QSS-CUS-5168



### Time and Coordinate System for QZSS (Quasi-Zenith Satellite System) PNT (Positioning, Navigation and Timing service)

# 4 December 2017 QZS System Services Inc. (QSS)

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# 1. QZSS PNT Time system(QZSST)



The <u>QZSS</u>(Quasi-Zenith Satellite System) <u>PNT</u>(Positioning, Navigation and Timing service) time system, called "QZSST", is defined below.

#### 1. Definitions

- a. Length of 1 second The length of 1 second in QZSST shall be identical to the International Atomic Time (TAI).
- b. Offset between QZSST and TAI QZSST shall be delayed from TAI by 19 seconds.
- c. Starting point of week number for QZSST The starting point of the week number for QZSST shall be the same as the GPS time system (GPST), which is 0:00 am (UTC) on January 6, 1980.

#### 2. Navigation message reference time

The parameters relating to time such as the SV clock parameter, mean motion and UTC parameters shall all be based on QZSST.

### 1-1. Time Transfer Between QZSST and GPST(1/2)

 QZSST is almost aligned to GPST so that the difference between QZSST and GPST are nominally within 2ns.

Figure 1 shows the schematic view of QZSST and GPST.

- QZSSRT (QZSS reference time system) is defined by an ensemble average of receiver(\*1) clock bias which is constrained to zero on the system equation.
  - \*1: The QZSS has 4 "time reference monitor stations" which consists of an Atomic Hydrogen Maser.
- ② All satellite and receiver clock bias is corrected by the estimated residual clock bias between QZSSRT and GPST.
- ③ The difference between QZSSRT and GPST becomes almost zero, so the GGTO is broadcast always as zero.

### 1-1. Time Transfer Between QZSST and GPST(2/2)



#### Figure 1: Schematic View of QZSST and GPST

## 1-2. Time Transfer Between QZSST and UTC

- UTC parameter is broadcasted on L1C/A, L1C, L2C and L5.
- Figure 2 shows the schematic view of QZSST and UTC(NICT).
  - ① Each monitor station's clock bias from UTC(NICT) is estimated using GPS common view method.
  - ② So the difference between UTC(NICT) and QZSST can be derived using Monitor station's clock bias from UTC(NICT) and QZSST.



Figure 2: Schematic view of QZSST and UTC(NICT)

# 2. QZSS PNT Coordinate system



The QZSS/PNT coordinate system(JGS) is defined as follows. Thus the frame uses the IGb08. The difference between ITRF and JGS is nominally within 20mm.

#### Definitions

The QZSS/PNT coordinate system is the same as the International Terrestrial Reference System (ITRS) stipulated by IERS, and conforms to the International Terrestrial Reference Frame (ITRF).

- Origin: Mass center of the earth
- >Z-axis: the IERS Reference north pole
- **X-axis:** Intersection of the IERS Reference Meridian (IRM) and the equatorial plane
- Y-axis: Completes a right-handed, Earth-centered, Earth-Fixed (ECEF) orthogonal coordinate system

## 3. Evaluation of QZSS/PNT Data(1/2)



QZSS/PNT is now under trial service.
 GGTO error is within 0.604ns (95%). (Figure 3)
 The evaluated error of "UTC(NICT)-GPST(PNT)" is good relation to "UTC(NICT)-GPST(BIPM). (Figure 4)



[From 2017/5/18 to 2017/6/22]

Figure 3: GGTO ERROR

Figure 4: Difference between UTC(NICT) and GPST

# 3. Evaluation of QZSS/PNT Data(2/2)



 Difference between ITRF and JGS is within 8.4mm (95%). (Figure 5)



#### [From 2017/5/18 to 2017/6/22]

Figure 5: Difference between ITRF and JGS



Considering ICG WG-D recommendation #23 and IGS White Paper, Satellite Property Information (SPI) and Operational History Information (OHI) for each QZS SV are published on our web-site.

March, 2018(Target)

March, 2018(Target)

http://qzss.go.jp/en/technical/qzssinfo/index.h

#### Current status on the SPI and OHI Publication

SPI

Published(2017.10.6)

Published(2017.7.4),

Updated(2017.11.30)

Published(2017.11.30)

Published(2017.11.30)

**Satellite** 

QZS-1

QZS-2

QZS-3

QZS-4

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I Publication		QZS-1	SPI-QZS1 (in preparation)	-	OHI-QZS1 (in preparation)	-		
		QZS-2	SPI-QZS2	2017.7.4	OHI-QZS2 (in preparation)	-		
		076.3	SPI-QZS3 (in preparation)	-	-	-		
OHI			SPI-QZS4 (in preparation)	-	-	-		
Published(2017.10.6)			other information, pleas QZS satellites is provider request.	e contact thro d to the reque	ough the following URL. The estors individually accordin	e further g to the		
December, 2017					<b>a</b> 95%	•		
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# Satellite Property Information(SPI)

- Contents
  - 1. Reference Frame
- 2. Attitude Law
- 3. Mass and Center of Mass
- 4. Navigation Antenna Phase Center Corrections
- 5. Geometry
- 6. Satellite dimension
- 7. Optical Property
- 8. Laser Retro Reflector Location
- 9. Differential Code Bias
- 10. Antenna Transmit Power

✓ Current status : In preparation

 $QZS-1 \rightarrow Not Published$ (The material can't be obtained any more.)

QZS-2 to QZS-4  $\rightarrow$  Under analysis

✓ Plan: These information aim at publish on January 2018.



#### Optical Property

#### > Activity Status on Data acquisition of Optical and Thermal property

Property	Acquisition Data	Disclosure/ Non-disclosure	Status				
Optical	Absorption coefficients(Ca) Specular reflectivity(Cs) Diffuse reflectivity(Cd) of each material	Disclosure	Under measuring and analysis (MLI <sup>(*2)</sup> ,OSR <sup>(*3)</sup> and so on)				
	BRDF <sup>(*1)</sup>	Non-disclosure	Scheduled to be measured after acquisition of the above optical property data				
Thermal	Surface temperature	Non-disclosure	Under consideration of possibility of installing sensor necessary for measuring surface temperature on QZS-1R. (infrared camera, temperature sensor and so on)				
	Infrared emissivity	Disclosure					

**\*1** : BRDF (Bidirectional Reflectance Distribution Function)

**\*2** : MLI (Multi Layer Insulation)

**\*3**: OSR (Optical Solar Reflector)



#### Operational History Information(OHI)

#### Contents

- 1. Attitude Change history
  - mode/start-end time Since January 2018, the plan and the result information will be included.
- 2. Orbit maintenance maneuver history (Planned value only)
  - time/duration/delta-V/direction
- 3. Estimated mass history

#### Sample

The history information of QZS-1 operation												
date	event		attitude change (*3)			orbit maintenance maneuver						
		mode	start(UT)	stop(UT)	No.	start	stop	duration	ΔVx	∆Vy	ΔVz	(*6)
2011/11/9 orbit m a intenance m aneuver (*1)				#1	2011/11/9 1332	2011/11/9 1335	0 0 3 1 1	-0.877	-0.034	-0.105	2281	
				#2	2011/11/9 1830	2011/11/9 1836	0 06 48	1.625	-0.029	-0.07		
					#3	2011/11/9 2331	2011/11/9 2336	0 05 15	-1.449	-0.057	0.04	
2012/1/4	un bad ng											2281
2012/3/3	change of attitude mode	YS→0N	9 54									
2012/3/7	unbading											2281
2012/4/16	change of attitude mode	0 N → Y S	7 39									
2012/5/2 orbit maintenance r					#1	2012/5/2 1 59	2012/5/2 200	0 01 47	-0.492	0.007	-0.057	
	orbit maintenance maneuver(*1)				#2	2012/5/2 657	2012/5/2 7 02	0 0 4 4 4	1.275	-0.006	-0.041	2277
					#3	2012/5/2 11 57	2012/5/2 1202	0 05 24	-1.488	0.021	0.049	

# 5. Summary



- Definition of QZSS/PNT's time system(QZSST) and coordinate system(JGS) was explained.
  QZSST is nominally aligned to GPST within 2ns.
  JGS is nominally aligned to ITRF within 20mm.
- QZSS/PNT started trial service from March 28th 2017, and these system errors are well sustained within target values.
- Web-site of QZSS Satellite information for Precise Orbit Determination is introduced.

# Thank you for your attention.

# For more information, please visit our web site <a href="http://gzss.go.jp/en/">http://gzss.go.jp/en/</a>



- A large circle illustrated "Q" as Quasi-Zenith Satellite System
- Green and blue circle composes 8 shapes: the coverage area of QZSS and they are represented earth and satellite.
- Blue line symbolized precise positioning information as well as enlargement of brand new service to society.
- Color of green stands for environment and safety, and blue stands for space and technology.

#### Backup: Satellites system

#### 4-Satellite Constellation;

#### 3 QZ Orbit, 1 Geostationary Orbit



#### Backup: QZSS Monitor Stations in the World

