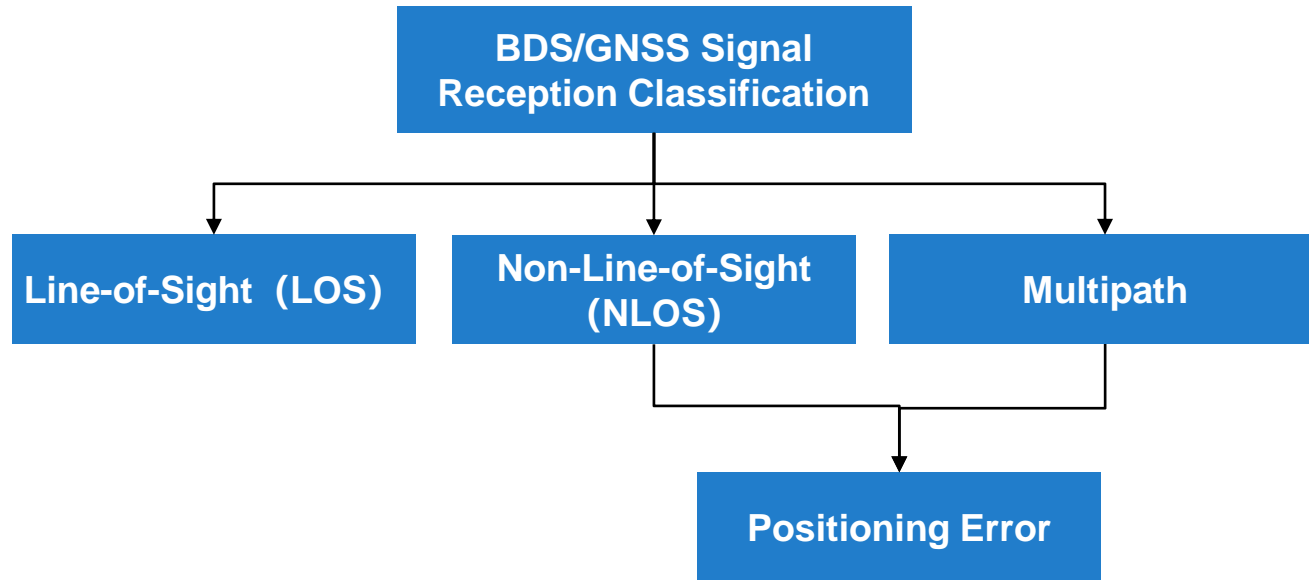
- in terms of reliability (*Integrity risk*)
- efficiency and usability (size of the *Protection level*).

Availability: generally speaking, it refers to the percentage of time during which the output of the positioning terminal is available.

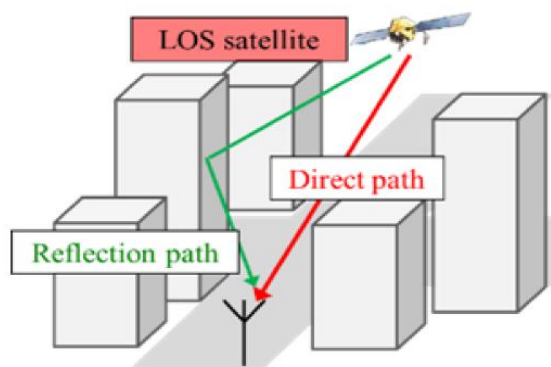
Timing performance: it refers to timestamp resolution, output latency, rate stability and Time To First Fix.



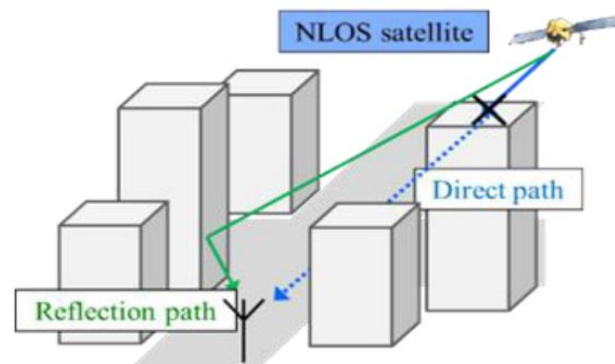
Notes:
¹ Vertical error is about 1.3–2 times horizontal error
² TCAR—Three Carrier Ambiguity Resolution; MCART—Multiple Carrier Ambiguity Resolution Techniques
 © I.D. McDonald, Navtech Consulting 4.04
 Rev 3.5.2.04; Rev 5.5.6.04; Rev 6.5.20.04



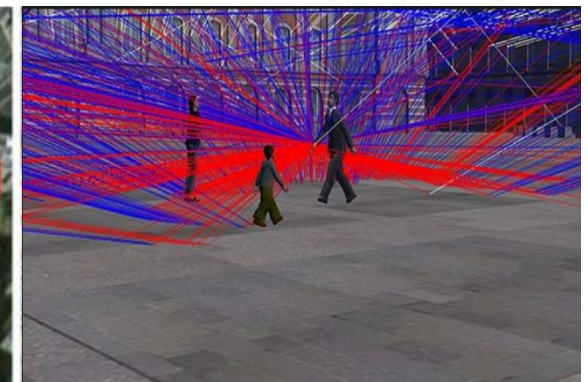
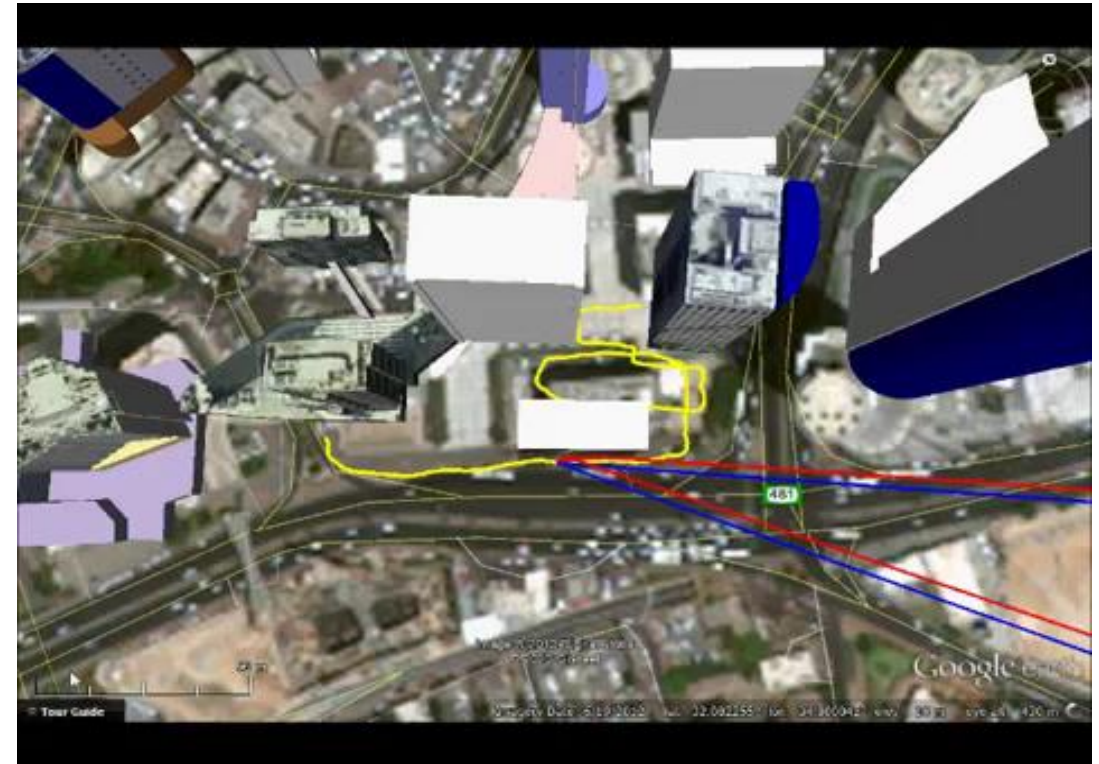
- Cannot be corrected by differential positioning method
- The error caused by NLOS can be tens of meters



(a) Multipath effect



(b) NLOS effect



• BDS/GNSS-Based Quality Control Techniques

Measurement weighting based techniques

satellite elevation angle

pseudorange rate

C/N₀ ...

Basic variables

Inertial Measurement Units (IMU)

camera

3D maps

...

Other sources

transferability issues to different environmental scenarios

Fault Detection and Exclusion based Techniques

range and position comparison methods

least squares residuals methods

parity space methods

maximum slope (MS) methods

...

widely used for RAIM

consistency checking-based methods

for ARAIM

based on the assumption of a single failure at a time

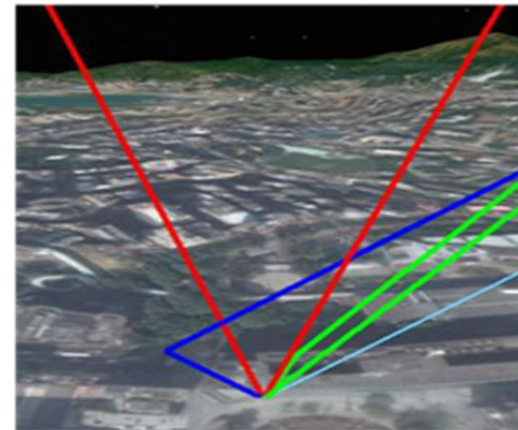
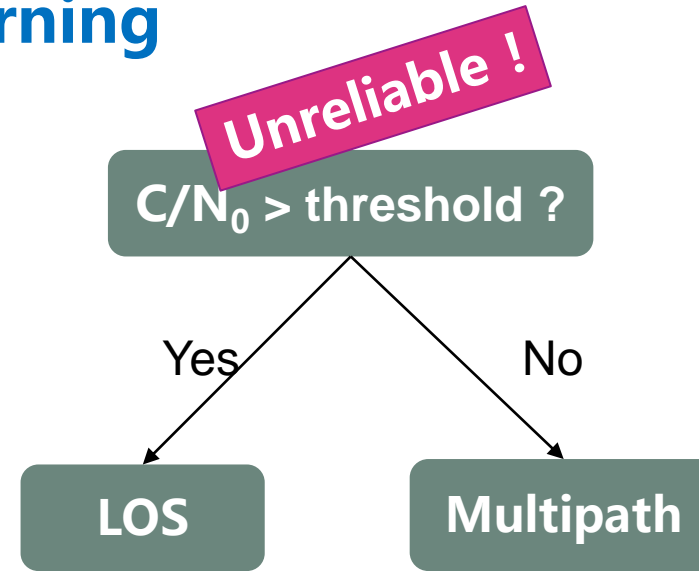
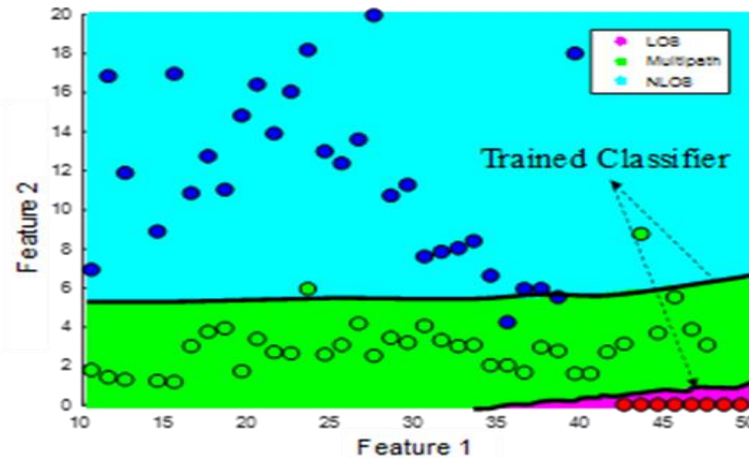
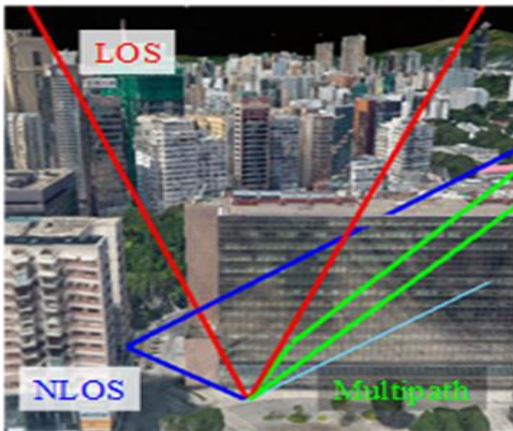
only effective when at least six satellites can be observed

high computational cost

Signal Reception Type Classification

9 Input Variables + PCA + machine learning

- ✓ Received Signal Strength (C/N_0)
- ✓ Temporal Difference of Received Signal Strength ($\Delta C/N_0$)
- ✓ Horizontal Dilution Of Precision (HDOP)
- ✓ Vertical Dilution Of Precision (VDOP)
- ✓ Satellite Elevation Angle (EA)
- ✓ Azimuth Angle (AA)
- ✓ Pseudorange residual (η)
- ✓ Consistency between delta pseudorange and pseudorange rate (ζ)
- ✓ Number of visible Satellites (NS)



Other Features

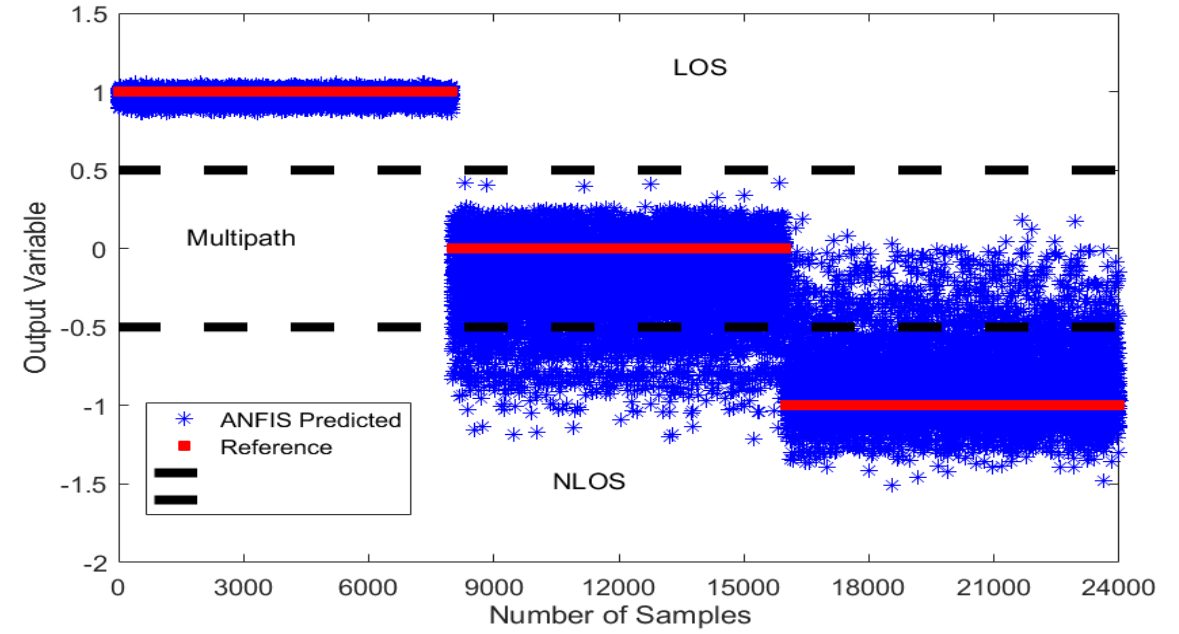
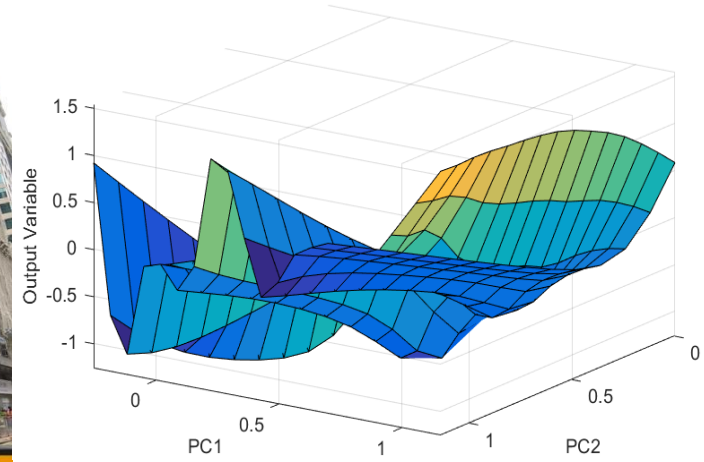
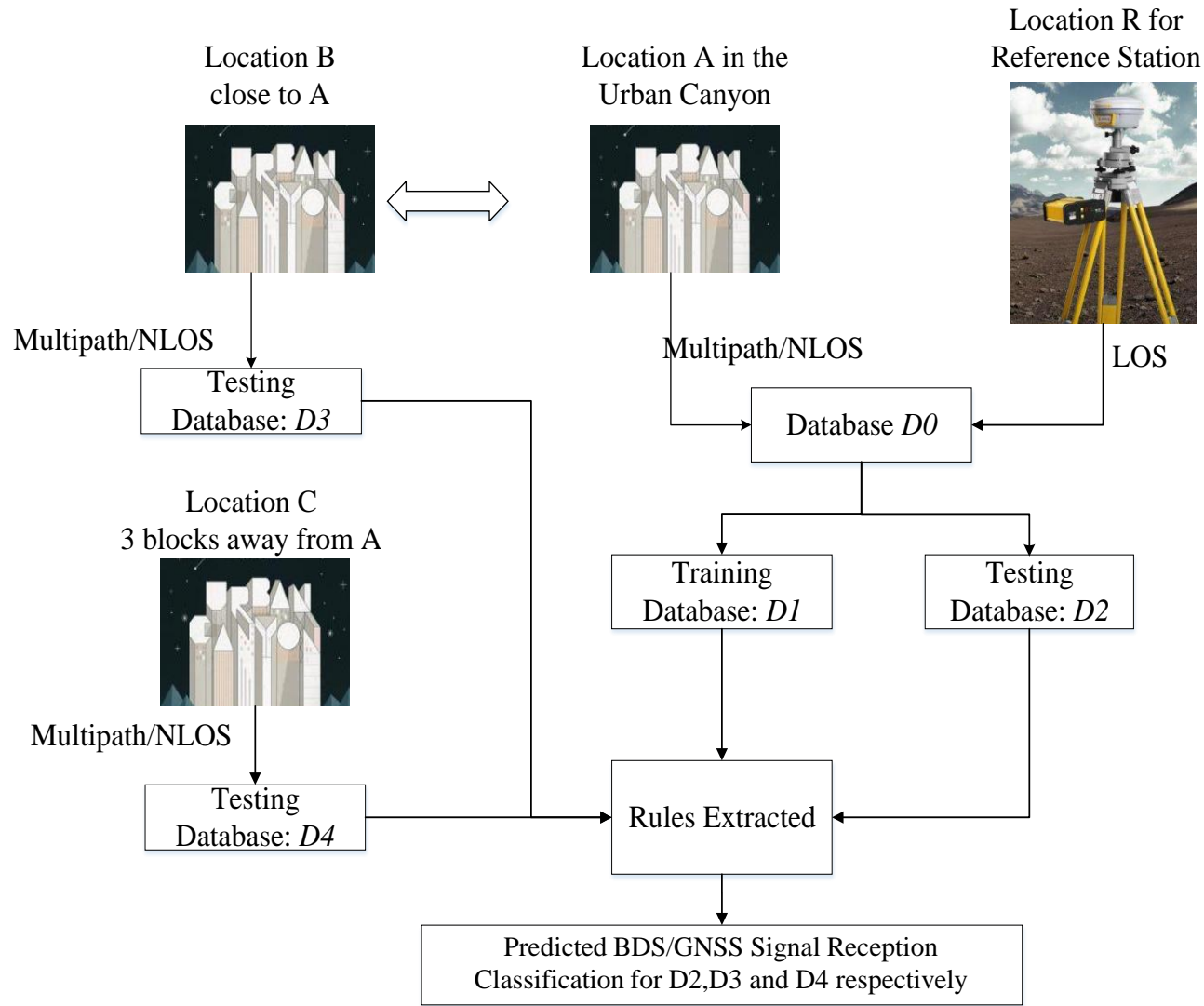
Satellite elevation angle

Pseudorange residual

Pseudorange rate

HDOP

VDOP



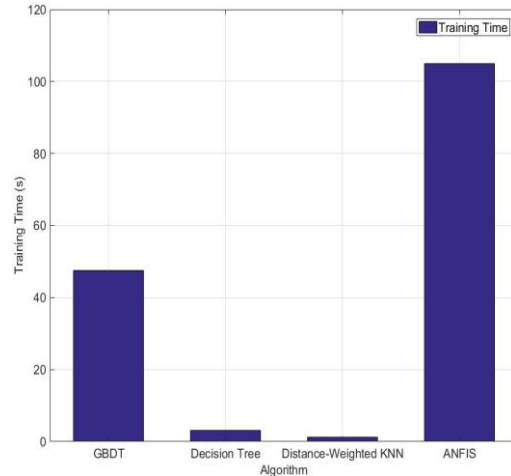
		Proposed ANFIS			Decision Tree			SVM			C/N ₀ based Classification		
		-1	0	1	-1	0	1	-1	0	1	-1	0	1
Label Result	-1	7291	709	0	3963	4036	1	5206	2793	1	13499	6603	708
	0	1269	6731	0	837	7163	0	1995	6005	0	3152	3339	1902
	1	0	0	8000	0	0	8000	0	0	8000	0	779	10018
Accuracy		91.76%			79.69%			80.05%			67.14%		
Accuracy for each class		-1	0	1	-1	0	1	-1	0	1	-1	0	1
		91.14%	84.14%	100%	49.54%	89.54%	100%	65.08%	75.06%	100%	64.87%	40%	93%

		Proposed ANFIS		Decision Tree		SVM		C/N ₀ based Classification	
Class Type		-1	0	-1	0	-1	0	-1	0
Dataset D3	Accuracy	72.98%		56.24%		75.43%		64.84%	
	Accuracy for each class	76.60%	66.35%	50.82%	66.19%	80.64%	66.19%	100%	0.31%
Dataset D4	Accuracy	71.51%		53.17%		71.20%		71.88%	
	Accuracy for each class	64.24%	91.48%	39.27%	91.38%	65.65%	86.46%	96.81%	3.82%

Classification + NLOS Elimination

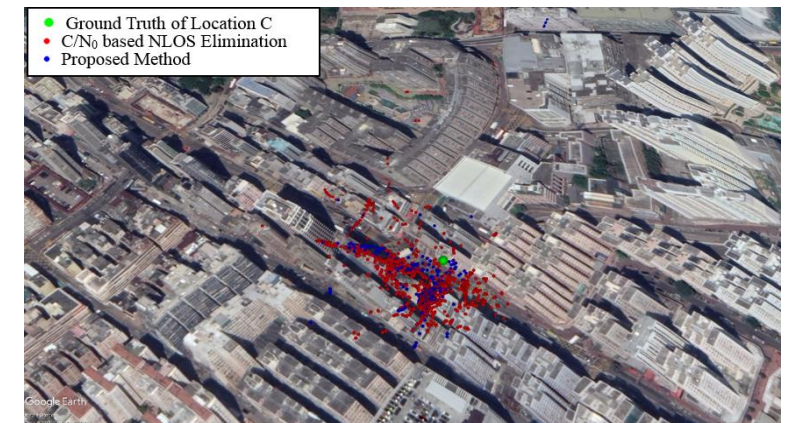
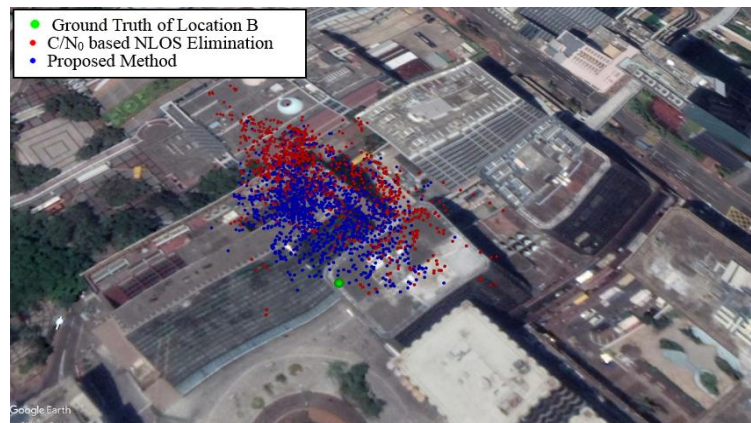
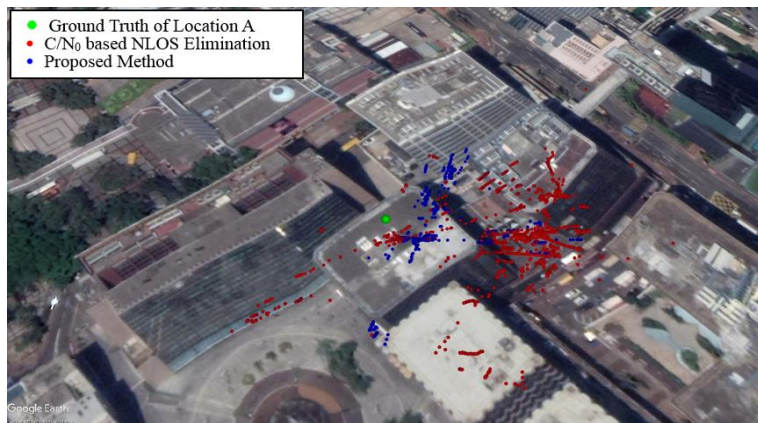
Input:

- ✓ C/N_0
- ✓ Pseudorange residuals
- ✓ Satellite elevation angle



	RMSE (m)	E	N	U	3D	horizontal
Location A	C/ N_0 Based NLOS Elimination	40.92	17.9	79.01	90.76	44.67
	GBDT with Multi-Feature-Based NLOS Elimination	26.19	17.02	51.02	59.82	31.23
	Improvement (%)	36.0	4.9	35.4	34.1	30.1
Location B	C/ N_0 Based NLOS Elimination	20.13	45.41	63.72	80.80	49.67
	GBDT with Multi-Feature-Based NLOS Elimination	18.35	35.61	50.89	64.77	40.06
	Improvement (%)	8.8	21.6	20.1	19.8	19.4
Location C	C/ N_0 Based NLOS Elimination	25.4	29.5	127.67	133.37	38.59
	GBDT with Multi-Feature-Based NLOS Elimination	25.07	32.27	123.83	130.39	40.86
	Improvement (%)	1.3	-9.4	3.0	2.2	-5.9

Classification Accuracy: 89%, 77.2%, 55.3%



Using Gradient Boosting Decision Tree (GBDT) to Fit the Pseudorange Error

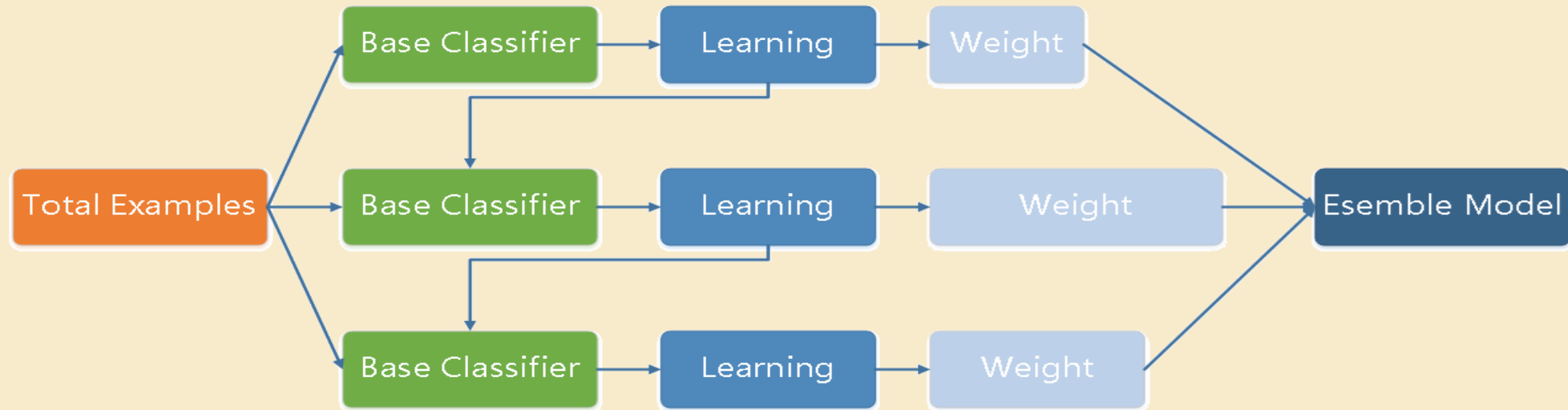
Carrier-to-Noise Ratio: C/N_0 (dB-Hz)

Pseudorange residual: The difference between the observed pseudorange measurements and the pseudorange values deduced from the positioning solution.

Pseudorange Error

Satellite Elevation Angle: The angle between the satellite-to-receiver line and the horizon

- One of the best algorithms for fitting the true distribution
- Data classification is achieved by using an **additive model** to continuously reduce the residuals generated during the training process
- each iteration produces a weak classifier, and each classifier is trained based on the **residual error** of the previous classifiers.



Pseudorange preprocessing

Pseudorange after model correction :

$$\rho^c = R + c(\delta t^r - \delta t^{sv}) + I + T + \varepsilon - c(\delta t_c^r - \delta t_c^{sv}) - I_K - T_S$$

$$\rho^c = R + c(\Delta \delta t^r - \Delta \delta t^{sv}) + \Delta I + \Delta T + \varepsilon$$

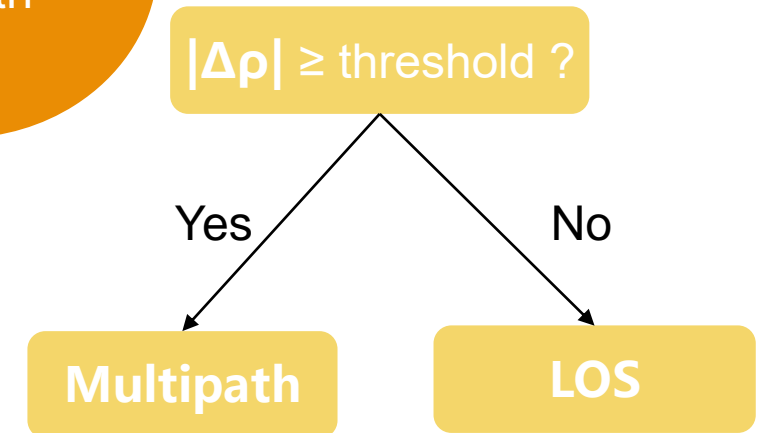
The pseudorange error $\Delta \rho$ is calculated by the following formula:

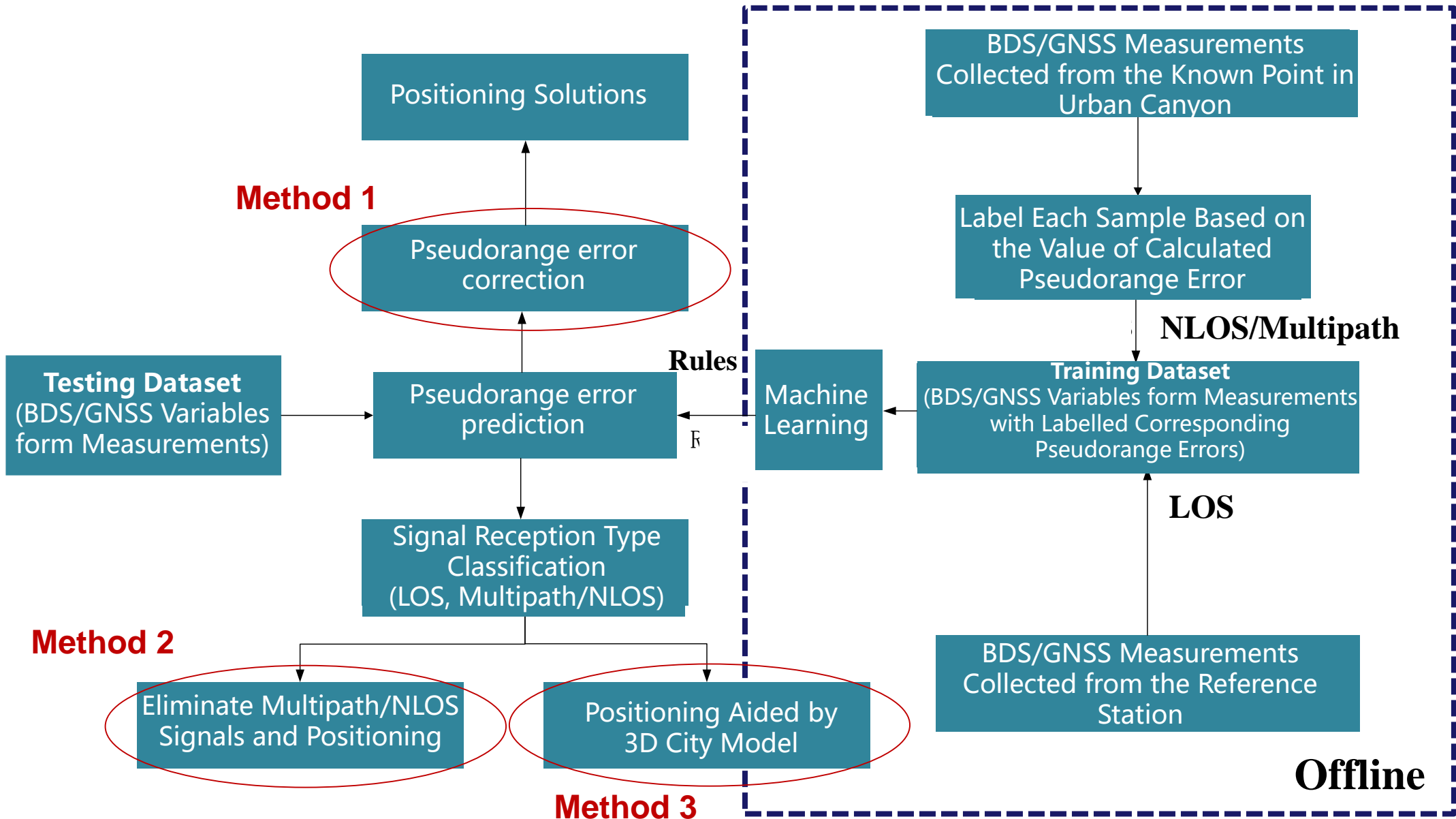
$$\Delta \rho = \rho^c - R = c(\Delta \delta t^r - \Delta \delta t^{sv}) + \Delta I + \Delta T + \varepsilon$$

Pseudorange error $\Delta \rho$ includes the errors caused by clock errors, ionospheric delay, troposphere delay, and multipath effects that have not been fully corrected (ε).

- NLOS : Positive pseudorange error
- Multipath : Positive or negative pseudorange error

The main part is multipath error



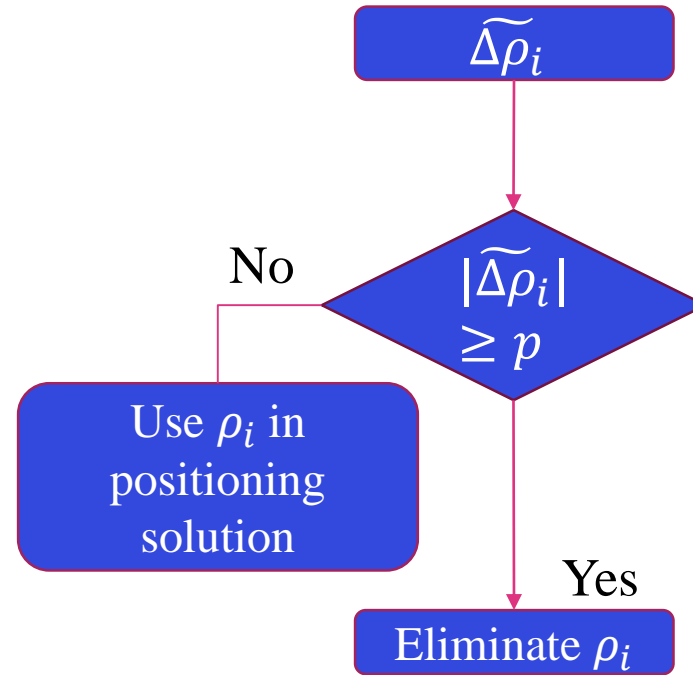


1. Positioning Method Based on Pseudorange Correction

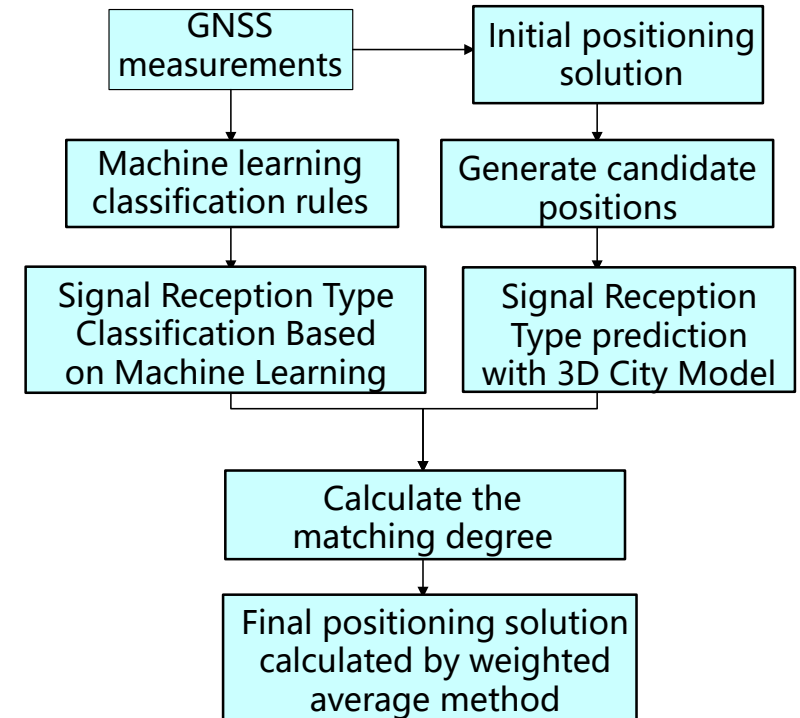
$$\begin{cases} \rho_1^c = \rho_1 - \widetilde{\Delta\rho}_1 \\ \rho_2^c = \rho_2 - \widetilde{\Delta\rho}_2 \\ \vdots \\ \rho_i^c = \rho_i - \widetilde{\Delta\rho}_i \end{cases}$$

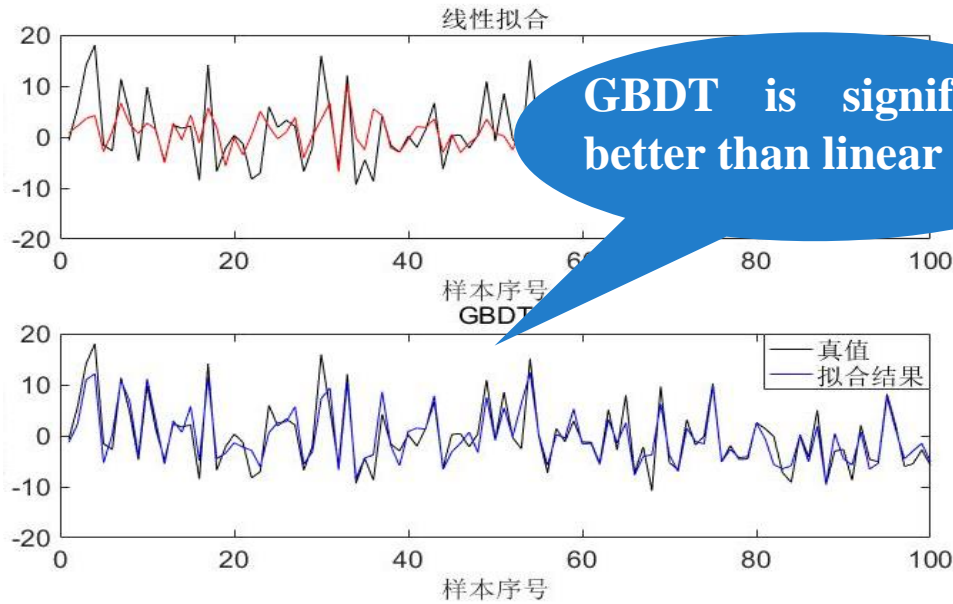
Positioning Solution
(x^r, y^r, z^r)

2. Positioning Method Based on Multipath Signal Elimination



3. Positioning Method Aided by 3D City Model



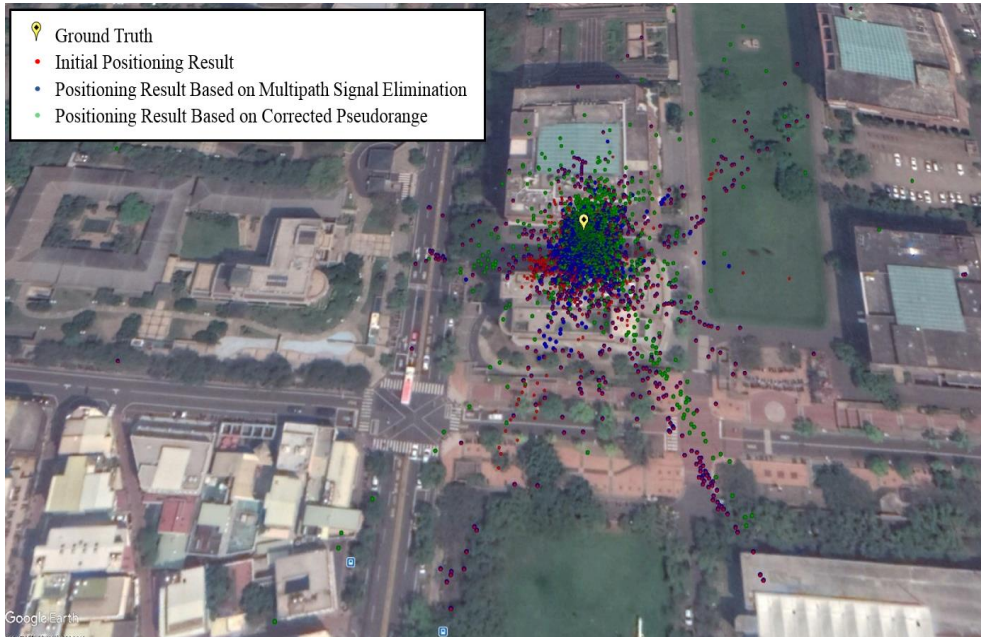


Test Case 1: Narrow road with buildings on both sides

Reference Station: 7 hours of data

Urban Canyon: 1 Hz, NovAtel Propak 7

Training Dataset	Urban Canyon		Reference Station
Signal Reception Type	Multipath	LOS	LOS
Number of samples	48000	24000	24000
Testing Dataset	Urban Canyon		
Signal Reception Type	Multipath		LOS
Number of samples	46759		50285



Classification Accuracy	Total	LOS	Multipath
	75.90%	80.29%	71.49%

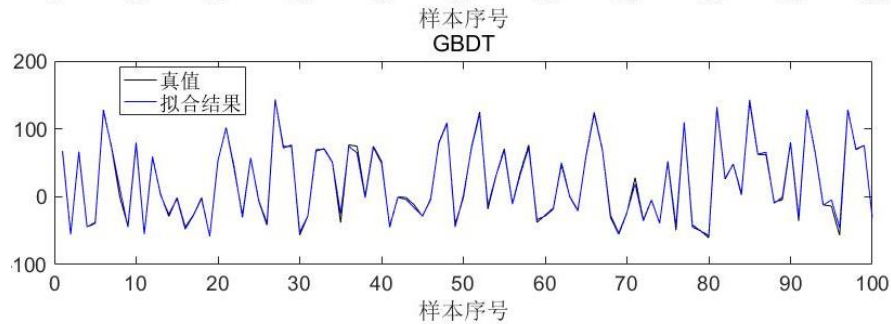
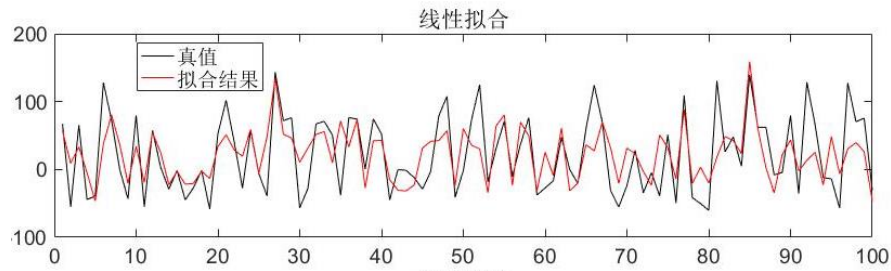
Positioning Accuracy	E	N	U	3D	2D
	Initial Positioning Result (RMSE/m)	18.02	25.85	44.36	54.41
Positioning Result Based on Multipath Signal Elimination (RMSE/m)	13.74	13.80	32.46	37.86	19.48
Improvement (%)	23.75	46.62	26.83	30.42	38.18
Positioning Result Based on Corrected Pseudorange (RMSE/m)	14.82	21.49	36.27	44.69	26.11
Improvement (%)	17.76	16.87	18.24	17.86	17.14

Test Case 2: High rise building on one side

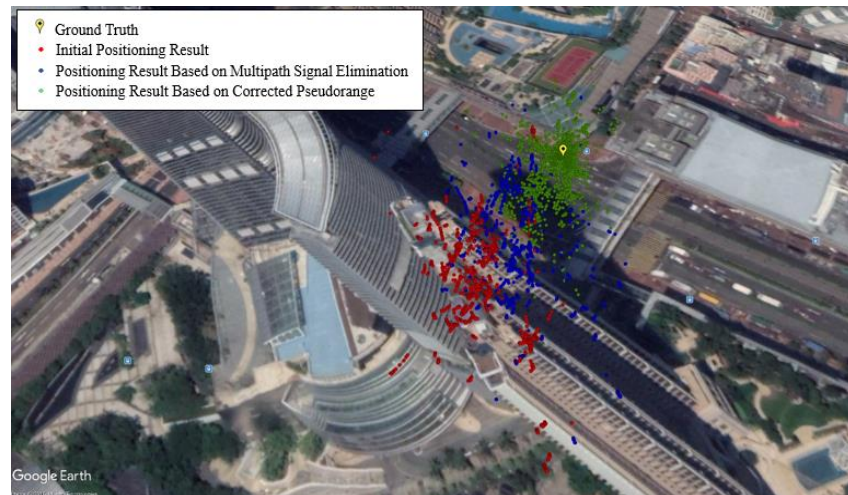
Reference Station: 4 hours of data with a sampling interval of 5s

Urban Canyon: P1、 P2

Receiver : NovAtel OEM 6



	linear fitting	GBDT fitting
RMSE/m	38.14	1.24



Training Dataset	Urban Canyon				Reference Station
	P1		P2		
Signal Reception Type	Multipath	LOS	Multipath	LOS	LOS
Number of samples	16000	24000	16000	24000	16000
Testing Dataset	Urban Canyon (P1)				
Signal Reception Type	Multipath				LOS
Number of samples	4686				14869

Classification Accuracy	Total	LOS	Multipath
	91.13%	96.42%	74.49%

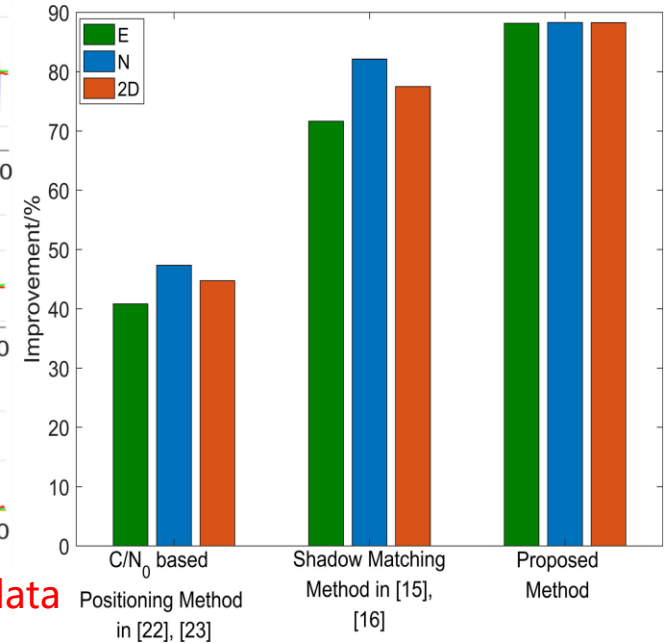
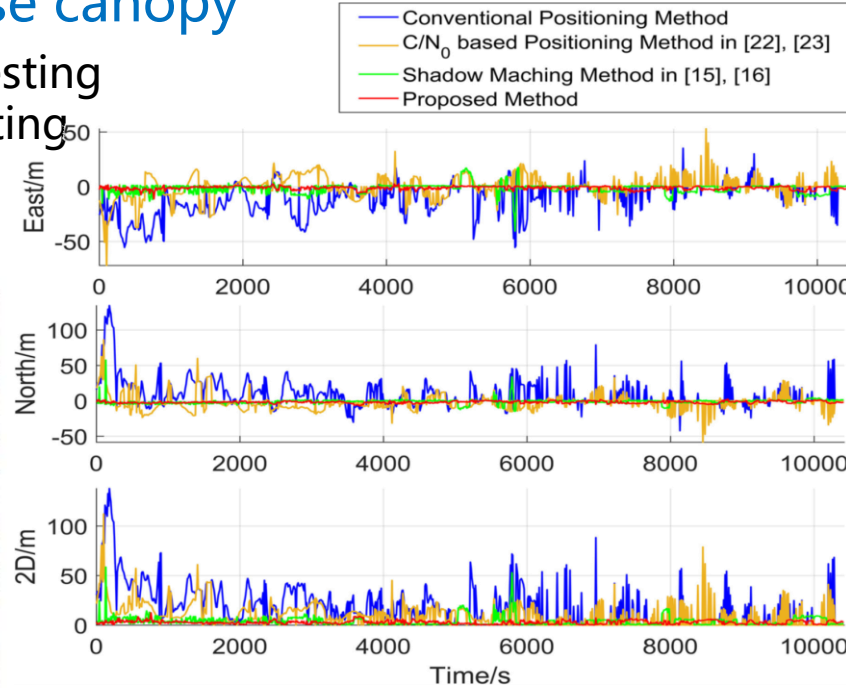
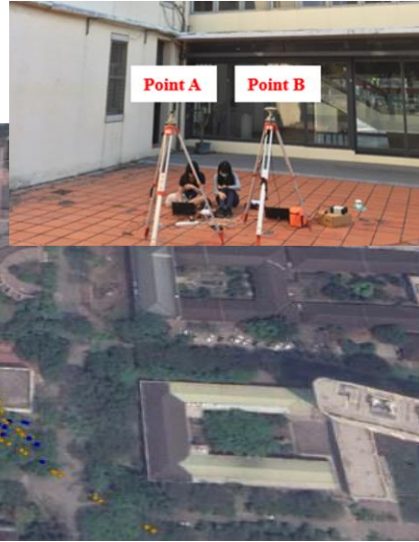
Positioning Accuracy	E	N	U	3D	2D
	Initial Positioning Result (RMSE/m)	36.87	55.16	39.39	77.16
Positioning Result Based on Multipath Signal Elimination (RMSE/m)	10.30	14.56	16.00	23.96	17.83
Improvement (%)	72.06	73.60	59.38	68.95	73.12
Positioning Result Based on Corrected Pseudorange (RMSE/m)	24.73	40.21	30.08	61.29	47.21
Improvement (%)	32.93	27.10	23.64	20.57	28.84

Test Case 3: L-shaped corner + Dense canopy

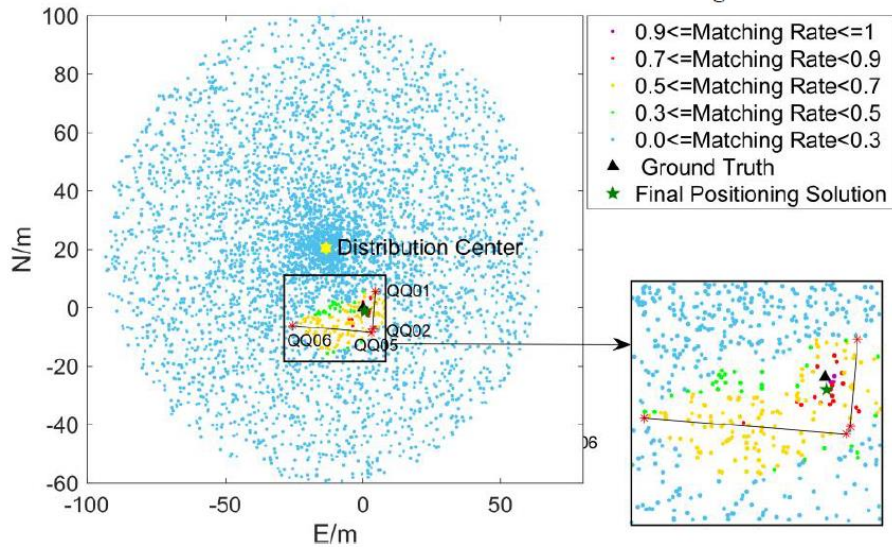
PA: UBLOX M8T, 5 Hz, 30 mins, training and testing

PB: Novatel Propak OEM-6, 1 Hz, 30 mins, testing

- 📍 Ground Truth
- Conventional Positioning Method
- C/N₀ based Positioning Method in [22], [23]
- Shadow Matching Method in [15], [16]
- Proposed Method



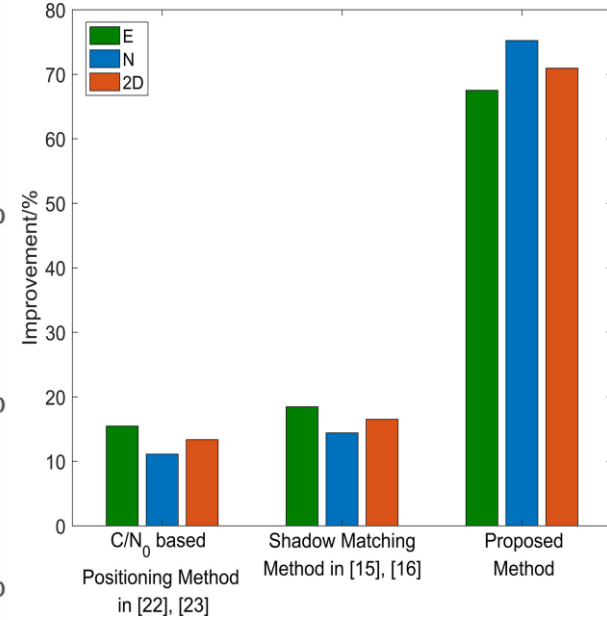
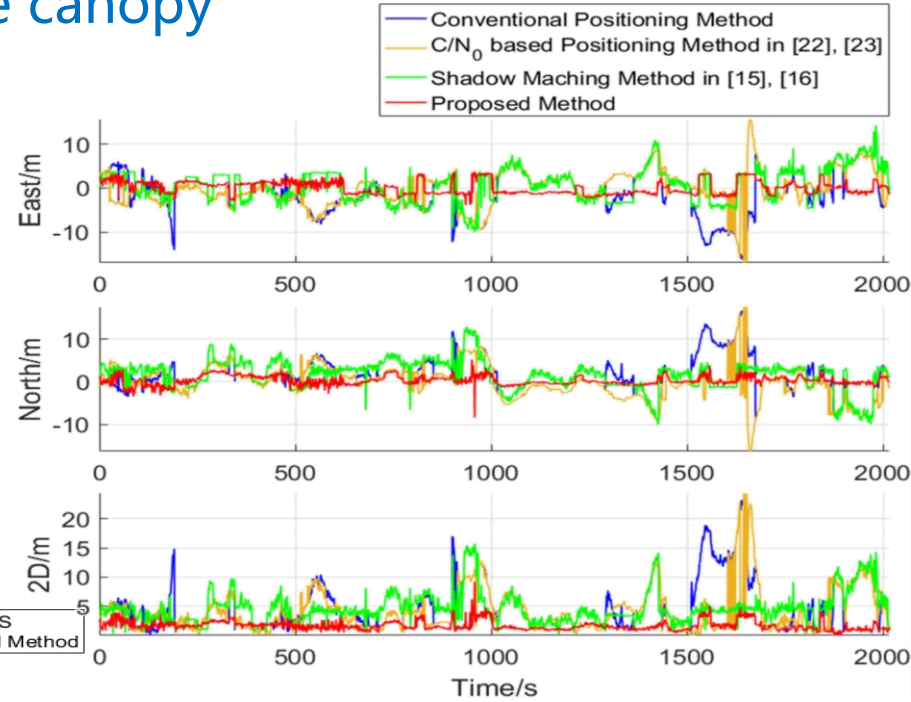
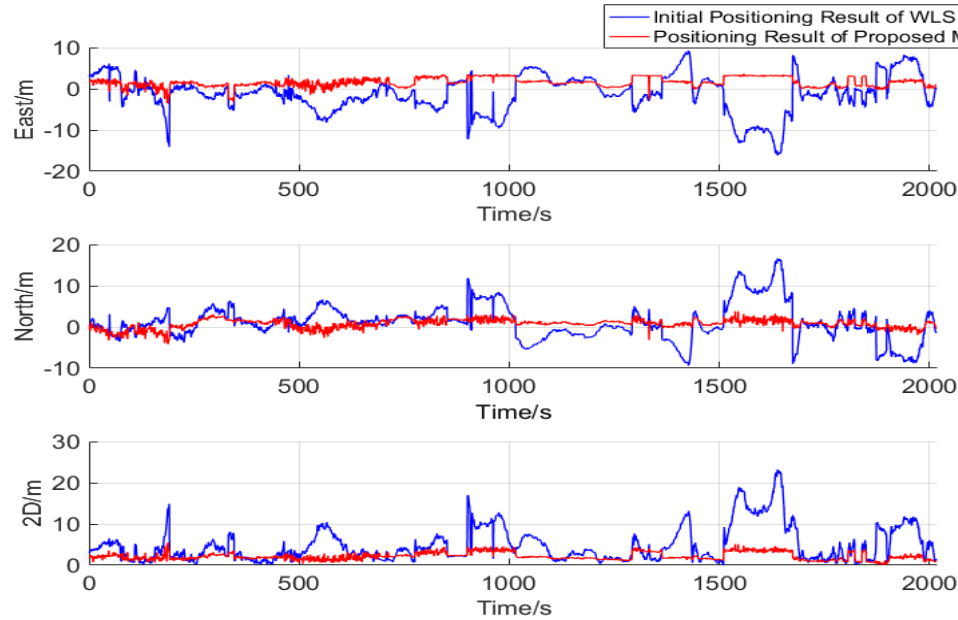
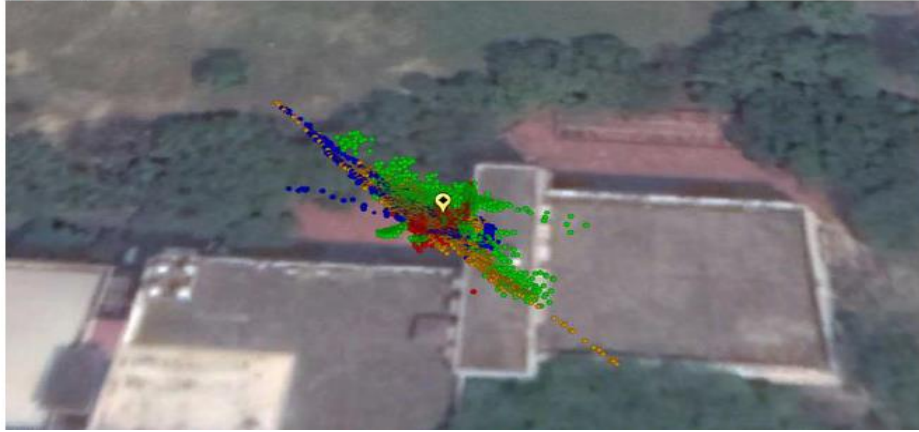
Positioning method using raw pseudorange data with a weighted least squares



	Method	E	N	2D
RMSE/m	Conventional Positioning Method	16.87	21.32	27.19
	ML+3D City Model	1.90	2.59	3.21
	Improvement (%)	88.74	87.85	88.19
95 th Percentile of Positioning Error/m	Conventional Positioning Method	38.06	40.36	52.44
	ML+3D City Model	4.41	4.15	5.59
	Improvement (%)	88.41%	89.72	89.34

Test Case 3: L-shaped corner + Dense canopy

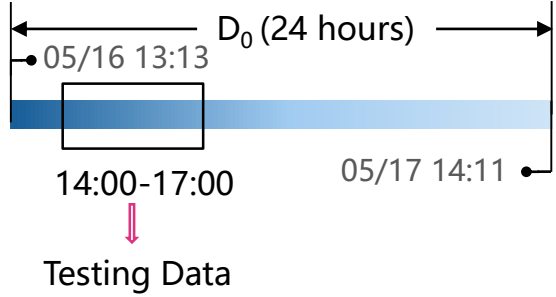
- 📍 Ground Truth
- Conventional Positioning Method
- C/N₀ based Positioning Method in [22], [23]
- Shadow Matching Method in [15], [16]
- Proposed Method



	Method	E	N	2D
RMSE/m	Conventional Positioning Method	4.68	4.48	6.47
	ML+3D City Model	1.90	1.35	2.34
	Improvement (%)	59.40	69.87	63.83
95 th Percentile of Positioning Error/m	Conventional Positioning Method	9.78	8.95	13.23
	ML+3D City Model	3.25	2.47	3.84
	Improvement (%)	66.77	72.40	70.98

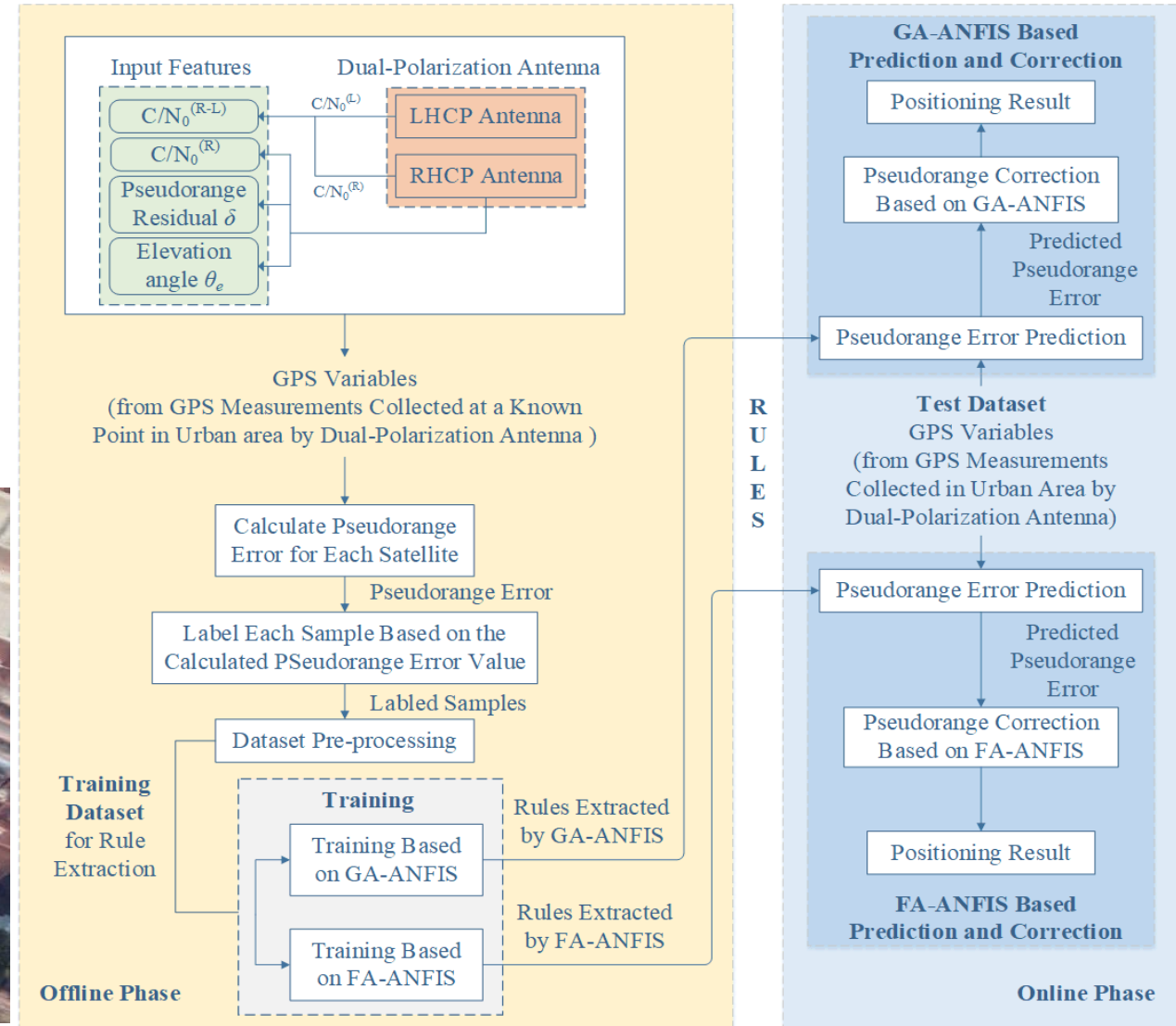
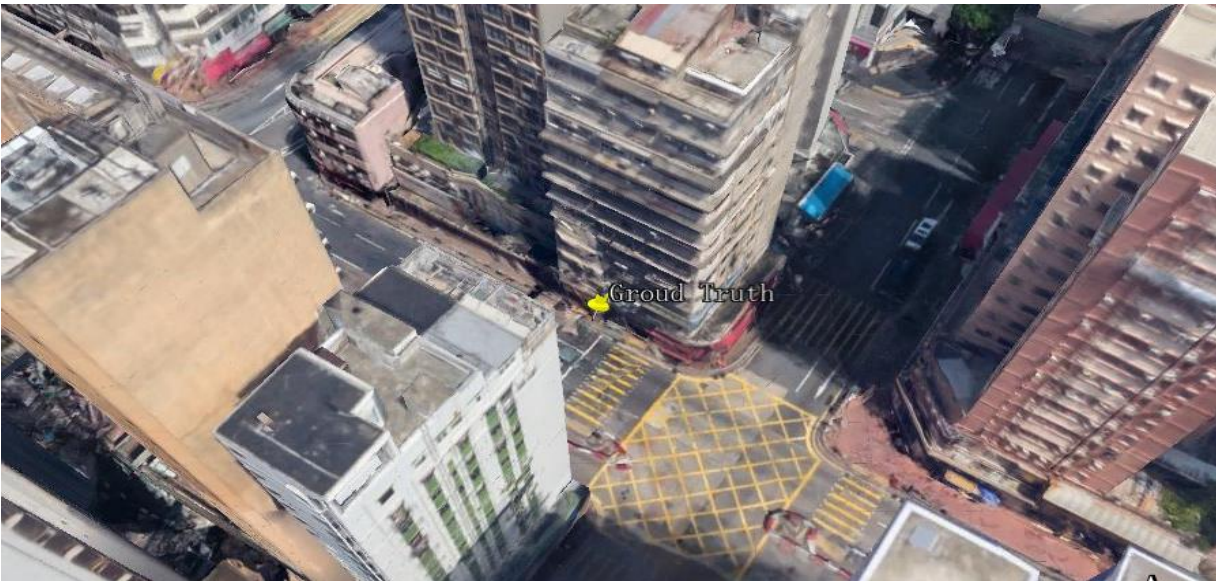
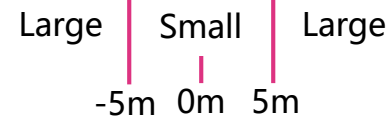
Dual-Polarization BDS/GNSS Antenna with Optimized Adaptive Neuro-Fuzzy Inference System to Improve Single Point Positioning Accuracy in Urban Canyons

Dataset:

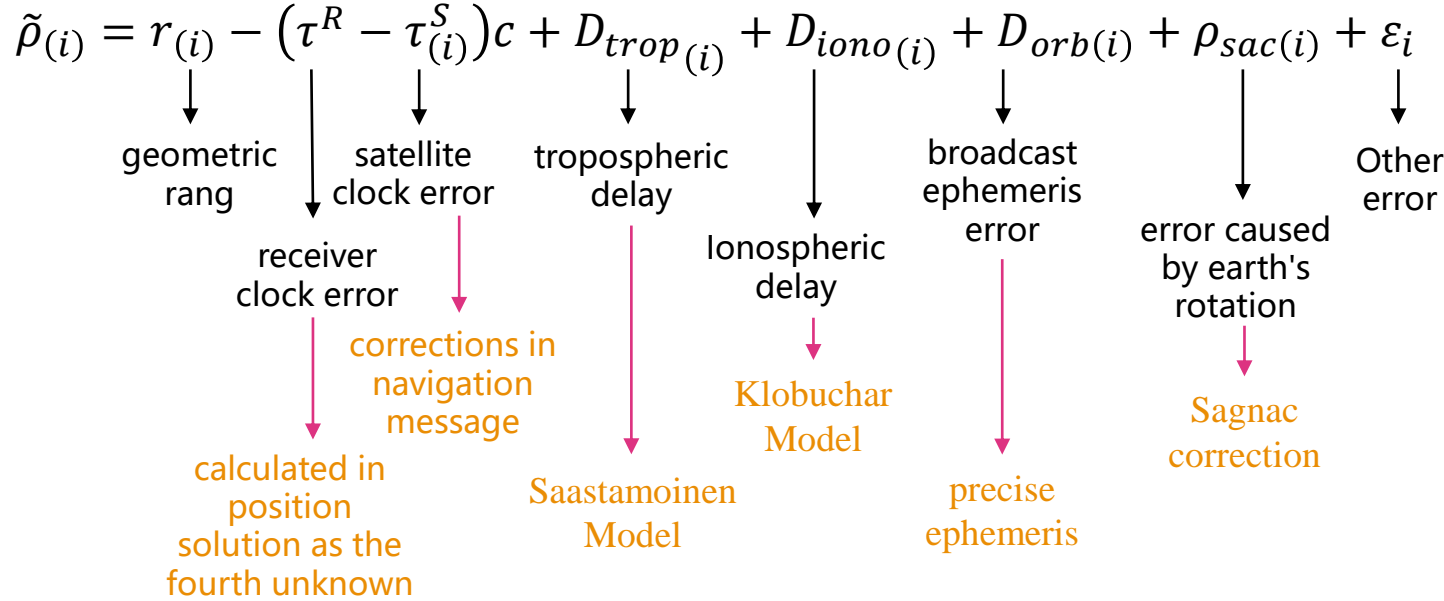


Dataset	Pseudorange Error		
	Large	Small	Total
testing	46,805	8,689	55,494
training	30,000	12,000	42,000
D ₀	/	/	388,022

Pseudorange Error Threshold:



Pseudorange Observation Equation :



Consists of the contribution to the range **error of the effects of Multipath/NLOS**, and observation noise

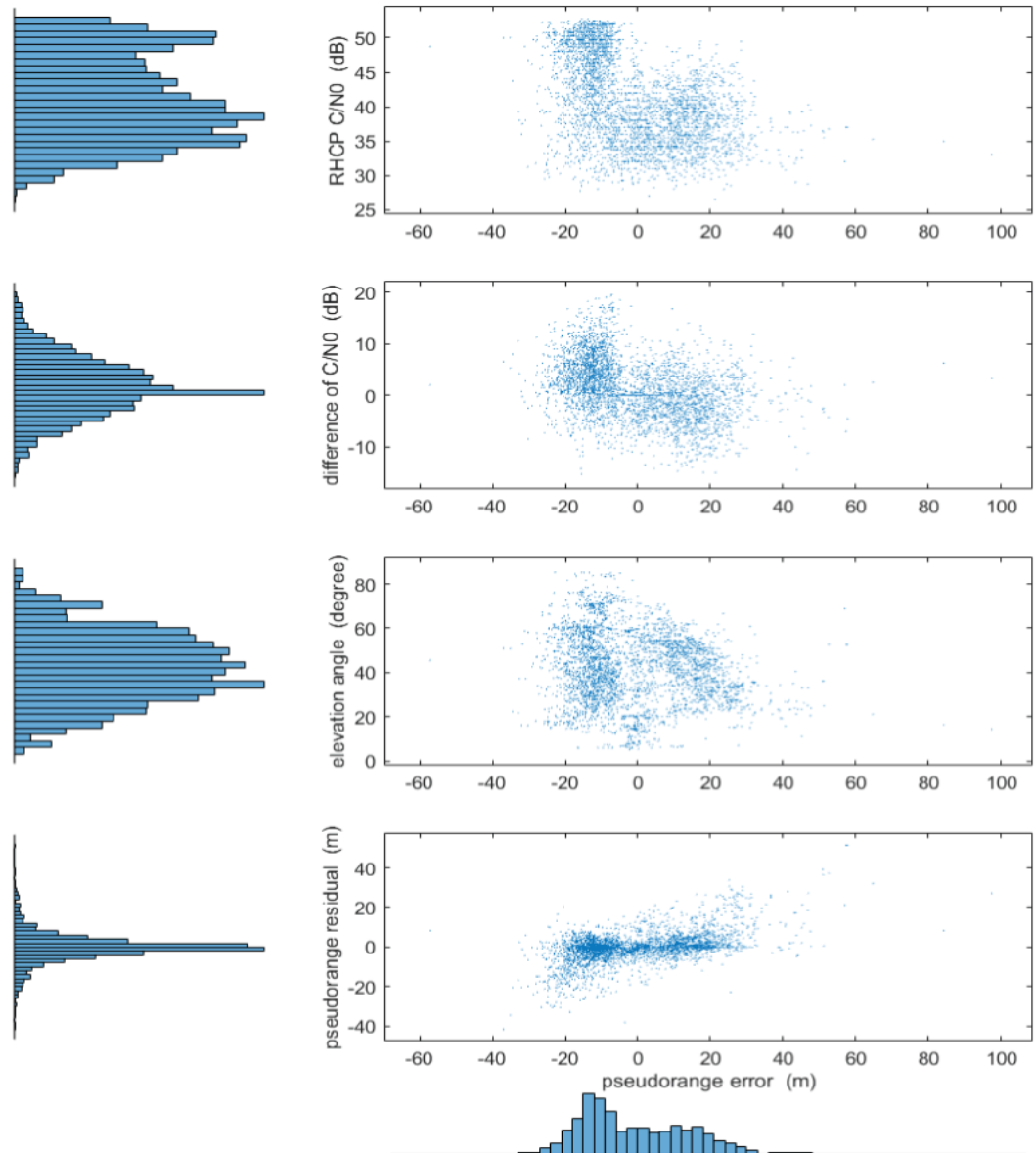
Given the limitations of the current mitigation methods, the error caused by **Multipath/NLOS** can reach tens of meters particularly in built environments, making it **dominant**.

Geometric Range: $r(i) = \sqrt{(X_{(i)}^S - X^R)^2 + (Y_{(i)}^S - Y^R)^2 + (Z_{(i)}^S - Z^R)^2}$

Satellite Position $(X_{(i)}^S, Y_{(i)}^S, Z_{(i)}^S)$ Receiver Position (X^R, Y^R, Z^R)

Pseudorange Error: $\Delta\rho(i) = \rho_{(i)}^c - r(i) = (\Delta\tau^R - \Delta\tau_{(i)}^S)c + \Delta D_{trop}^K + \Delta D_{iono}^S + \Delta D_{orb} + \varepsilon_i$

Correlation Analysis Between Input Features And Pseudorange Errors



Input Features:

- RHCP signal strength ($C/N_0^{(R)}$)
- Signal strength difference obtained from RHCP and LHCP antenna ($C/N_0^{(R-L)}$)
- Elevation angle (θ_e)
- Pseudorange residual (δ)

Spearman correlation coefficient	Correlation
0.00-0.19	Very weak
0.20-0.39	Weak
0.40-0.59	Medium
0.60-0.79	Strong
0.80-1.00	Very Strong

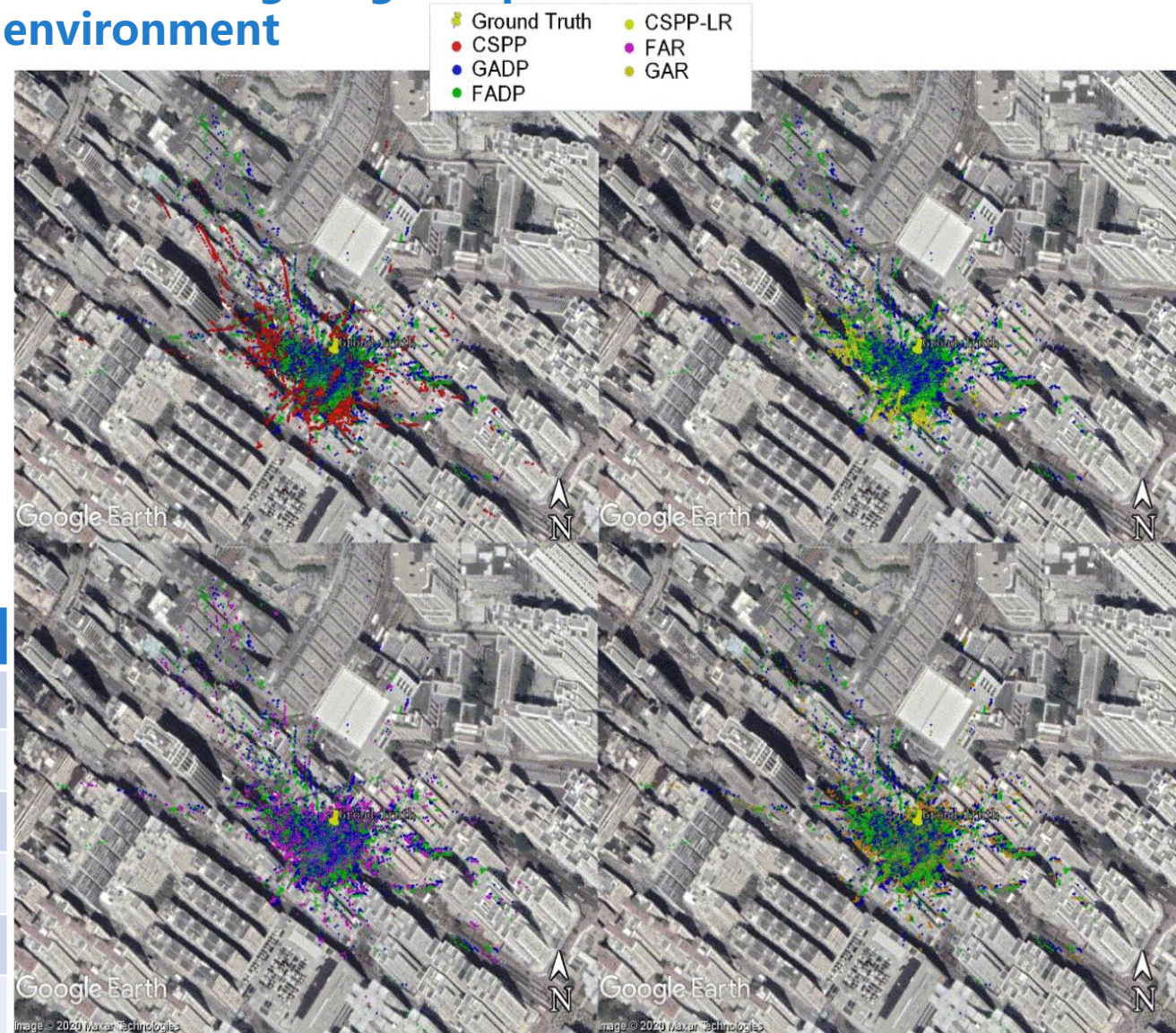
Input Feature	$C/N_0^{(R)}$	$C/N_0^{(R-L)}$	Elevation angle	Pseudorange residual
Spearman correlation coefficient	-0.4934	-0.4318	-0.1523	0.4107
Correlation	Medium	Medium	Very weak	Medium

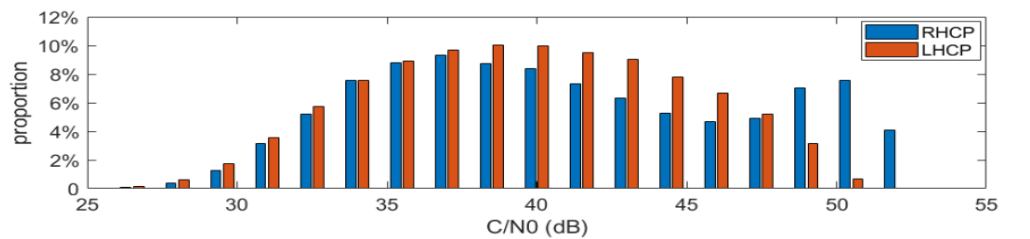
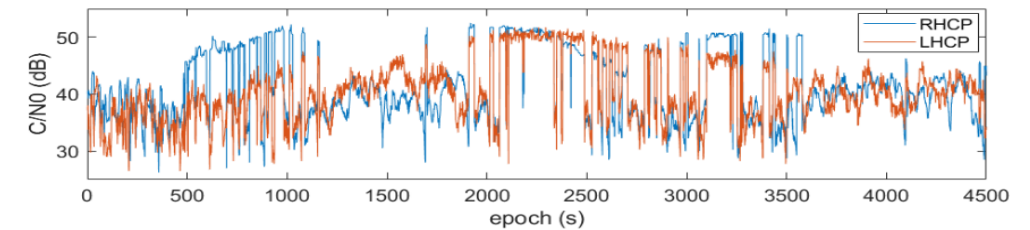
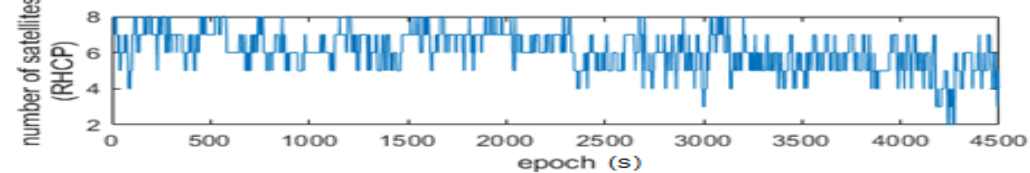
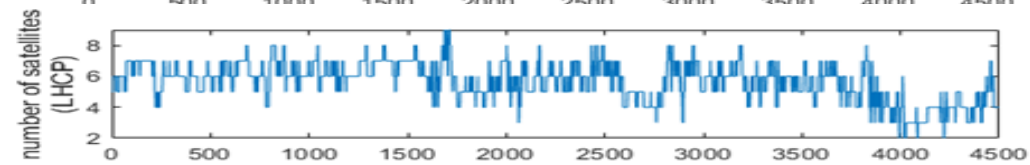
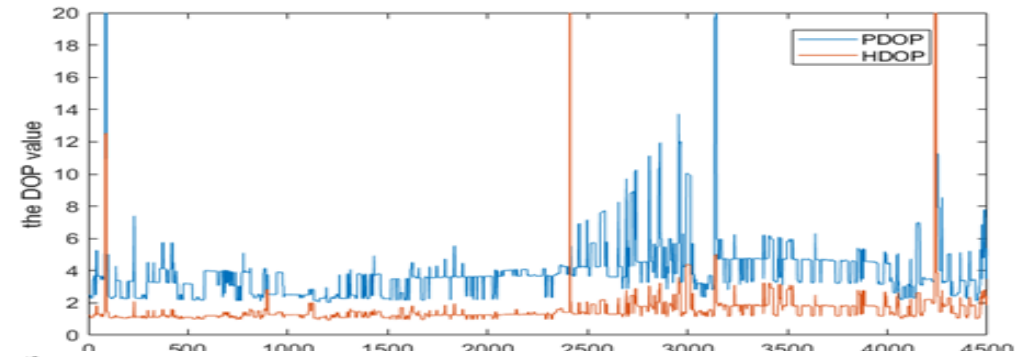
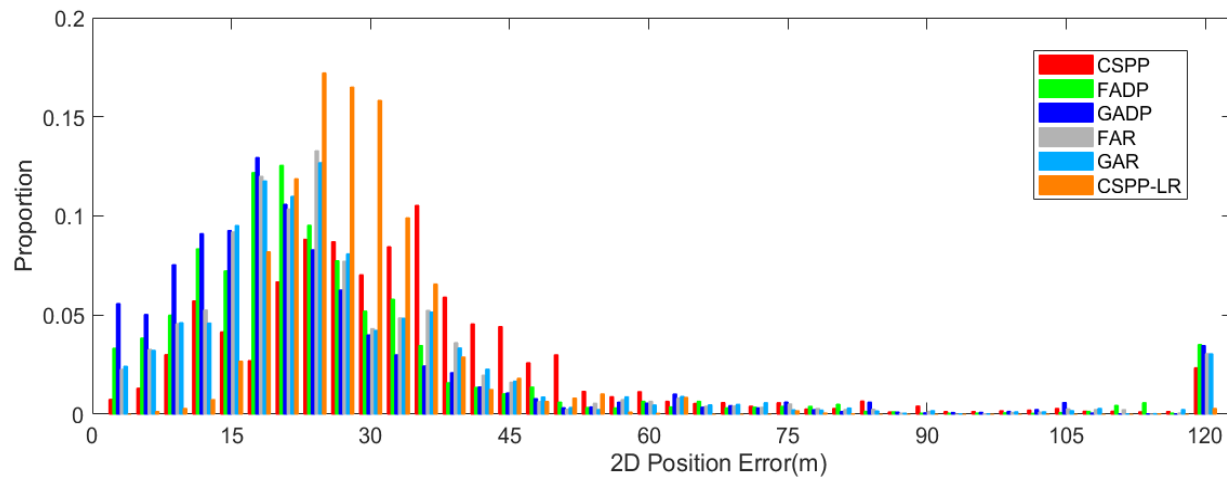
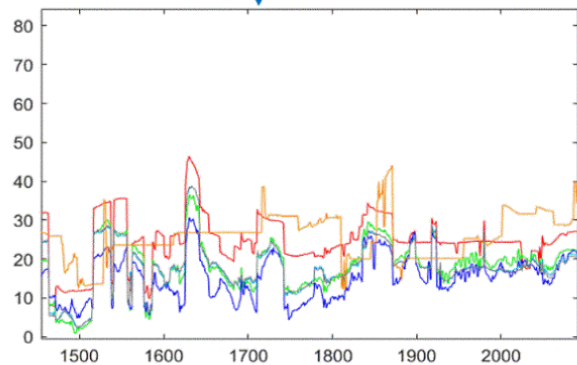
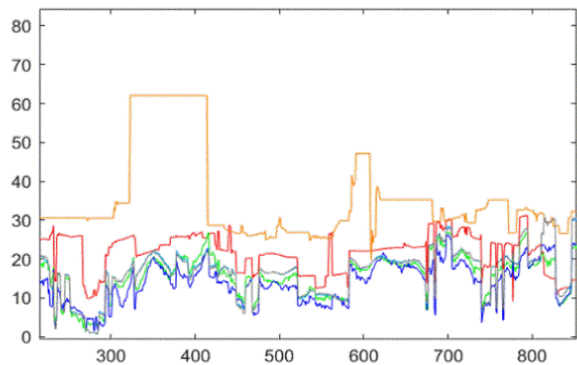
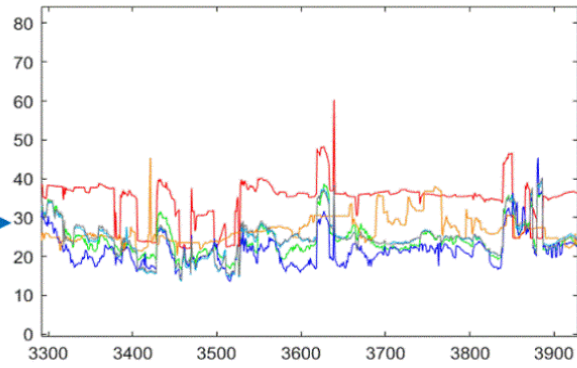
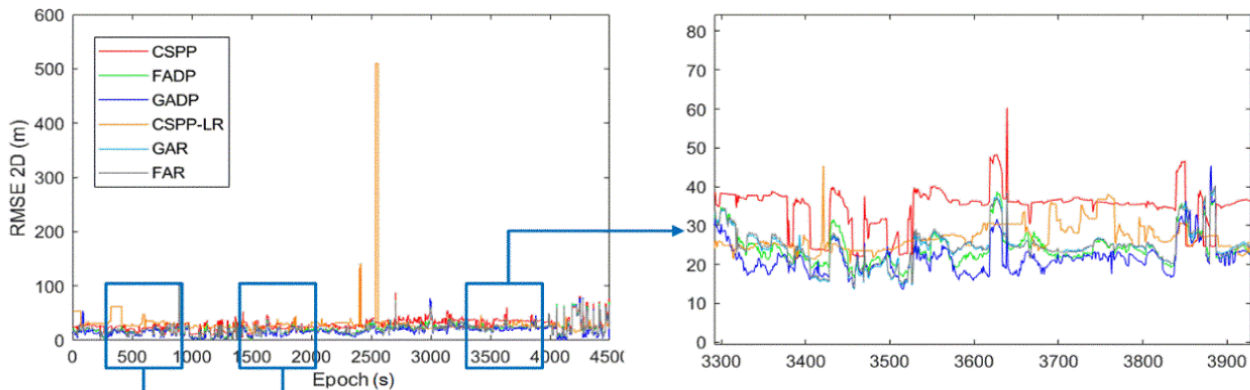
Several Algorithms for Comparison

- **Conventional Single Point Positioning (CSPP) method**, i.e. positioning with outlier detection and exclusion, which uses Efficient Leave One Block Out (ELOBO) approach to identify outliers and exclude them from the positioning process.
- Conventional Single Point Positioning using the **LHCP-RHCP C/N₀ difference** and **satellite elevation angle** to select and weight the measurements. (**CSPP-LR**)
- **FA-ANFIS** using **RHCP** measurement data only. (**FAR**)
- **GA-ANFIS** using **RHCP** measurement data only. (**GAR**)
- Pseudorange errors predicted by **GA-ANFIS** and **FA-ANFIS** with **Dual Polarization antenna** (noted as **GADP** and **FADP**)

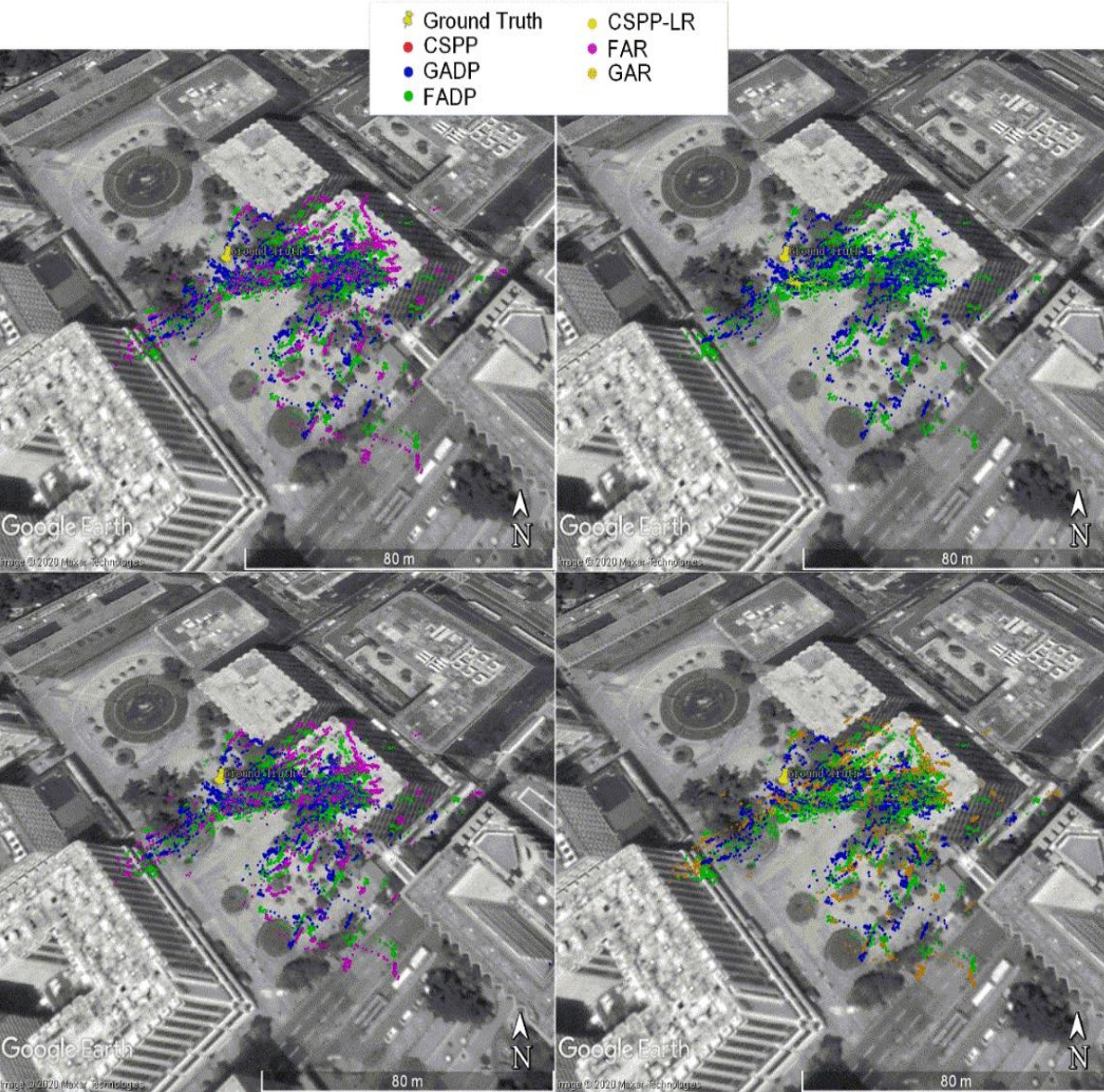
RMSE (m)	E	N	U	2D	3D
CSPP	33.46	28.57	112.14	44.00	120.46
CSPP-LR	27.31	31.54	143.49	41.72	149.44
FAR	29.86	21.61	95.21	36.86	102.10
GAR	29.09	21.90	98.52	36.42	105.03
FADP	26.90	19.22	82.92	33.06	89.27
GADP	25.29	16.65	76.19	30.28	81.99

Case 1: Hongkong, deep urban environment





Case 2: Hongkong, mid urban environment

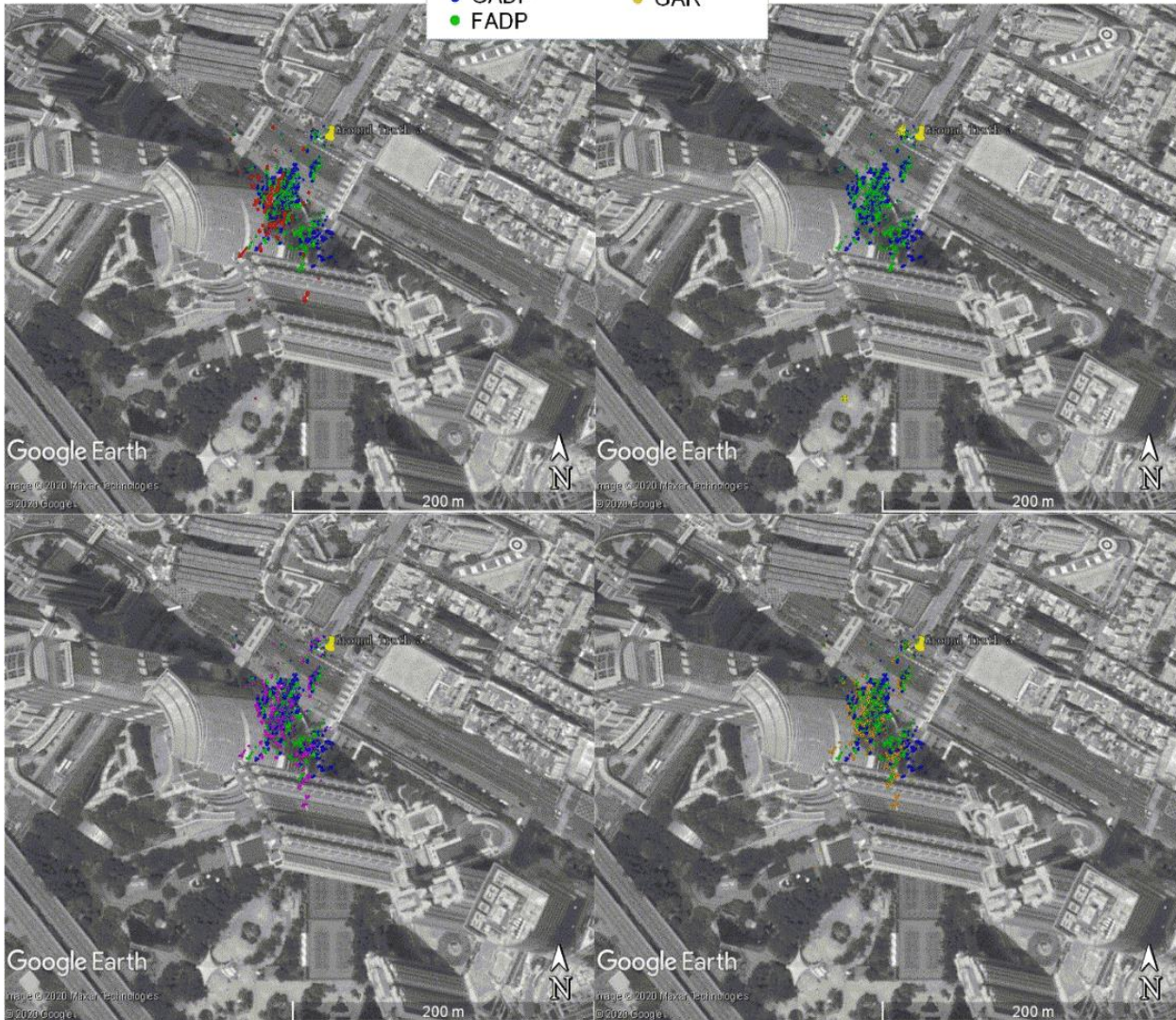


RMSE (m)	E	N	U	2D	3D
CSPP	40.92	17.90	79.01	44.67	90.76
CSPP-LR	27.57	23.92	24.12	36.50	43.75
FAR	38.30	16.21	59.38	41.59	72.50
GAR	36.64	15.89	59.86	39.94	71.96
FADP	36.27	15.11	56.98	39.30	69.21
GADP	33.13	13.70	58.61	35.85	68.96

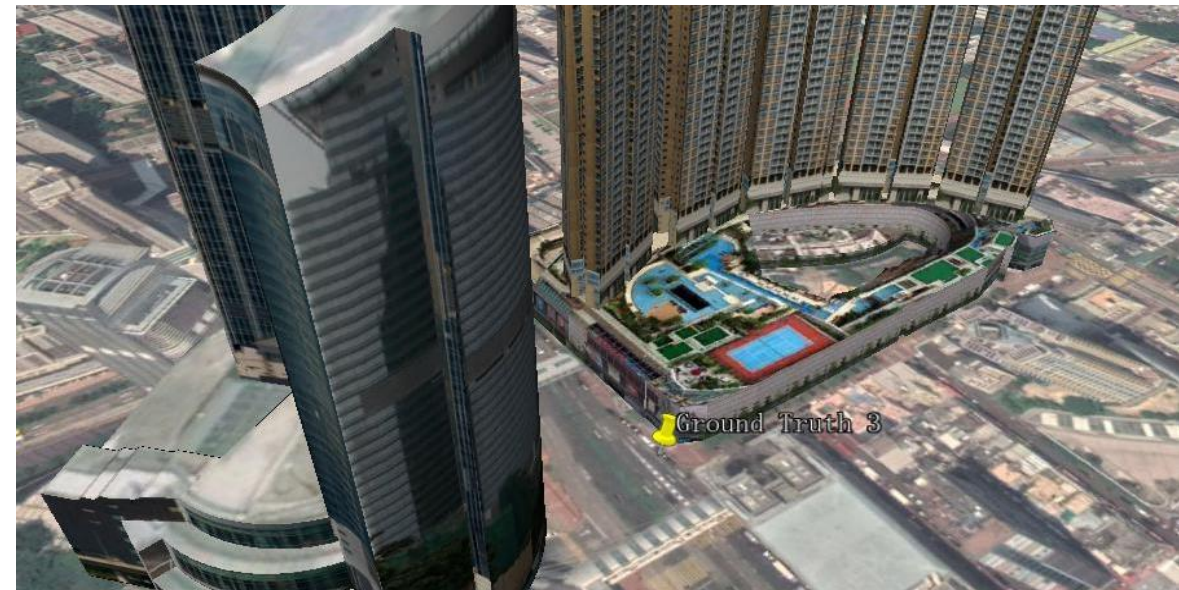


Case 3: Hongkong, tall buildings on both sides

- ★ Ground Truth
- CSPP
- GADP
- FADP
- CSPP-LR
- FAR
- GAR



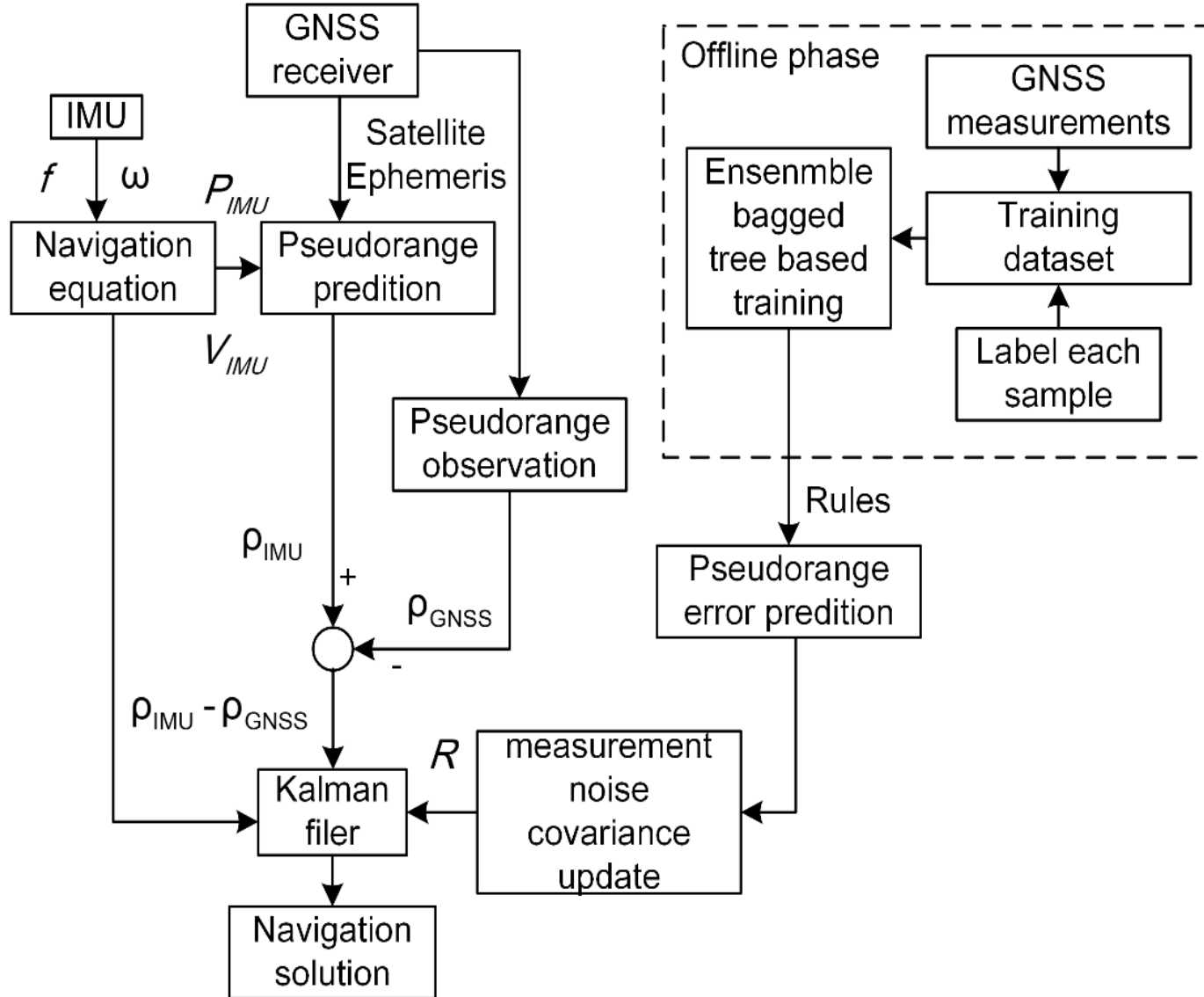
RMSE (m)	E	N	U	2D	3D
CSPP	36.45	55.27	40.52	66.21	77.62
CSPP-LR	20.51	60.92	25.84	64.28	69.28
FAR	34.87	53.06	37.02	63.49	73.49
GAR	34.66	53.43	37.36	63.69	73.84
FADP	32.65	48.89	42.03	58.79	72.27
GADP	31.73	47.92	48.38	57.48	75.12





- The proposed algorithm results in a **30%** improvement in Root Mean Square Error (RMSE) in the 2D (horizontal) component for static applications when the training and testing data are collected at the same location. This corresponds to **13 to 20%** when the testing data is from locations away from that of the training dataset

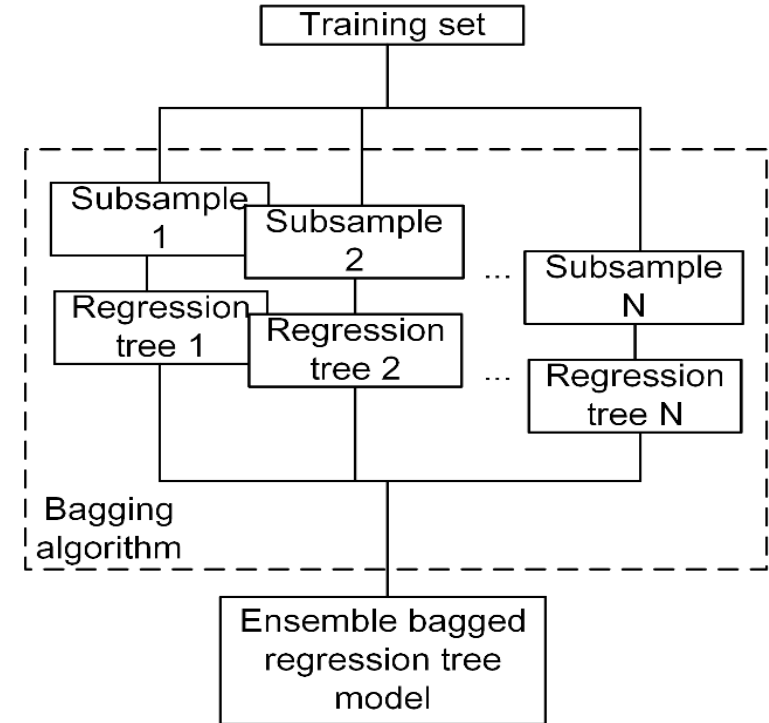
Pseudorange Error Prediction for Adaptive Tightly-Coupled BDS/GNSS/INS Navigation in Urban Areas



➤ In the traditional Kalman filter

The BDS/GNSS measurement noise is fixed **based on factors determined a priori**, instead of reflecting the **impact of the surrounding environment** on the received BDS/GNSS signal.

Degrading the **position accuracy** and **posteriori quality indicators**.



1) Construct the adaptive indicator

$$f(\cdot) = (|h_{bag}(\cdot)| + \eta)^2$$

$h_{bag}(\cdot)$: designed ensemble bagged tree model, $\eta = 0.1$

2) Adjust measurement noise covariance matrix

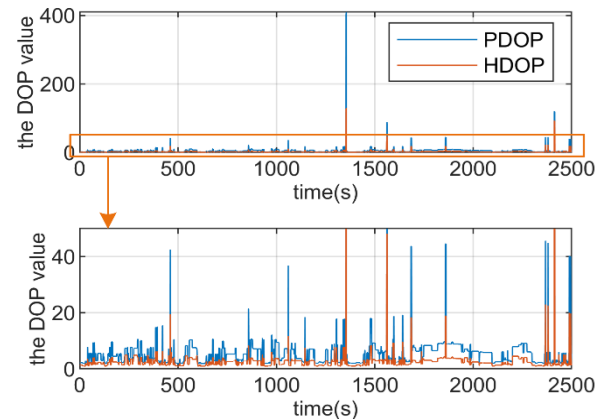
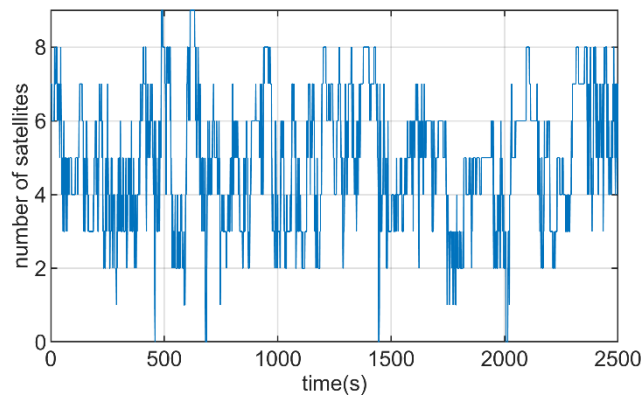
$$[R_{(m,m)}]_k = f([x_m]_k)$$

$x = (C/N_0, \theta, L, atLon)$

m means the serial number of the current satellite in the received satellite at the epoch k ;

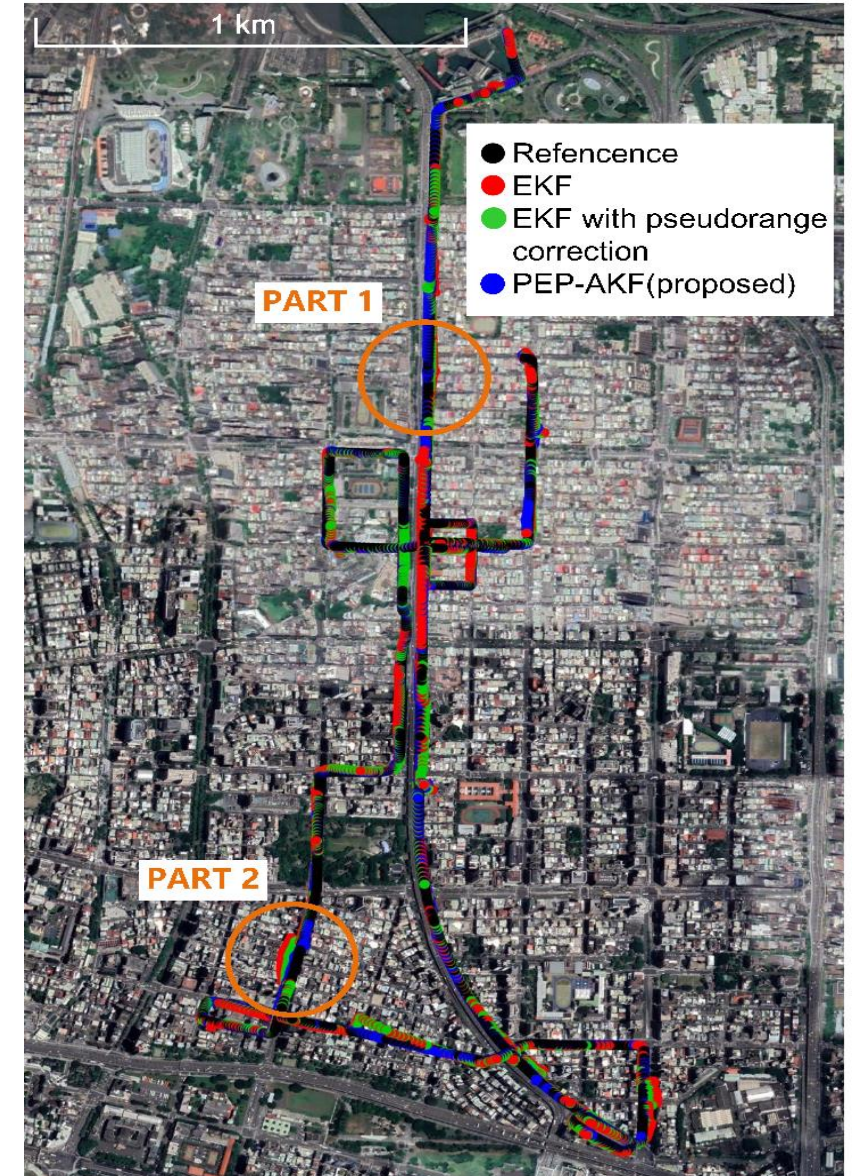
$R_{(m,m)}$ denotes the m row, m column of the measurement noise covariance matrix.

The **closer** the **adaptive indicator estimation** is to **R**, the **closer** the **filter outputs** are to **ideal results**.



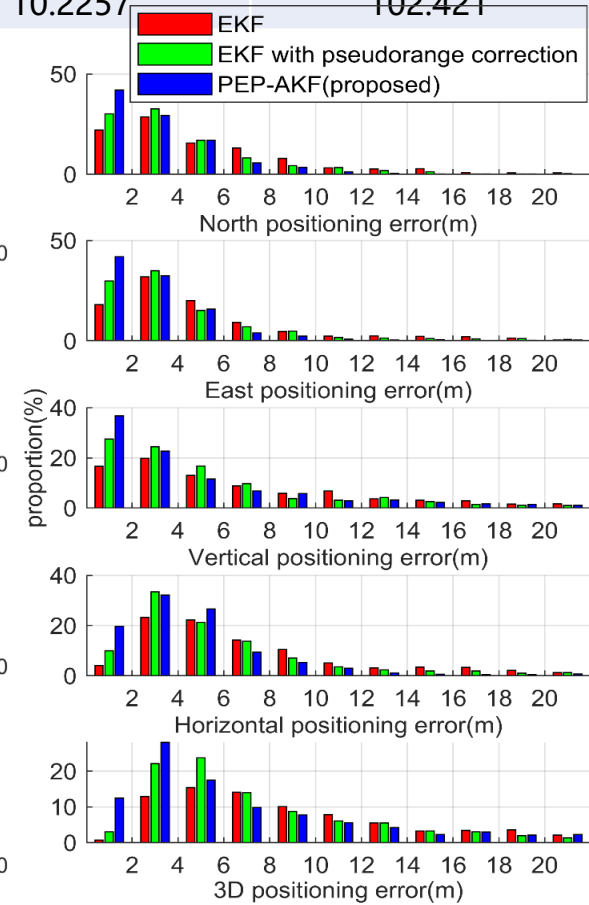
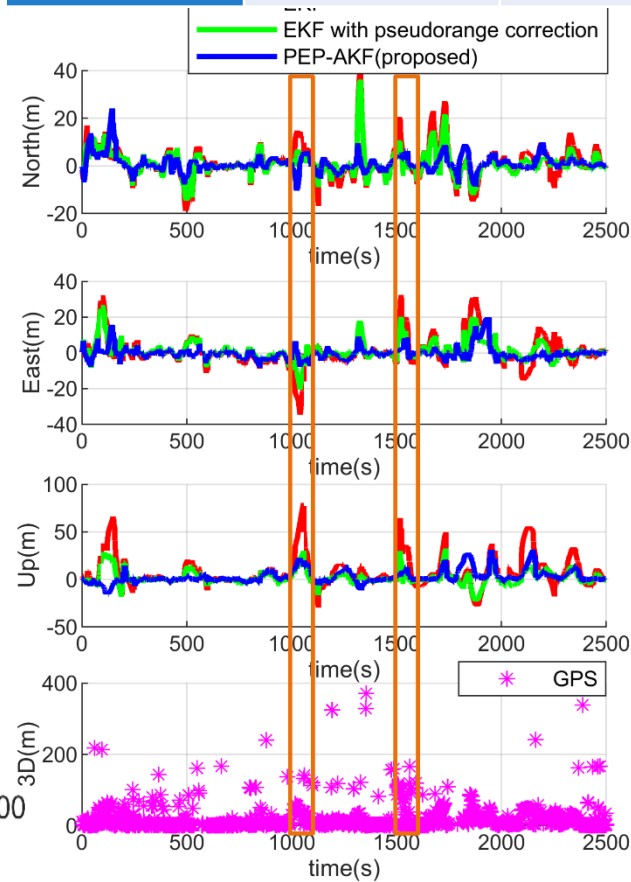
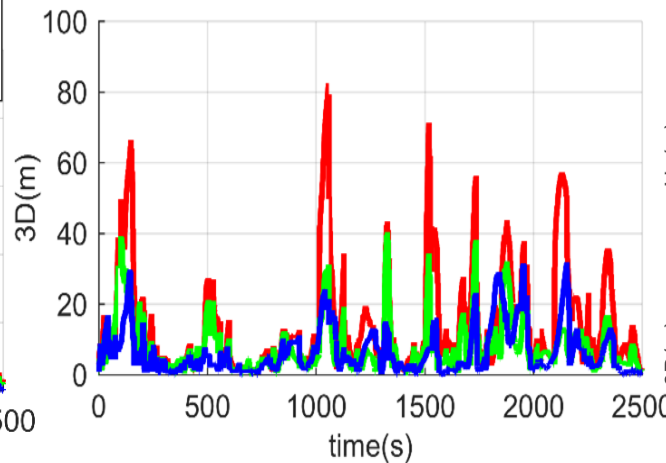
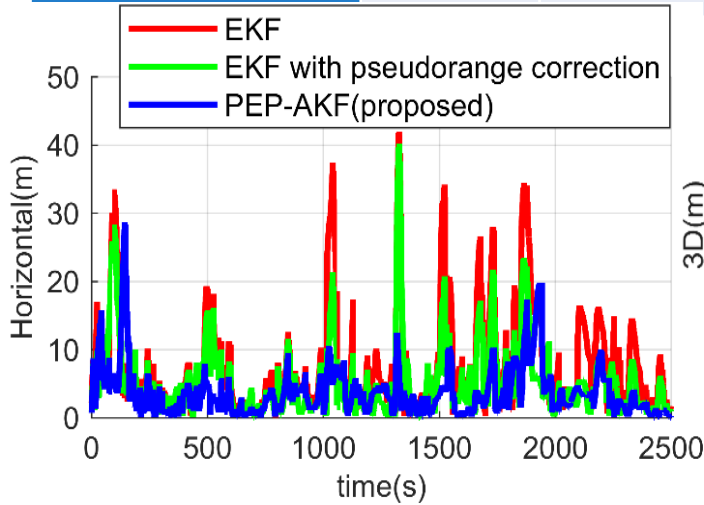


Algorithm	Algorithm Description	Color
EKF	Tightly integration of BDS/GNSS and INS with Extended Kalman filter	Red
EKF with pseudorange correction	<p>Step 1: Using the trained ensemble bagged trees model to predict the pseudorange error and then correct pseudorange.</p> <p>Step 2: Using the corrected pseudorange to form the measurements vector Z in the EKF of the tightly fusion.</p>	Green
PEP-AKF (proposed)	<p>Step 1: Using the trained ensemble bagged trees model to predict the pseudorange error and then construct adaptive indicator.</p> <p>Step 2: Using adaptive indicator to adjust measurement noise covariance matrix in the KF of the tightly fusion.</p>	Blue

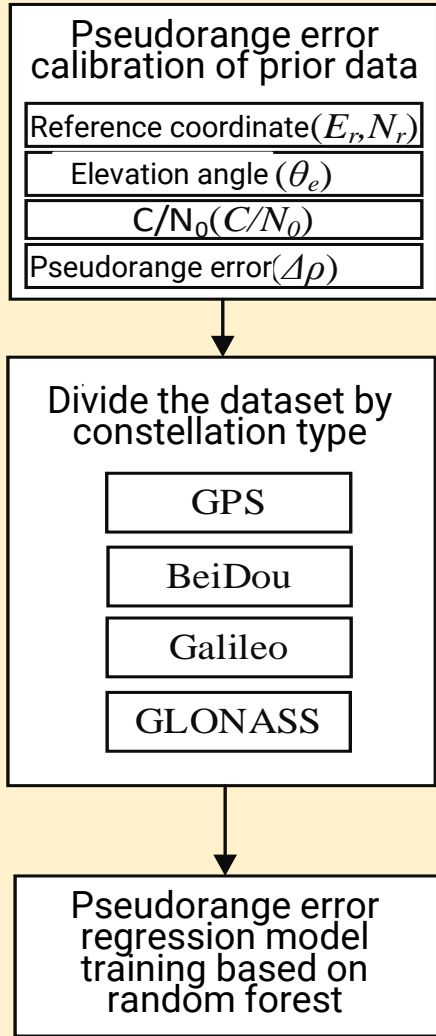


BDS/GNSS/IM U tightly fusion algorithms	RMSE/m				
	North	East	Up	Horizontal	3D
EKF	6.83	8.20	17.80	10.67	20.76
EKF with pseudorange correction	5.27	5.38	7.81	7.53	10.85
Improvement	22.75%	34.46%	56.12%	29.43%	47.72%
PEP-AKF (proposed method)	3.60	3.63	7.66	5.11	9.21
Improvement	47.26%	55.78%	56.98%	52.11%	55.64%

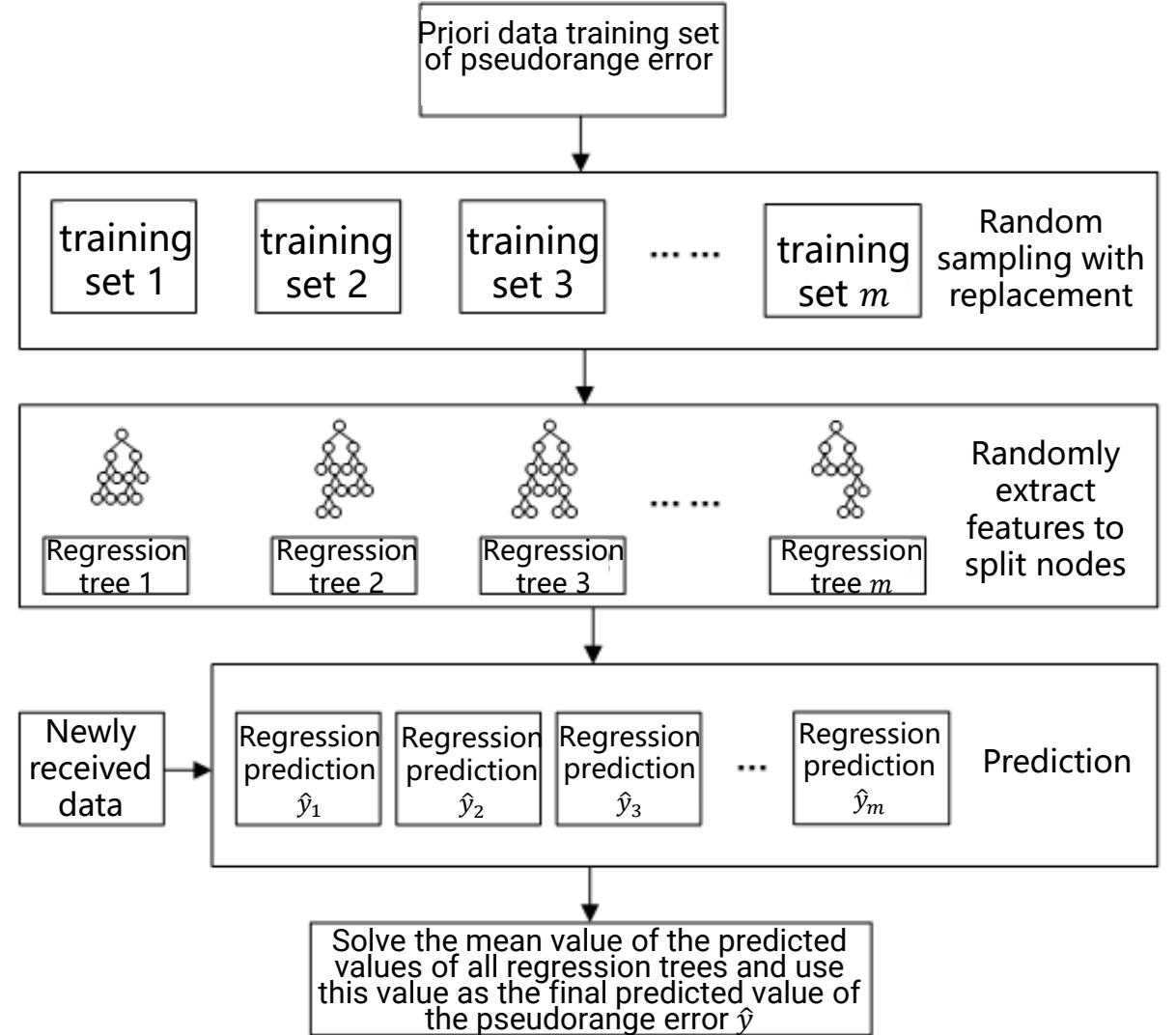
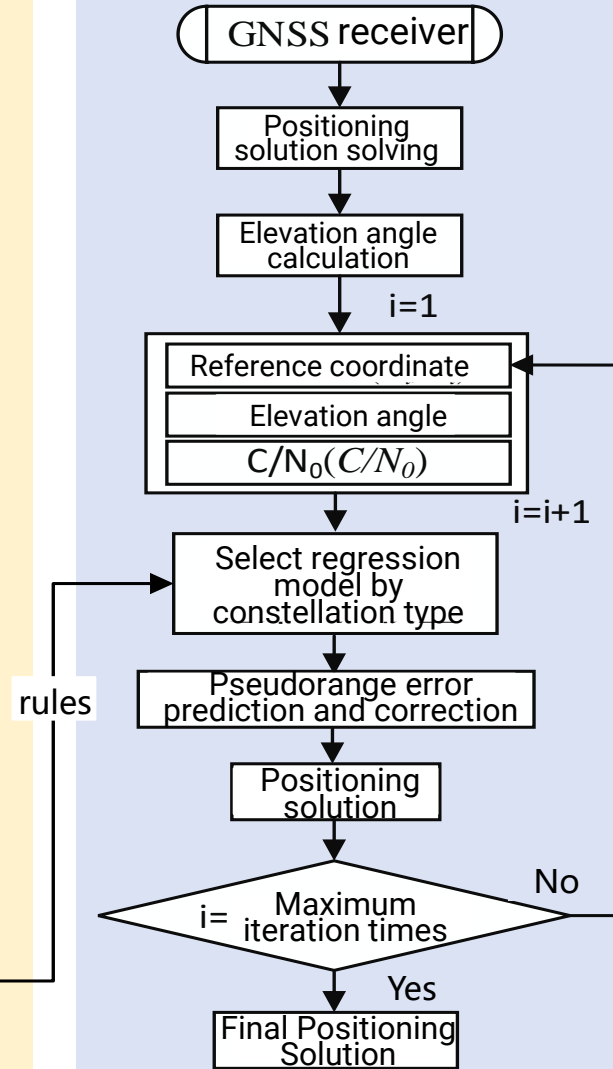
Time(s)	Satellite	Pseudorange error/m	Adaptive indicator
633	30	0.5538	0.296
647	30	0.9201	0.799
1015	30	18.9460	298.856
1071	30	10.2257	102.421



Offline phase



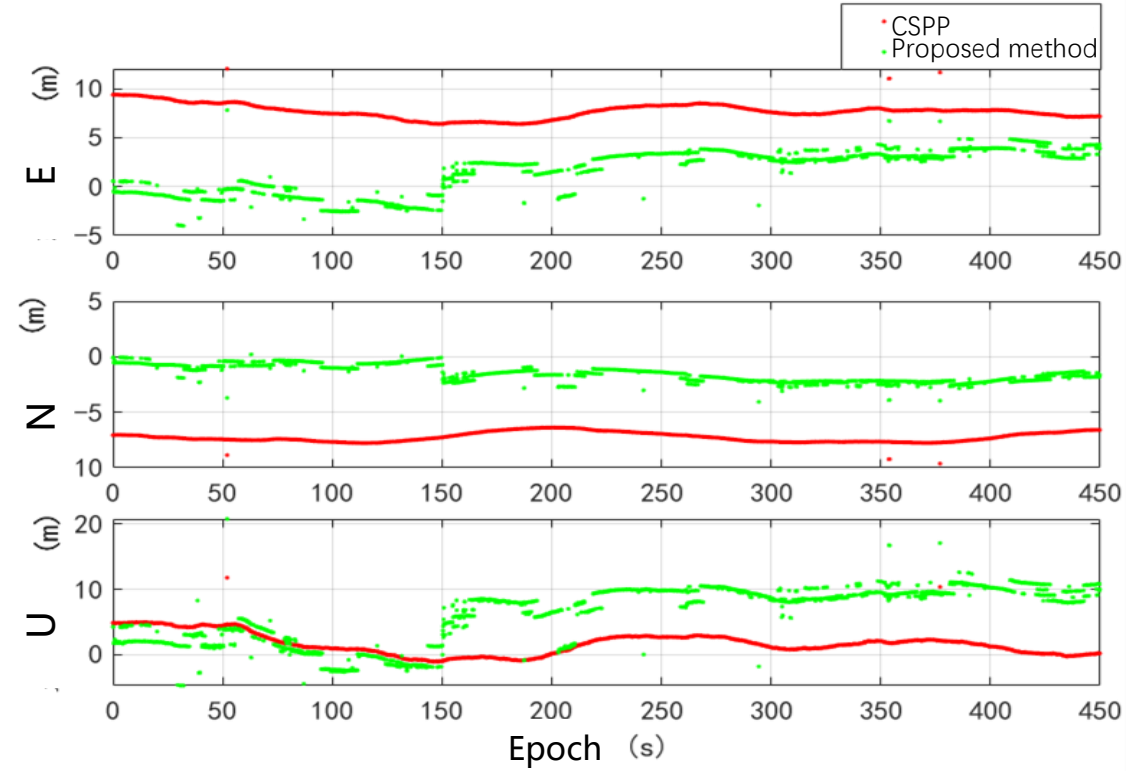
Online phase



Static Test : Hongkong, mid urban environment

10Hz, 15 mins **BDS/GNSS**

Receiver: NovAtel OEM6; Antenna: ZYACF-L004



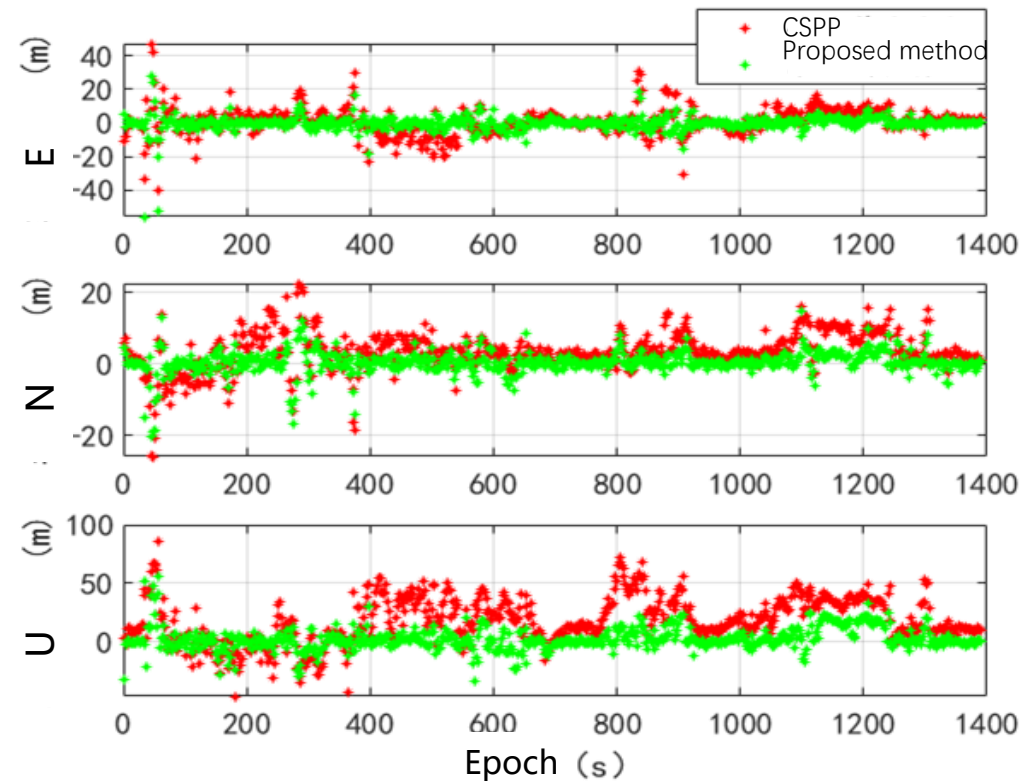
RMSE (m)	1D			2D	3D
	E	N	U	horizontal	
CSPP	7.7334	7.2615	2.3196	10.6083	10.8589
Proposed method	2.6775	1.6737	7.5285	3.1576	8.1639
improvement	65.38%	76.95%	-224.56%	70.23%	24.82%

Road Test: deep urban environment

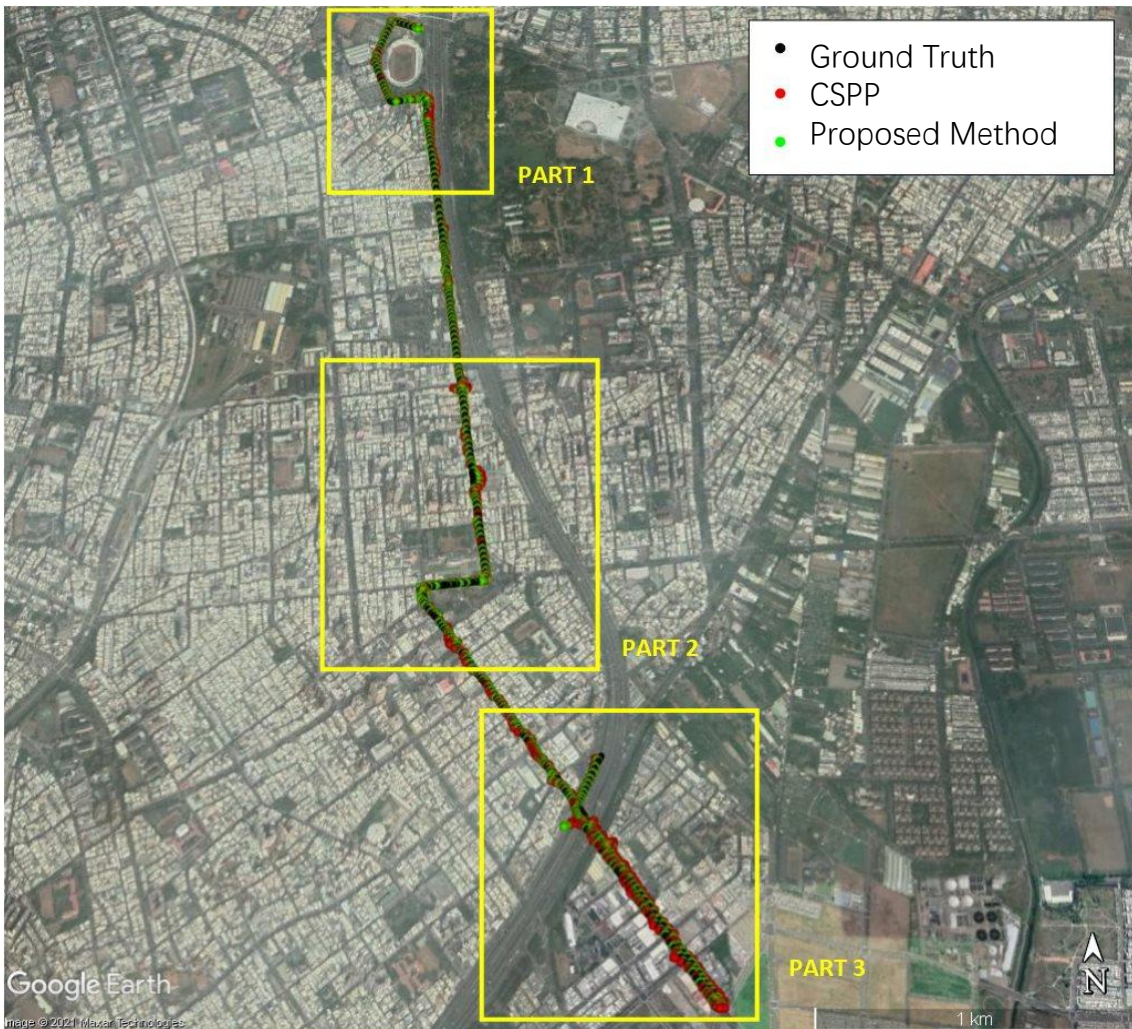
Reference: Receiver: NovAtel SPAN®-LCI; Antenna: NovAtel GPS-GGG-703-HV

Testing Data: Receiver: Allystar EVK-2024 ; Antenna: Allystar AGR 6301

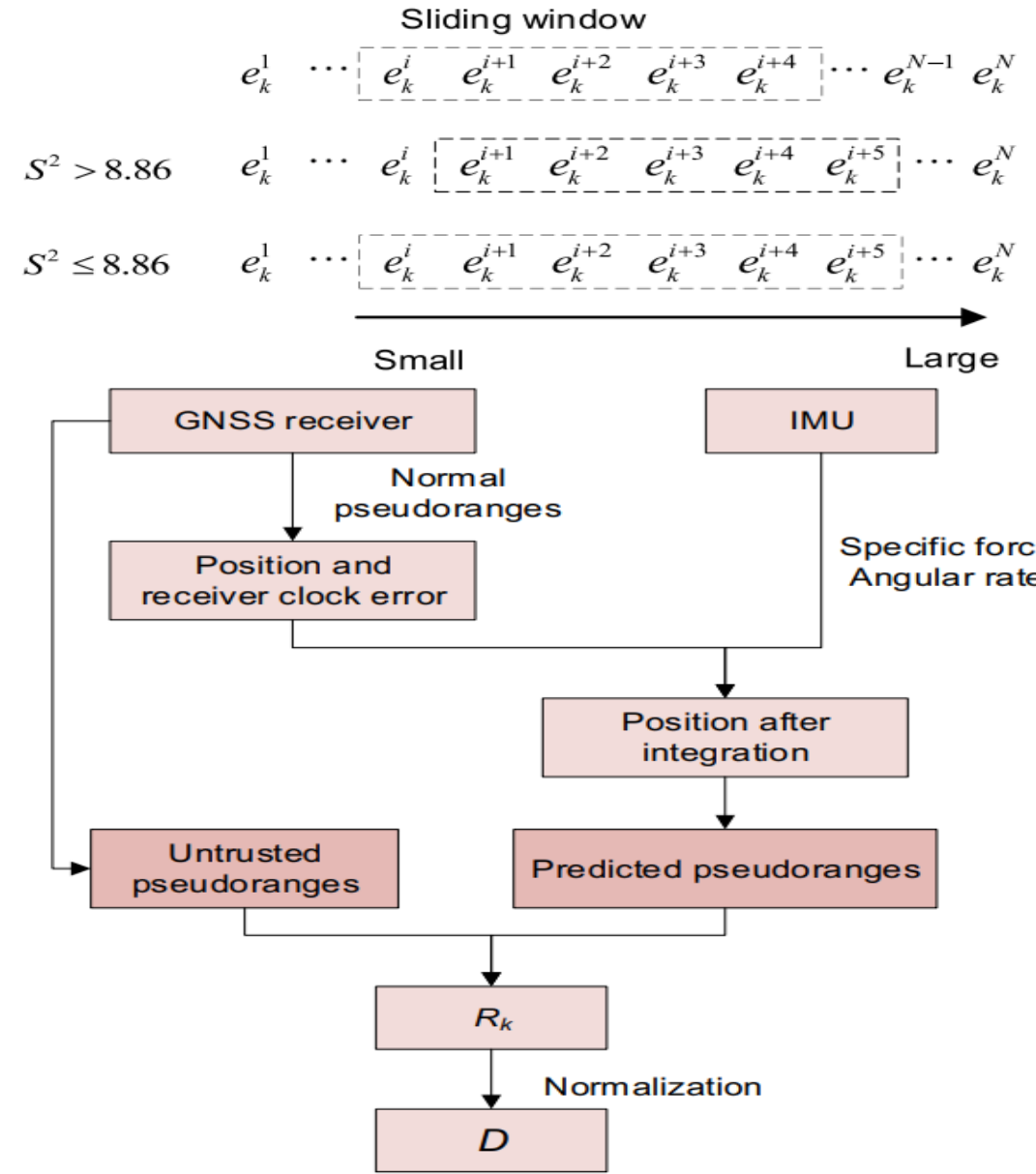
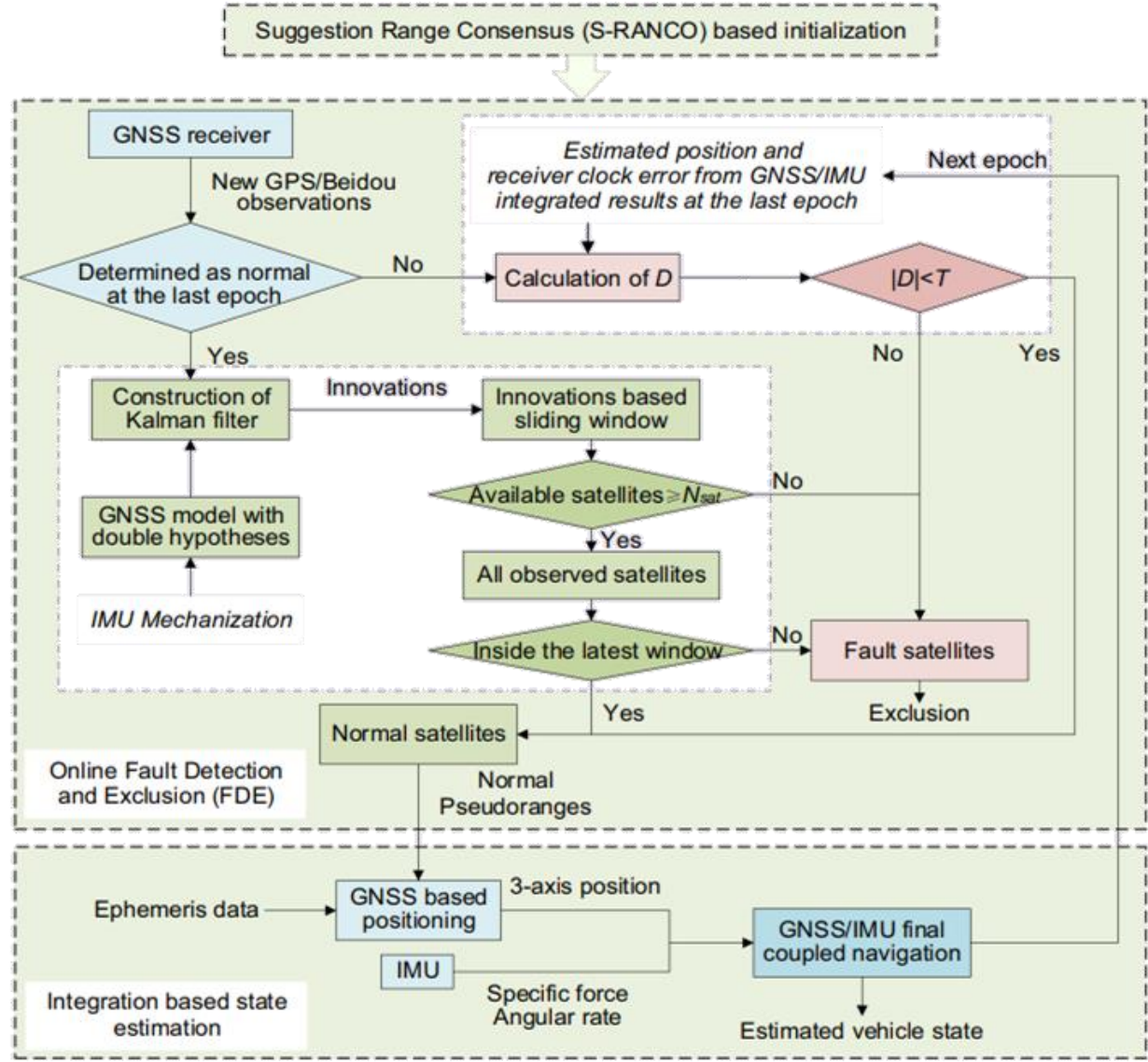
BDS/GNSS, 10 Hz, 20 mins, Average speed: 26 km/h **Crowdsourced Data**



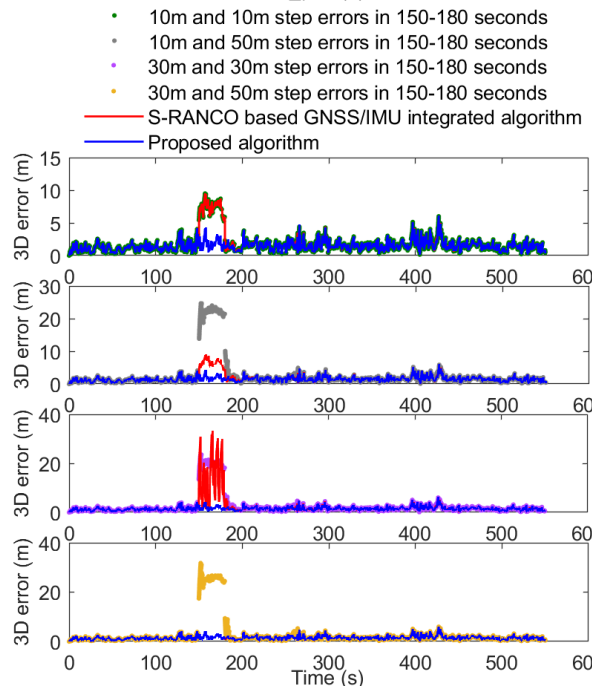
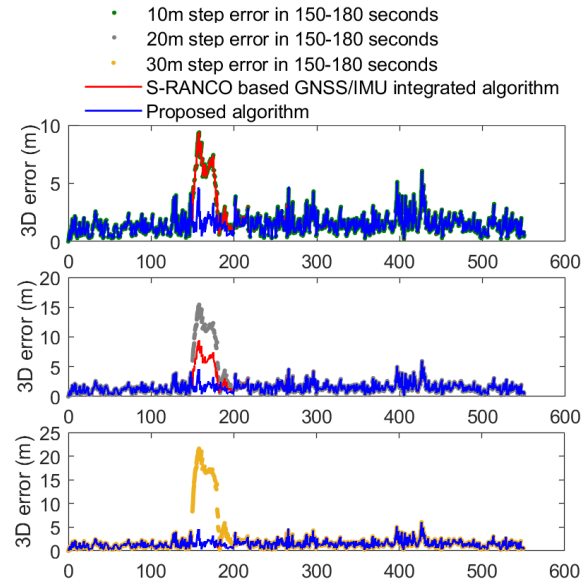
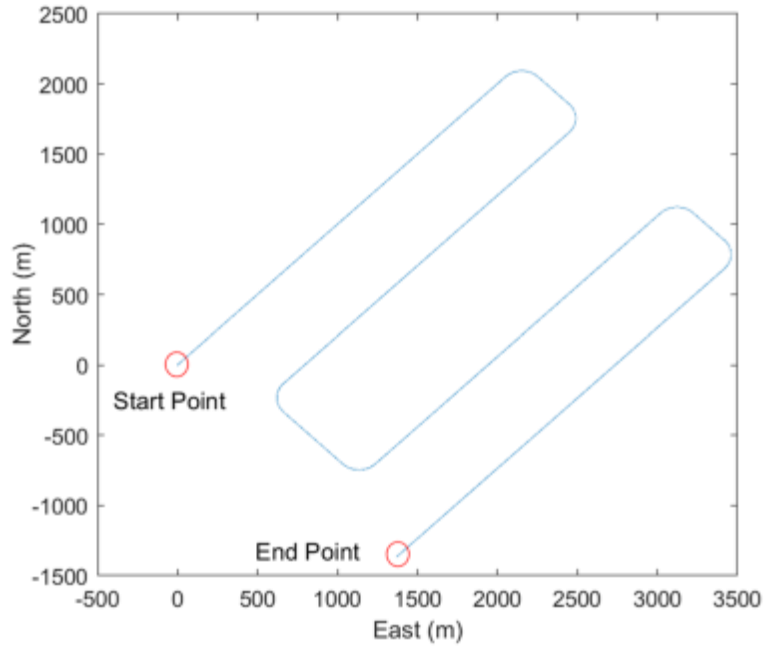
RMSE (m)	1D			2D	3D
	E	N	U	horizontal	
CSPP	7.5293	6.3063	26.8642	9.8214	28.6032
Proposed method	4.6455	3.2010	9.8762	5.6415	11.3739
improvement	38.30%	49.24%	63.24%	42.56%	60.24%



IMU-Aided Multiple BDS/GNSS Fault Detection and Exclusion Algorithm for Integrated Navigation in Urban Environments



Simulation



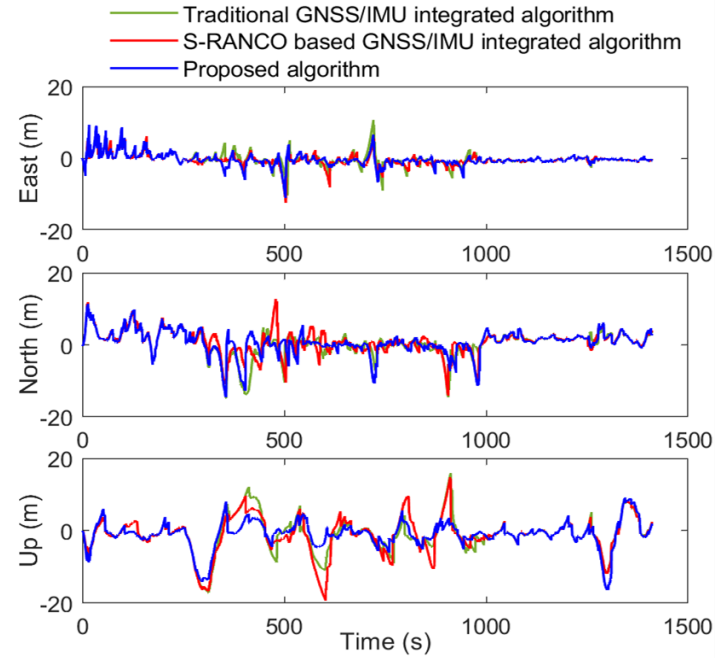
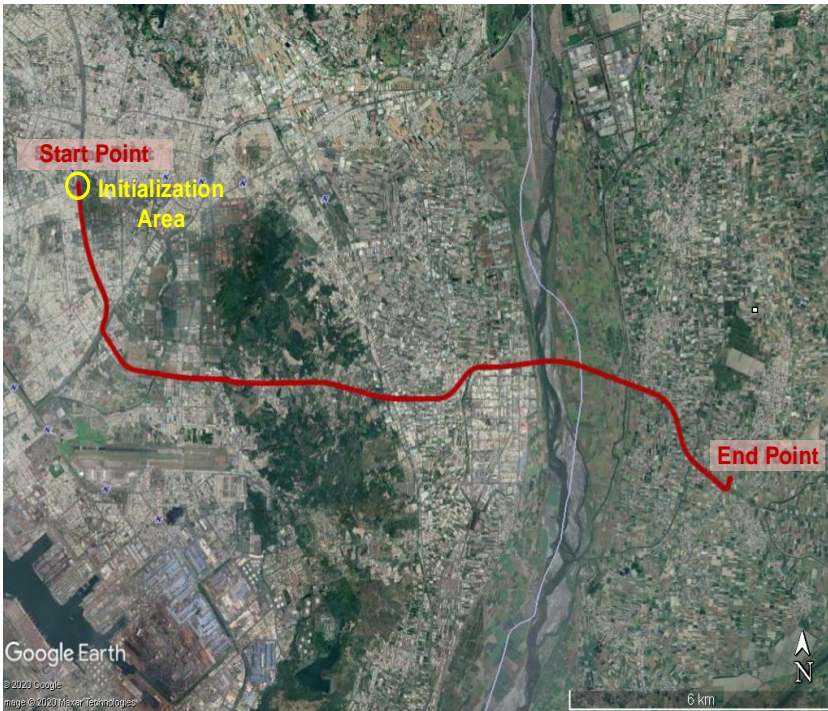
Case	Number of fault satellites	Time interval of faults	Error source
1	1	30 s	10 m, 20 m and 30 m step errors added to the pseudorange of one satellite
2	2	30 s	10 m, 30 m and 50 m step errors added to the pseudoranges of two satellites

Step error	Traditional GNSS/IMU integration	S-RANCO based GNSS/IMU integration		Proposed algorithm	
	Positioning accuracy	Correct detection rate	Positioning accuracy	Correct detection rate	Positioning accuracy
10 m	6.41 m	0%	6.41 m	100%	1.91 m
20 m	11.71 m	53.3%	7.63 m	100%	1.91 m
30 m	17.06 m	100%	1.91 m	100%	1.91 m

Step error of PRN 2	Step error of PRN 12	Traditional GNSS/IMU integration	S-RANCO based GNSS/IMU integration		Proposed algorithm	
		Positioning accuracy	Correct detection rate	Positioning accuracy	Correct detection rate	Positioning accuracy
10 m	10 m	7.40	0%	7.40 m	100%	1.95 m
10 m	50 m	22.11	0%	22.11 m	100%	1.95 m
30 m	30 m	19.91	40%	17.12 m	100%	1.95 m
30 m	50 m	25.61	100%	1.95 m	100%	1.95 m

Test Case 1: mid urban environment

BDS/GNSS, Novatel PwrPak7, 1 Hz
IMU, Bosch BMI055, 50 Hz

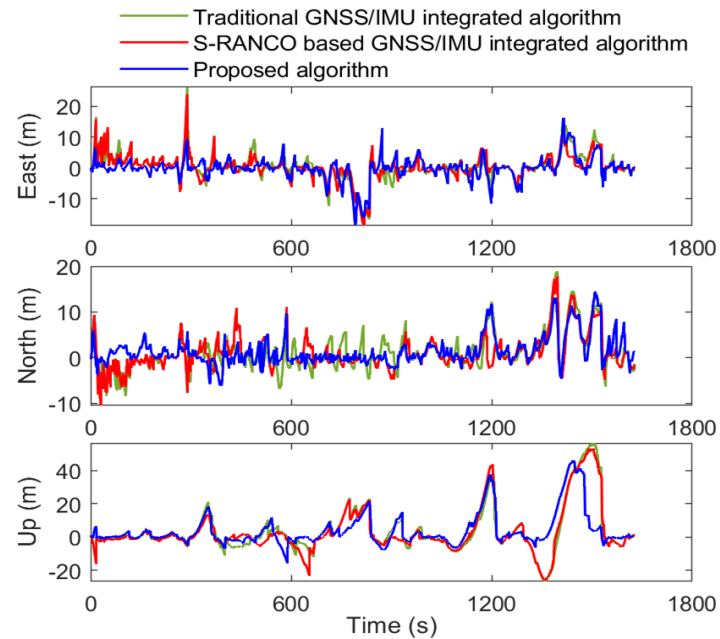
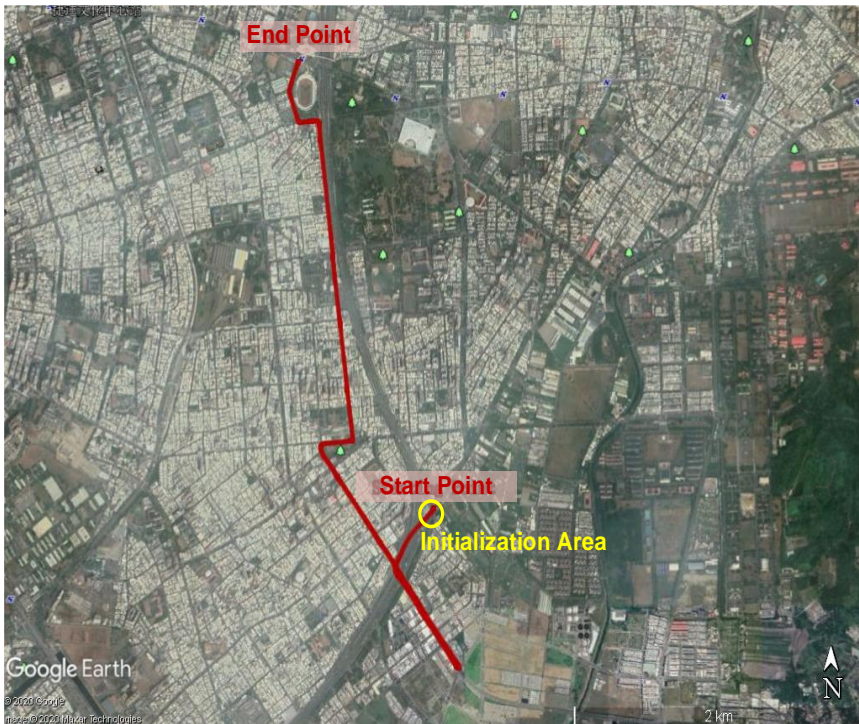


Methods	RMSE	Without FDE	S-RANCO based GNSS/IMU integrated algorithm	Proposed algorithm
Velocity (m/s)	East	0.37	0.37	0.35
	North	0.36	0.35	0.34
	Up	0.28	0.29	0.27
Attitude (degree)	3D	0.58	0.59	0.56
	Roll	0.483	0.493	0.452
	Pitch	0.392	0.409	0.604
Yaw	2.046	2.060	2.043	

Methods	Without FDE		S-RANCO based GNSS/IMU integrated algorithm		Proposed algorithm	
	RMSE	Improvement	RMSE	Improvement	RMSE	Improvement
Position (m)						
East	1.81	6.6%	1.69		1.70	6.1%
North	3.59	6.7%	3.35		3.34	7.0%
Up	4.88	-6.4%	5.19		3.88	20.5%
Horizontal	4.02	6.7%	3.75		3.75	6.7%
3D	6.32	-1.3%	6.40		5.39	14.7%

Test Case 2: deep urban environment

BDS/GNSS, Novatel PwrPak7, 1 Hz
IMU, Bosch BMI055, 50 Hz



Methods	RMSE	Without FDE	S-RANCO based GNSS/IMU integrated algorithm	Proposed algorithm
Velocity (m/s)	East	0.51	0.48	0.51
	North	0.49	0.48	0.49
	Up	0.42	0.40	0.40
Attitude (degree)	3D	0.82	0.79	0.81
	Roll	0.734	0.756	0.766
	Pitch	0.233	0.217	0.228
Yaw	8.477	9.045	8.949	

Methods	Without FDE	S-RANCO based GNSS/IMU integrated algorithm		Proposed algorithm	
Position (m)	RMSE	RMSE	Improvement	RMSE	Improvement
East	4.35	4.25	2.3%	3.73	14.3%
North	4.07	5.47	-34.4%	3.86	5.2%
Up	14.64	12.08	17.5%	10.98	25.0%
Horizontal	5.96	6.93	-16.3%	5.37	9.9%
3D	15.81	13.93	11.9%	12.22	22.7%

Conclusions

- ✓ Improving the BDS/GNSS data quality in urban areas can be benefit for many ITS applications.
- ✓ The proposed **machine learning based BDS/GNSS quality control methods can** effectively improve positioning accuracy.
- ✓ The proposed pseudorange error correction based methods can achieved a **70%** positioning accuracy improvement in static mode and a **60%** improvement in dynamic mode.
- ✓ The IMU-aided multiple BDS/GNSS fault detection and exclusion algorithm can provide an positioning accuracy improvement of **15-23 in urban areas**.

Future Works

- **Adaptive multi-sensor fusion strategy based on BDS/GNSS quality control will be carried to support more ITS applications**