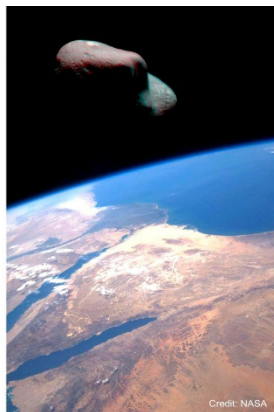


## Mitigation of the NEO Impact Threat



Credit: NASA

**Alan Harris** (Coordinator NEOShield)  
and **Line Drube**

Presented by: **Line Drube**  
**DLR Institute of Planetary Research,**  
**Berlin**



Deutsches Zentrum  
für Luft- und Raumfahrt e.V.  
in der Helmholtz-Gemeinschaft



# The NEOShield Consortium



Financed via European Commission's FP7 funding programme, 2011 Call:

**“Prevention of impacts from near-Earth objects (NEOs) on our planet”**

Total NEOShield funding = **5.8 million euro**. Duration = **3.5 years**. Start **Jan 2012**

**NEOShield's primary aim:** investigate in detail the three most promising mitigation techniques: ***Kinetic Impactor, Blast Deflection, Gravity Tractor.***

Partner organisations:
German Aerospace Center (DLR), Berlin <i>Coordinating partner</i>
Observatoire de Paris (LESIA and IMCCE)
Centre National de la Recherche Scientifique (Observatoire de la Côte d'Azur)
Open University
Fraunhofer – Ernst-Mach-Institut
Queen's University Belfast
Astrium UK+ DE + FR ( <i>supervisory interface for technical work packages</i> )
Deimos Space
Carl Sagan Center, SETI Institute
TsNIIMash (Roscosmos)
University of Surrey



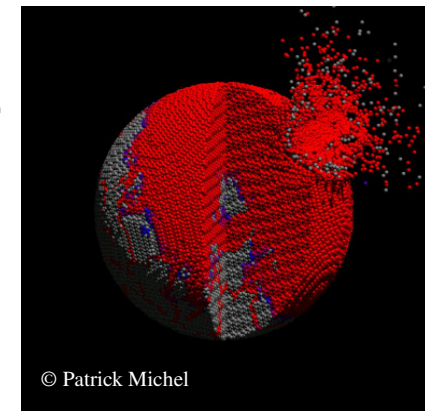


# The NEOShield Project: Science



Main scientific themes/tasks of the project:

- **Physical properties of NEOs:** Analyze properties from the point of view of mitigation requirements; estimate most likely properties of the next mitigation candidate; provide requirements for lab. impact experiments, modelling and demission target selection.
- **Lab. experiments on impacts** - into asteroid surface analogue materials; validation of impact modelling at small scales.
- **Numerical simulations:** Impact and momentum transfer modelling scaled to realistic NEO sizes.
- **Mitigation reconnaissance:** Determine requirements, strategy, instrumentation, for ground-based facilities and space missions. Verify that the NEO deflection was achieved.





# The NEOShield Project

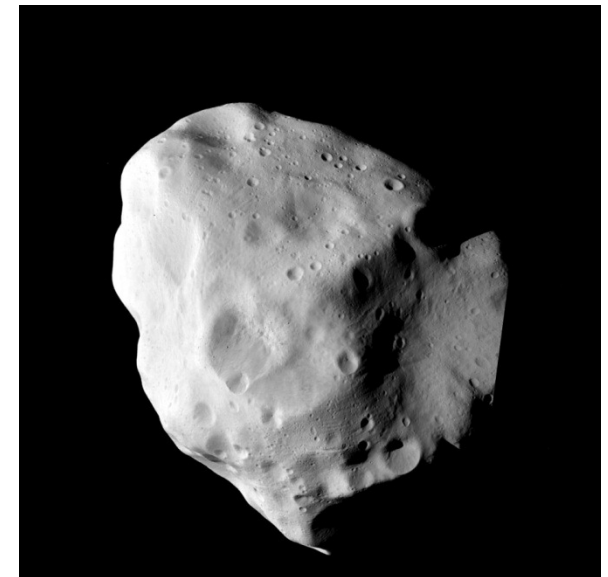
## Mitigation demonstration missions



Theoretical work, lab. experiments, computer modelling is essential, but must be followed up with attempts to change the orbit of a real NEO.

NEOShield tasks include:

- **Design demo missions:** Provide detailed designs of technically and financially realistic missions to demonstrate the effectiveness of mitigation techniques. Investigate mission funding and implementation options.
- **Choose suitable mission targets:** Identify and characterize suitable target NEOs for mitigation demo missions.
- **Decision-making tools:** Develop software to aid decision-making for an effective response to the particular circumstances of an impact threat.



[www.NEOShield.net](http://www.NEOShield.net)



# Kinetic Impactor

A massive spacecraft impacts the asteroid at high relative velocity ( $v > 10 \text{ km s}^{-1}$ ), **thereby transferring momentum and modifying its orbit.**

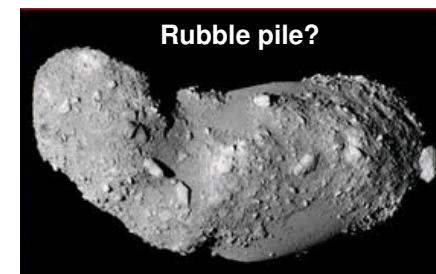
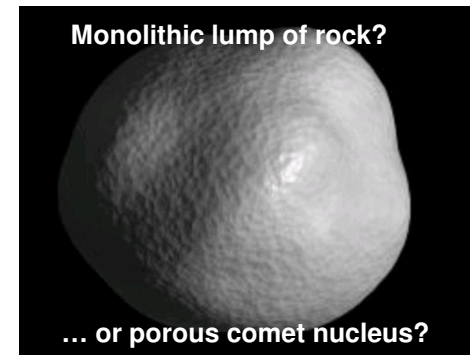
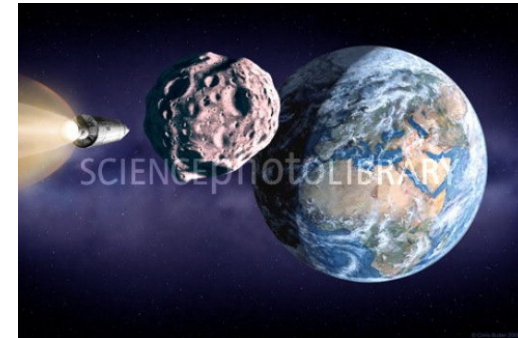
## Advantage:

The technology is straightforward and (almost) ready.

## Challenges:

- Autonomous guidance, navigation and control performance for a high speed impactor.
- How does **impactor momentum transfer** depend on the bulk density, porosity, mineralogy, internal structure, etc. of the target NEO?
- How much impactor kinetic energy may be wasted in fragmentation and restructuring?

**Momentum transfer in the case of a porous body is significantly less efficient. Rubble piles?**

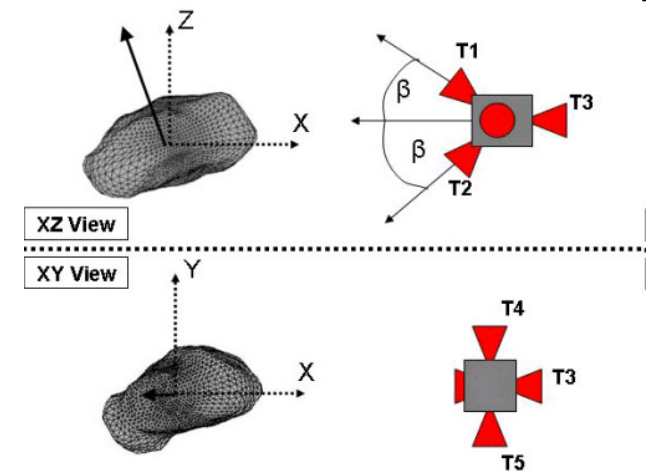


# Gravity Tractor

A massive spacecraft positions itself close to the NEO and fires its thrusters so as to maintain a constant distance from the target. **The weak gravitational force between tractor and NEO acts as a tow-rope.**

**Advantage:** no contact with the NEO; **very little prior knowledge of physical properties required** (only mass, shape, rotation vector).

**Challenge:** requirements for autonomous spacecraft control procedures to manage hovering station keeping and maintain stability of the traction system over a long period of time in the (very nearby) presence of an irregular rotating mass.

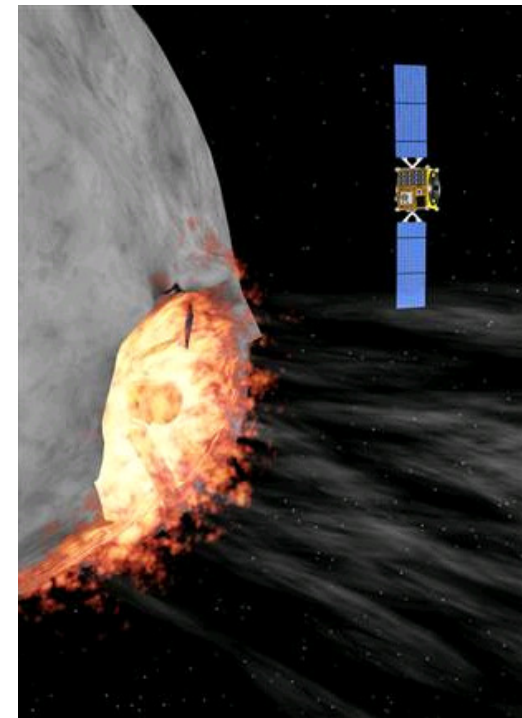


# Blast Deflection

Exploding a nuclear device near, on, or below the surface of the asteroid would transfer an impulse to the asteroid by vapourizing and ejecting material from the surface. Small objects ( $D < 100$  m) could be destroyed: the fragments should present less danger than the original object. Considered a method of last resort, in the absence of other alternatives.

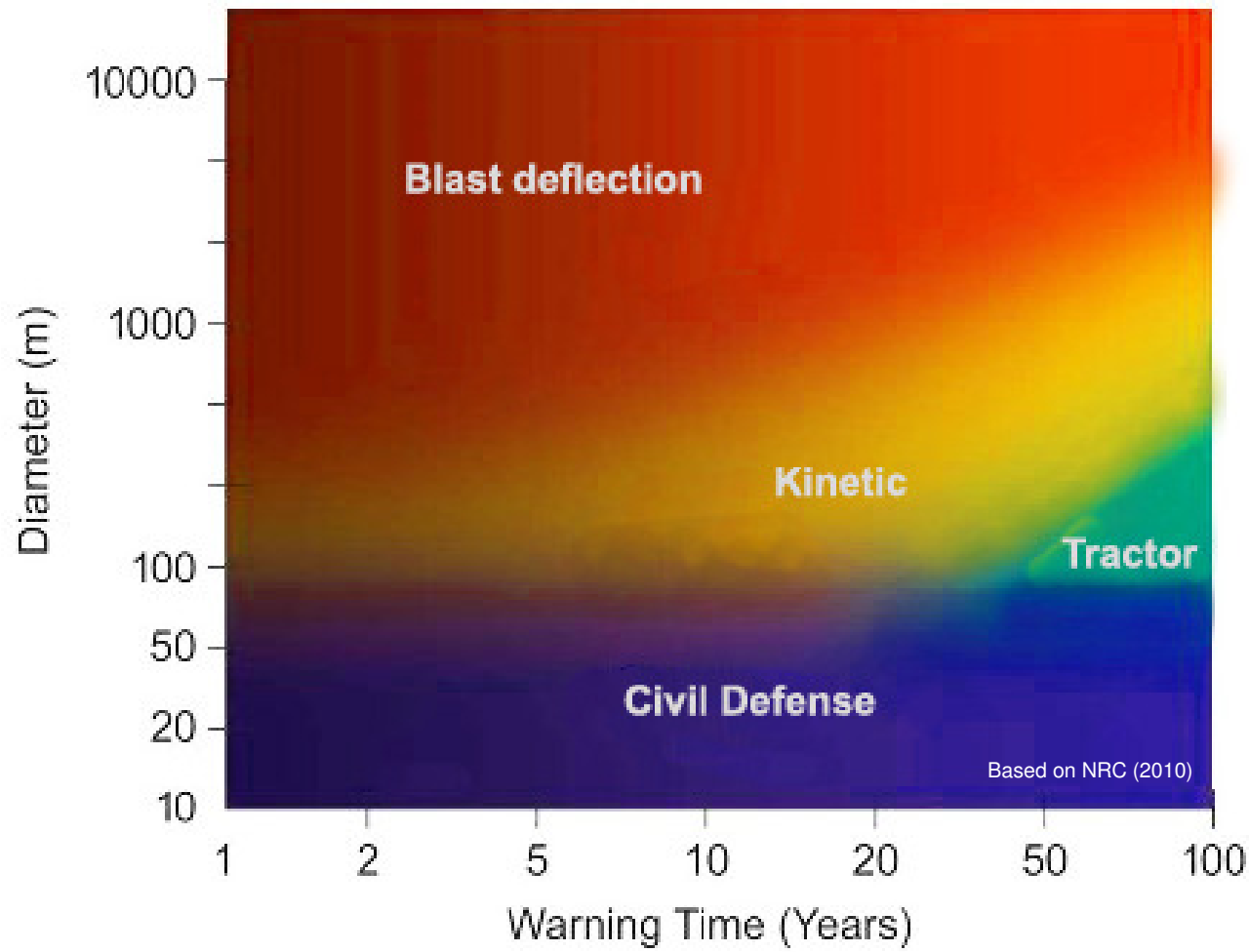
- **Advantage:** Provides the “biggest bang per buck” of all methods.

- **Challenges:** Requires knowledge of the internal structure and surface material. Disruption of large objects **may give rise to many hazardous impactors** instead of just one. Danger during launch. **Political issues.**





# Mitigation methods





# Mitigation demo-mission ideas

Example: ESA's **Don Quijote** Study 2005-2007

Two-spacecraft impactor mission:

*Sancho* (rendezvous)

*Hidalgo* (impactor)

Target size ~ 500 m

Objectives:

- **Pre-impact** (Sancho 7 months at target): Measure size, shape, bulk density, mass distribution.
- **Impact of Hidalgo**: Info. on regolith properties, internal structure.
- **Post-impact** (Sancho 3-4 months at target): Measure  $\Delta v$ , observe impact effects.



# Mitigation demo-mission ideas

Example: **AIDA / DART**

- currently at the feasibility study phase

1 or 2 spacecraft impactor mission:

**DART** (300kg impactor) NASA

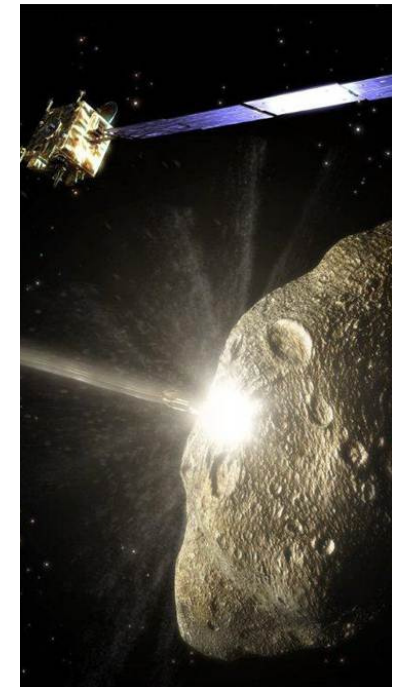
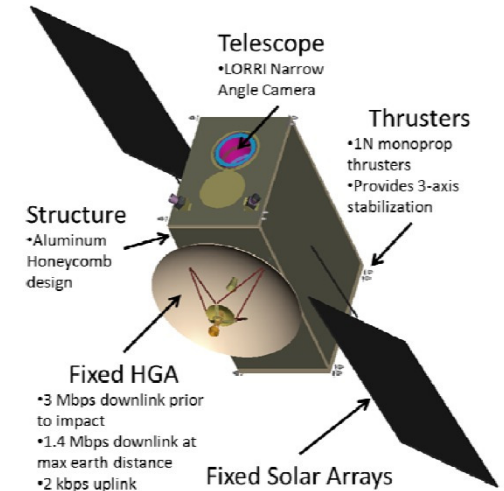
**AIM** (rendezvous) ESA

**Target:** 150 m moon of the binary NEO Didymos (mass of the moon ~9 million tonnes)

**Impact time:** 2022 during close Earth approach.

**AIDA (= AIM + DART)** would do a full characterization of the kinetic impact and of the asteroid and moon.

**DART without AIM** will use ground-based observatories to measure the 0.5-1% expected change in orbital period of the moon about the primary object. This will be a very basic demonstration of asteroid deflection for a low cost.

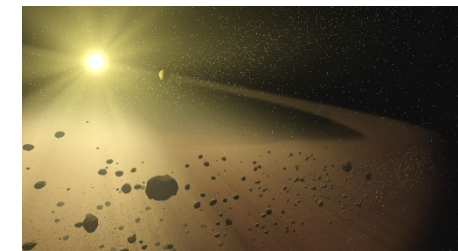
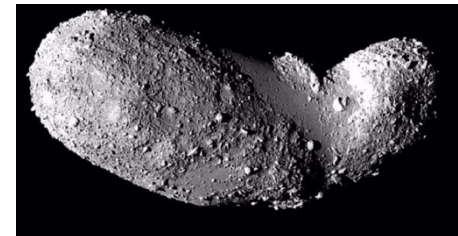


## NEOShield & The Space Mission Planning Advisory Group

**NEOShield tasks and those of the Space Mission Planning Advisory Group have much in common.**

Future work in this area should include:

- **Exploration of NEOs:** Further discovery and characterization of the NEO population is required. The population of small NEOs ( $D = 50 - 300$  m; expected about 200,000 objects) remains largely unexplored. We need to determine sizes, albedos, spin, mineralogy, shapes, densities, structures, porosities, frequency of binaries, frequency of rubble piles, etc. **Novel techniques specifically suited to very small asteroids should be investigated.**
- **Novel techniques for NEO mitigation:** The first hazardous NEO to trigger a space-borne mitigation action will probably be in the size range 50 m – 200 m (could destroy a large city or national region). **Alternative techniques to deal with very small asteroids should be investigated.**





## **In conclusion:**

- **The NEO impact threat is a truly global problem.**
- **Coordination of diverse competences and multiple technologies is vital.**
- **Collaboration among space agencies and international consortia is essential in providing a focused effort and an internationally agreed consensus of scientific and technical information in support of political decision-makers.**
- **We believe NEOShield is a good example of the type of scientific and technical collaboration that will be required to carry out the work of the Space Mission Planning Advisory Group.**