



CENTRE NATIONAL D'ÉTUDES SPATIALES

RECENT SPACE DEBRIS MITIGATION ACTIVITIES IN FRANCE

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BACKGROUND

- **Particularity of the GEO orbit: unique resource**
- **Need to protect and to keep available orbital positions**
- **Mitigation measures are necessary**
- **Rules are being prepared by agencies**
- **End of life operations already performed by some operators**
 - **dialog between operators and agencies necessary**
 - **workshop organized by CNES**

OBJECTIVES OF THE WORKSHOP

- **Objective 1: To inform operators on regulatory issues under preparation:**
 - ◆ to convince,
 - ◆ to support, encourage their implementation
 - ◆ to be prepared to future evolution

- **Objective 2: to get feed-back from operators having performed end of life operations**
 - ◆ to highlight implementation difficulties
 - ◆ to update the rules when necessary

- **Workshop took place on January 27, 2010 at CNES's Headquarters in Paris**

- **More than 60 participants represented:**
 - ◆ **Administrations: French Ministry of Defence**
 - ◆ **European space agencies: BNSC (STFC), CNES, DLR and ESA**
 - ◆ **Satellite operators: Eumetsat, Eutelsat, France Telecom, Hispasat, Inmarsat, SES-Astra, Paradigm Services**
 - ◆ **Launch operator: Arianespace**
 - ◆ **Industry: Thales Alenia Space, EADS Astrium, Astrium Space Transportation, Satel Conseil, Atos Origin**
 - ◆ **Insurance companies: Hiscox, AXA, Marsh**
 - ◆ **University : Cranfield University, Université de Bretagne**

Objective 1: information of operators and industry

- **general overview of the situation in GEO: population, reorbiting practices**

- **regulatory issues discussed at different levels:**
 - ◆ **Practical implementation of the UK's Outer Space Act: GEO satellites**
 - ◆ **French law on space operations, technical regulations**
 - ◆ **DLR Technical Standards and Implementation Practice**
 - ◆ **Status of ISO standards development**
 - ◆ **IADC guidelines and the GEO disposal orbit: key parameters, future scenarios, and collision risks**

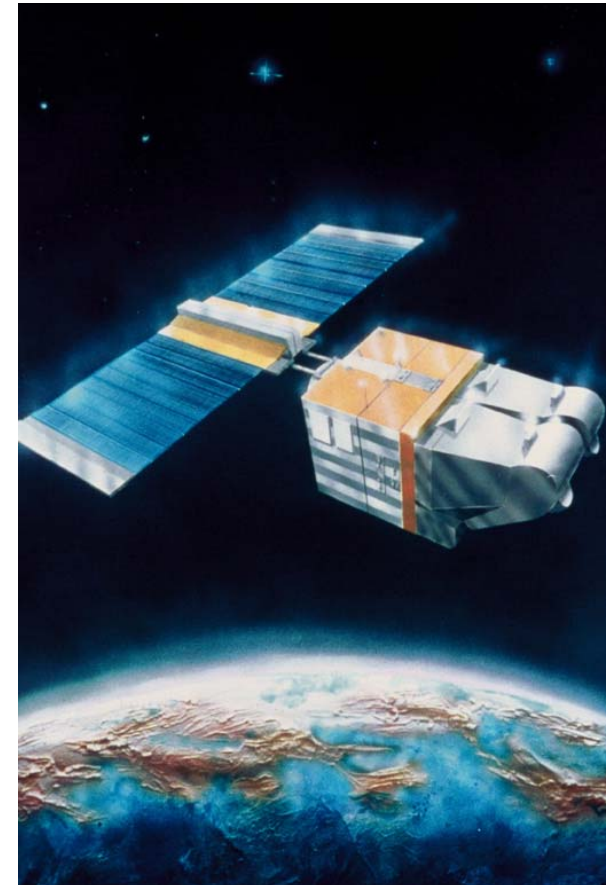
Objective 2: feed-back from operators having performed end of life operations

- **Long Term sustainability of space activities**
- **Eutelsat model of the effective cross section of reorbited satellites**
- **Long term evolution of Telecom 2B**
- **Thales Alenia Space experience on propulsion during reorbiting operations**
- **Telecom 2C reorbiting operations**
- **Eurostar end of life operations and strategy**
- **Paradigm End Of Life Operations Experience**
- **Approach for Meteosat-6 re-orbiting**

MAIN CONCLUSIONS

- **The rate of successful disposal is slowly increasing:**
 - ◆ 29% in 1997-2002
 - ◆ 57% in 2003-2009
- **Legal systems are implemented to reinforce application**
- **Accurate estimation of remaining propellant is difficult:**
 - ◆ Different methods exist with different results and accuracy
 - ◆ key issue to decide disposal operations
- **Passivation is a complex process:**
 - ◆ Time limitation due to ground station visibility (drift orbit)
 - ◆ Instability of attitude control due to gas bubbles in the propulsion lines
 - ◆ Need to keep control until the end for electric passivation and satellite switch-off

- **SPOT 2 launched by Ariane 4 on 21 January 1990**
- **Earth observation satellite**
 - ◆ Heliosynchronous orbit:
 - ◆ altitude 825 km
 - ◆ inclination 98 degrees
- **Main characteristics:**
 - ◆ Mass 1900 kg
 - ◆ dimensions: 2 x 2 x 4.5 m
 - ◆ Solar panel span 8 m
 - ◆ Hydrazine propulsion system
- **3-axis stabilization**



INTRODUCTION

■ Objectives:

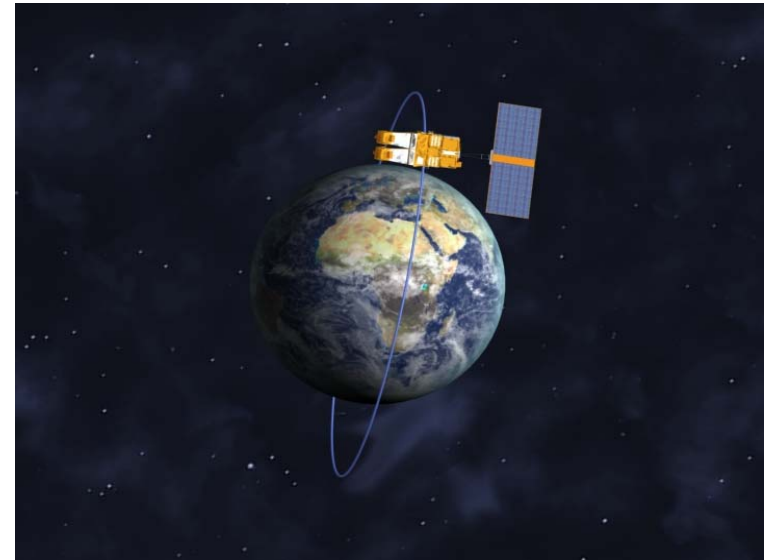
- ◆ Orbital lifetime lower than 25 years
- ◆ Passivation

■ Experience Spot 1 November 2003

■ Initial orbit

- ◆ Perigee altitude : 813.7 km
- ◆ Apogee altitude : 830.8 km

■ Hydrazin mass before de-orbiting: 60.3 kg



STRATEGY

■ First phase:

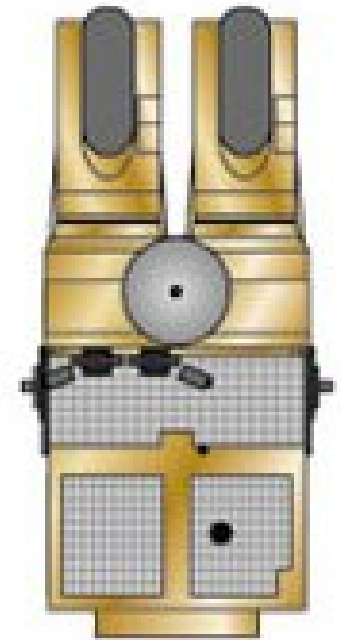
- ◆ Spot 2 orbit lowered 15 km below the operational orbit to avoid any collision risk with Spot 4 and 5
- ◆ positioning of the apogee above the Toulouse ground station

■ Second phase:

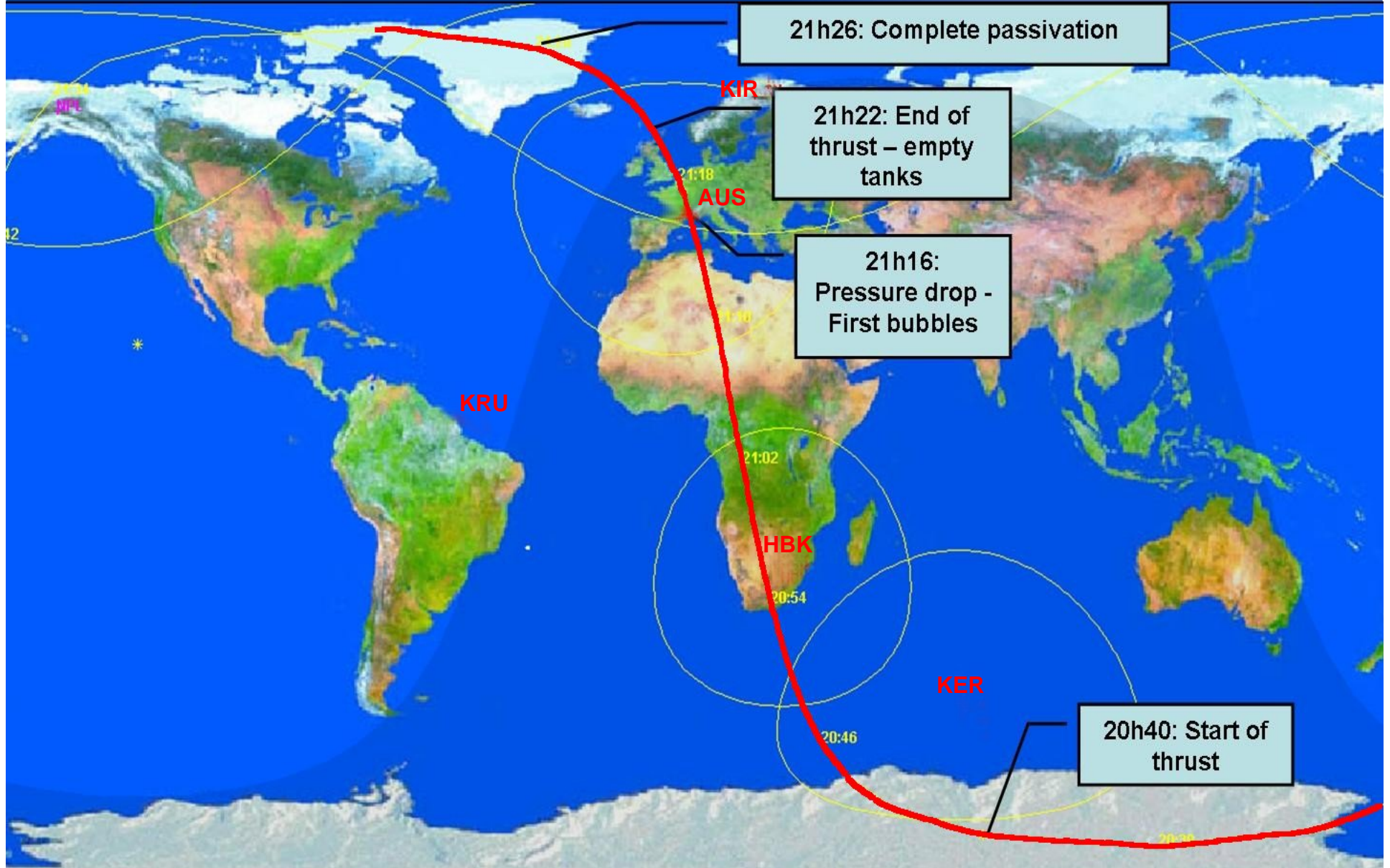
a series of 8 apogee manoeuvres (1000 s each) to decrease the perigee altitude

■ Third phase:

large last manoeuvre (2100 s) to decrease the perigee and to empty the tanks, passivation of the satellite

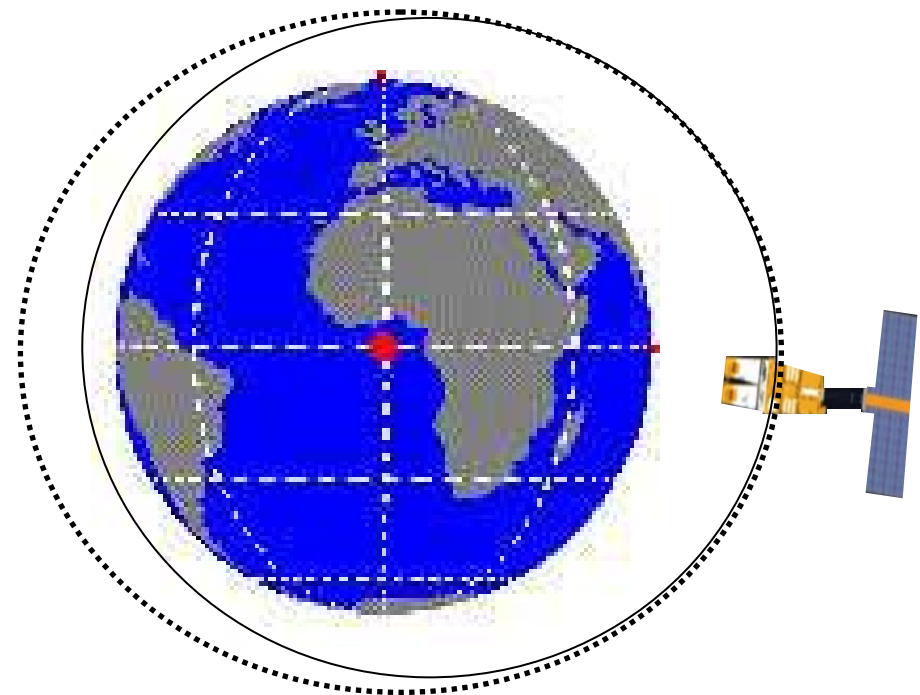


SPOT2 END OF LIFE OPERATIONS



FINAL STATUS

- **Orbit:**
 - ◆ Perigee altitude: 570 km
 - ◆ Apogee altitude: 796 km
- **Remaining orbital lifetime:**
 - ◆ Between 23 and 25 years
- **Telemetry confirmed fuel exhausted**
- **Battery disconnected**
- **Switch-off telemetry emitter**



- Launched by Ariane 44L on December 6, 1995
- Geostationary orbit at 3 ° East longitude
- Built by EADS Astrium and Thales Alenia Space
- Mass at launch 2275 kg, solar panels span 22 m
- Mission: telecommunications, television
- Operated by CNES on behalf France Telecom



SEQUENCE OF OPERATIONS

■ Phase 0 (18 October 2009):

- ◆ beginning of operations,
- ◆ satellite upside down to optimize thrust efficiency

■ Phase 1: exit from the GEO operational corridor:

- ◆ 1 apogee and 1 perigee manoeuvre to get out the GEO operational zone without crossing windows of neighbours
- ◆ → circular orbit 70 km above GEO

■ Phase 2: orbit raising

- ◆ sequence of apogee and perigee manoeuvres to progressively raise the altitude,
- ◆ IADC altitude obtained after 9 manoeuvres

SEQUENCE OF OPERATIONS

- **Phase 3: management of the tanks**
 - ◆ 8 additional manoeuvres
 - ◆ Use of one pair of tanks until bubble detection
 - ◆ Switch on the other pair of tanks when bubble detection

- **Phase 4: passivation:**
 - ◆ Sun pointing attitude,
 - ◆ use of thrusters to decrease pressure without attitude and altitude loss

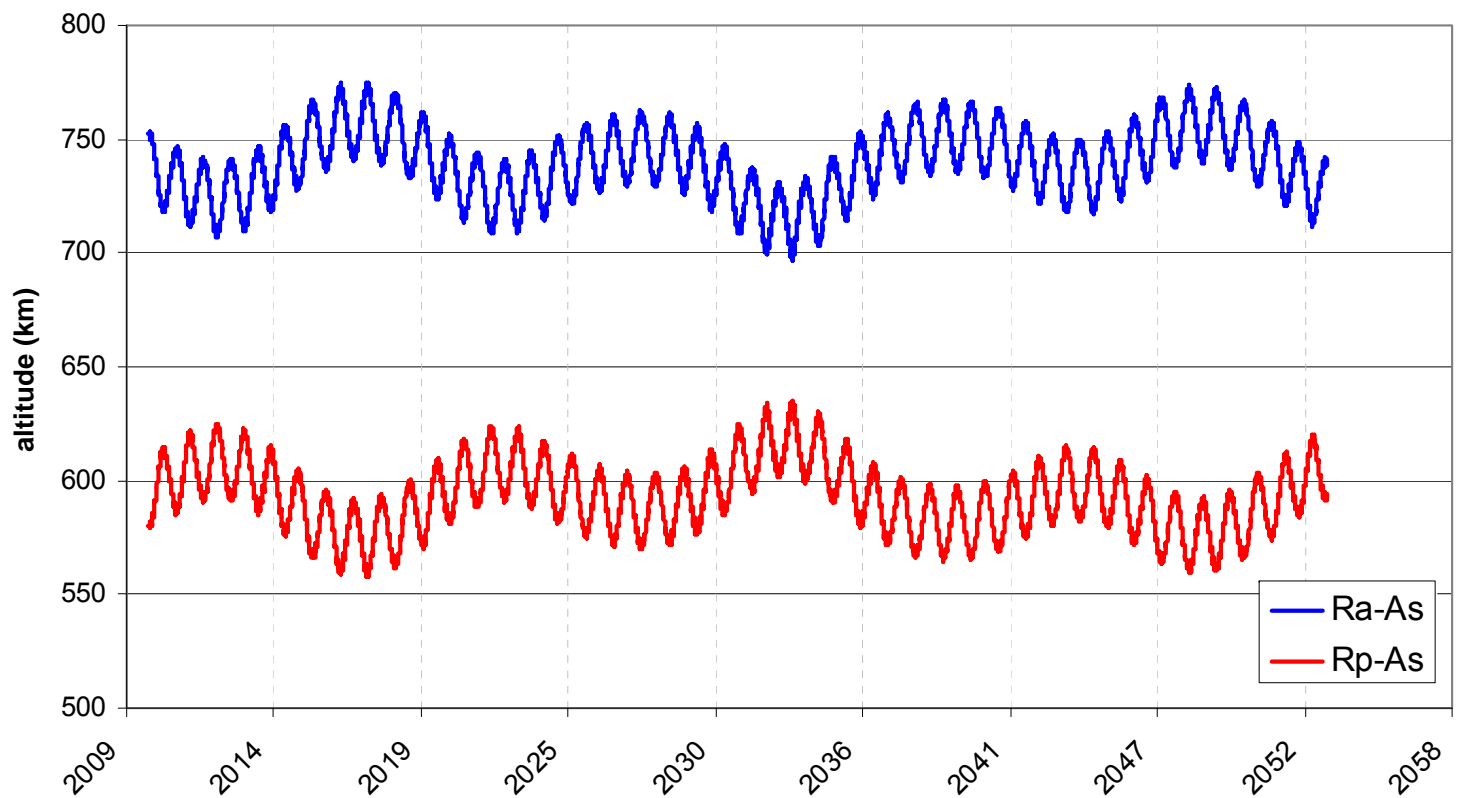
- **Phase 5: satellite switch-off**



FINAL STATE

- 17 East manoeuvres performed
- Duration of operations: 14 days (2 days off)
- 11.4 kg propellant mass
- Final orbit: 580 x 750 km above GEO
- Eccentricity $2 \cdot 10^{-3}$
- 4 propellant tanks empty
- Compliance with IADC guidelines

Long term simulations of perigee and apogee altitude evolution



non interference with the GEO protected region

■ As space agency CNES

- ◆ Prepares the Technical Regulations associated with French Space Act
- ◆ promotes the application of the guidelines: workshop in Paris with industry and operators
 - Information on regulatory issues given by agencies
 - Feed-back from the operators based on their experience

■ As operator of national satellites CNES applies the end of life guidelines:

- ◆ to LEO satellites: SPOT 2 in July 2009 after SPOT 1 in November 2003
- ◆ to GEO satellites: TELECOM 2C in October 2009 after TDF1 and 2, TELECOM 1A and C, TELECOM 2A and B