

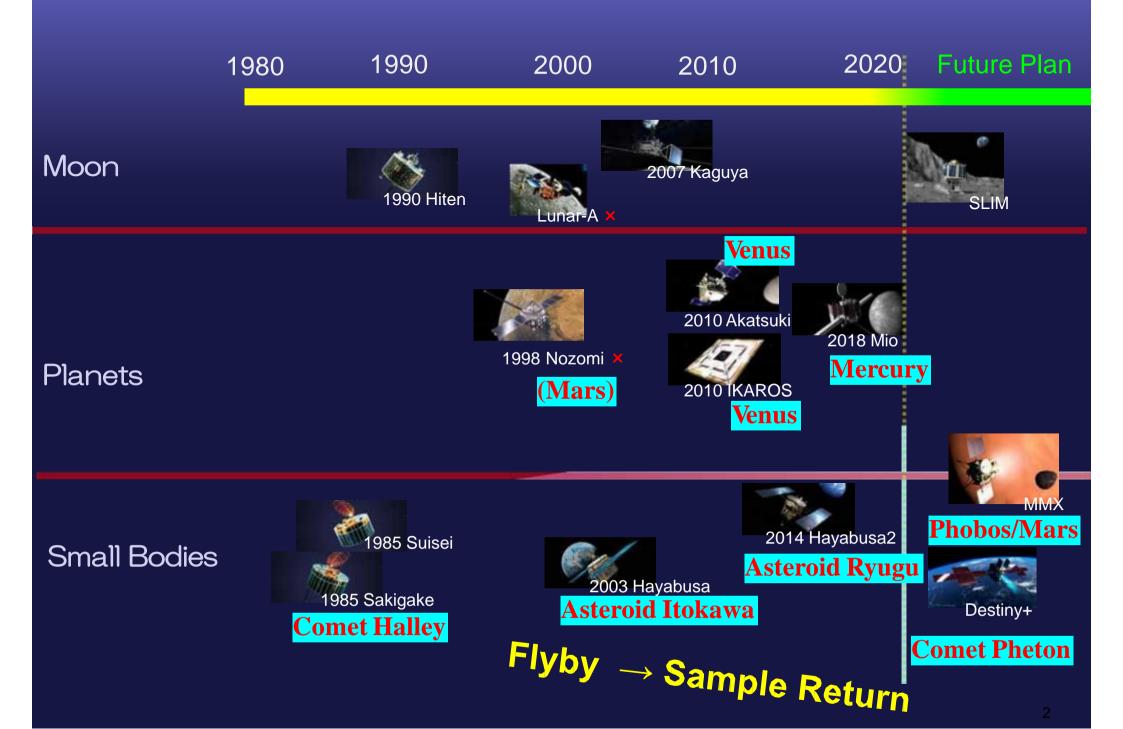
Achievements of Hayabusa2: Unveiling the World of Asteroid by Interplanetary Round Trip Technology

Yuichi Tsuda Project Manager, Hayabusa2 Japan Aerospace Exploration Agency

58th COPUOS, April 23, 2021



Lunar and Planetary Science Missions of Japan





Hayabusa2 Mission

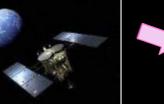


- Sample return mission to a C-type asteroid "Ryugu" \checkmark
- 5.2 billion km interplanetary journey.

Launch Dec.3. 2014







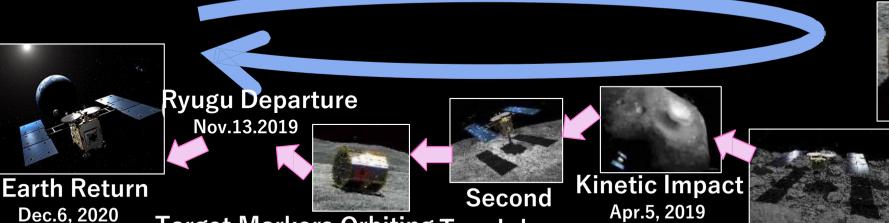
Target Markers Orbiting Touchdown





MINERVA-II-1 Deployment





Jul,11, 2019

First Touchdown Feb.22, 2019

MD[D VIp srvlxp #534<#

Oct.2, 2019

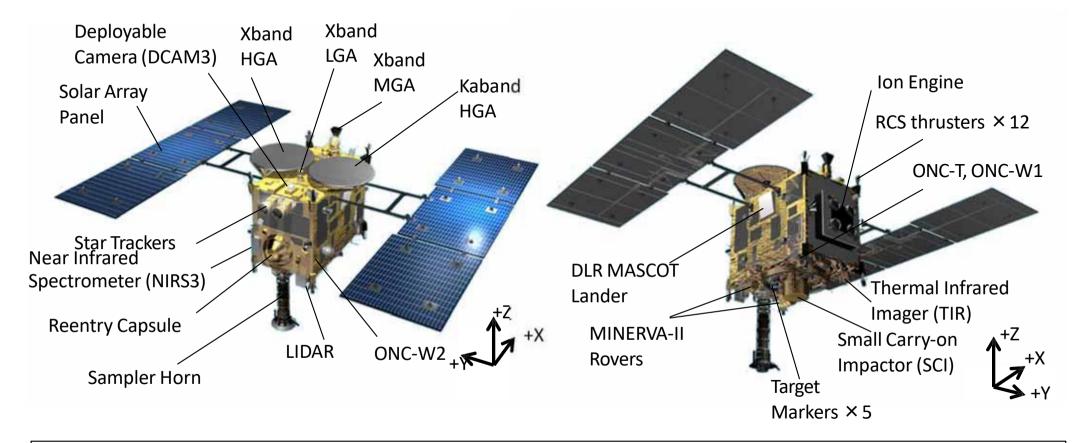
MINERVA-II-2 Orbiting

Sep.16, 2019



Hayabusa2 Spacecraft Overview





Launch Mass: 609kg Ion Engine: Total ΔV=3.2km/s, Thrust=5-28mN (variable), Specific Impulse=2800-3000sec. (4 thrusters, mounted on two-axis gimbal) Chemical RCS: Bi-prop. 20N thrusters × 12 (6 DOF maneuverability) Solar Array Paddle: 2.6kW @ 1 a.u. TT&C: X-band Uplink, X/Ka-band Downlink, 8-32Kbps, X/Ka RARR&DDOR capability



International Collaboration in Hayabusa2



Australia

200+ Japanese researchers, 100+ international researchers

USA



1. NASA

- Tracking and navigation support by JPL
- Asteroid observation
- Sample exchange with OSIRIS-REx mission

Europe



1. DLR (Germany)

- Provision of MASCOT Lander
- •Tracking support thr ESA
- •Drop tower experiment

2. CNES (France)

• Provision of instruments aboard MASCOT



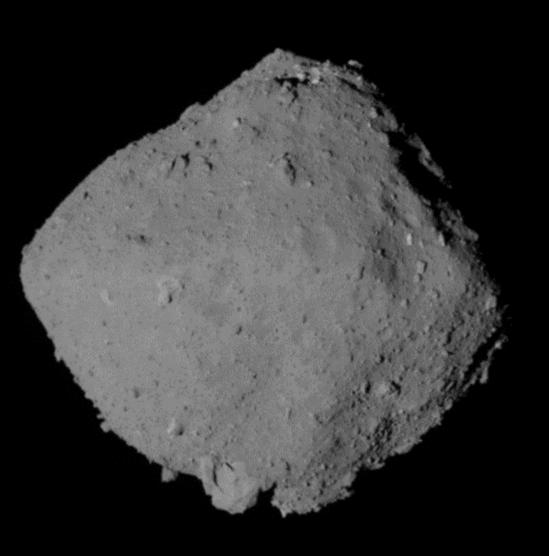




- 1. Australian Space Agency (ASA)
- Landing authorization
- 2. Department of Defense (DOD)
- Provision of Landing site

Arrival at Ryugu on June 27, 2018

- Top shape with a very circular equatorial bulge
- Spectrum type: Cb
- Diamter: ~900 m
- Mass: \sim 450 million ton
- Obliquity: $\sim 8^{\circ}$
- Rotation period : P = 7.63 hours
- Reflectance factor (v-band) : 0.02
- Terrain: Very bumpy

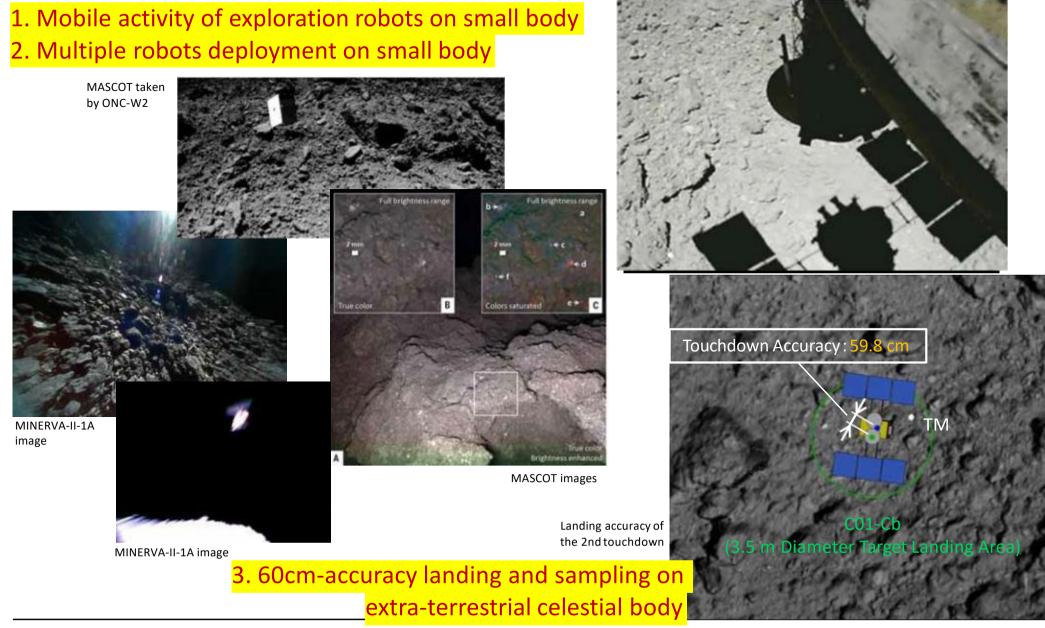




Accomplishments of Hayabusa2 (1/2)



CAM-H image at the 1st touchdown





Accomplishments of Hayabusa2 (2/2)



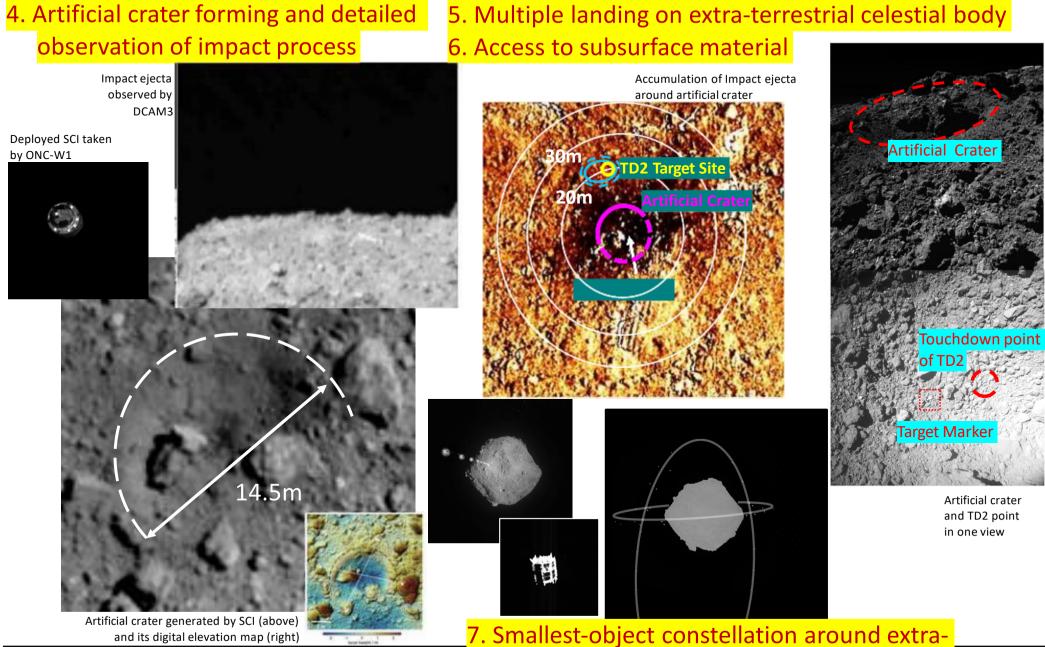


image credit: JAXA

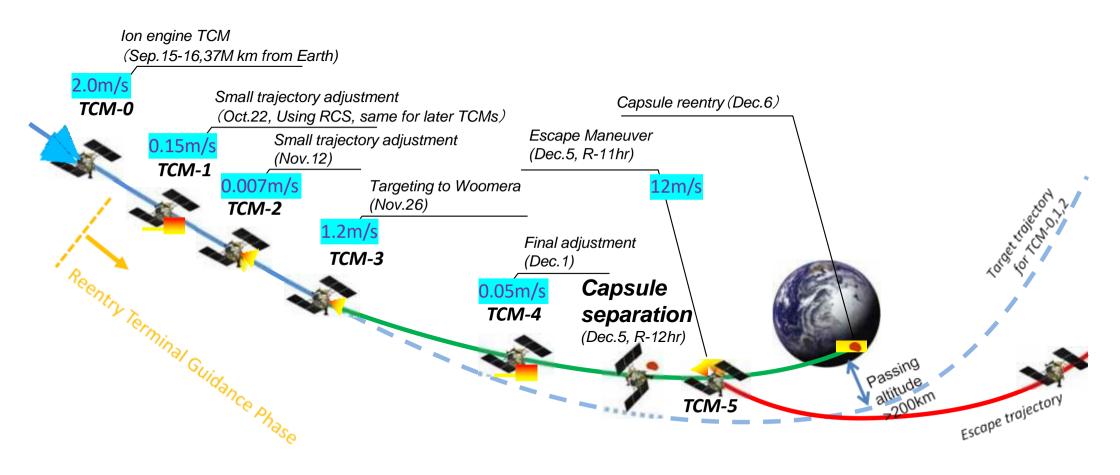
Smallest-object constellation around extra terrestrial celestial body



Reentry Terminal Guidance Phase



- 5 TCMs in the last 2 months before Earth return.
- The SRC was separated 12 hrs before reentry.
- The spacecraft diverted from the reentry trajectory 11 hrs before reentry.



Hayabusa2 has returned!

Coober Pedy, Australia, Dec.6, 2020, 2:28:48-2:29:11JST (Altitude 80~50km)



Sample Return Capsule recovery











Dec.6 (JST)

- 02:28 SRC reentry
- 02:32 SRC beacon signal detected
- 02:54 SRC landed (loss of beacon signal)
- 04:47 SRC found
- 08:03 SRC arrived at Quick Look Facility
- 11:13 Fore-heat shield found
- 12:31 Aft-heat shield found
- Dec.7

Dec.8

- 22:30 SRC shipped to Japan
- 11:27 SRC carried into curation facility

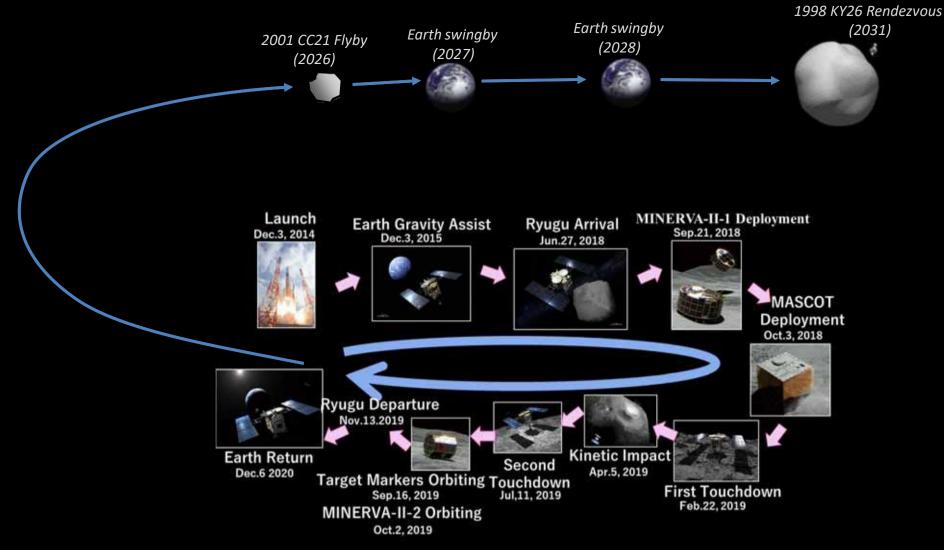
<mark>57hr!</mark> (requirement 100hr)

Ryugu samples found in the sample container! Sample yield : 5.4g (requirement: 0.1g)



Hayabusa2 Extended Mission

- ✓ Multi-swingby + solar electric propulsion
- ✓ Pursuing for *Planetary Defense* technologies and sciences
- ✓ High speed flyby of asteroid 2001 CC21
- ✓ Rendezvous to fast rotator asteroid 1998 KY26







Significance of Hayabusa2



Science

Space Exploration Engineering

Planetary Defense

Planetary Resource

Hayabusa2 is pushing forward the boundaries of small body surface activity ACCESS / ROVING / SAMPLING / IMPACTING Launch Mass: About 4,000 kg Mission Duration: About 5 Years Launcher: H3 Launch Vehicle

Target Launch Year: JFY2024

Martian Moons eXploration

The sample return mission from the Martian moon, Phobos

Thank you for your attention

Realica