

Galileo I/NAV Improvement

D/NAV

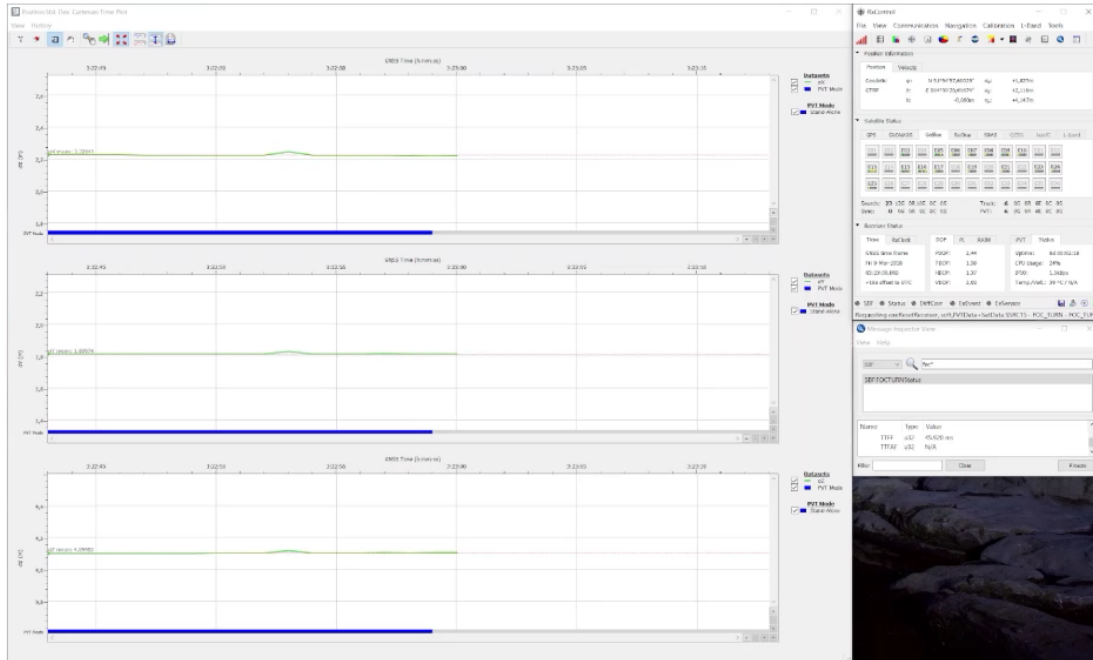
UNOOSA ICG Annual Meeting – October 2022

- **I/NAV Improvement – User Benefits**
- I/NAV Improvement – Technical Content
 - Reduced Clock and Ephemeris Data
 - Reed Solomon coded Clock and Ephemeris Data
 - Secondary Synchronization Pattern
- User Exploitation and Performance
- Deployment Status

I/NAV Improvement - User Benefit

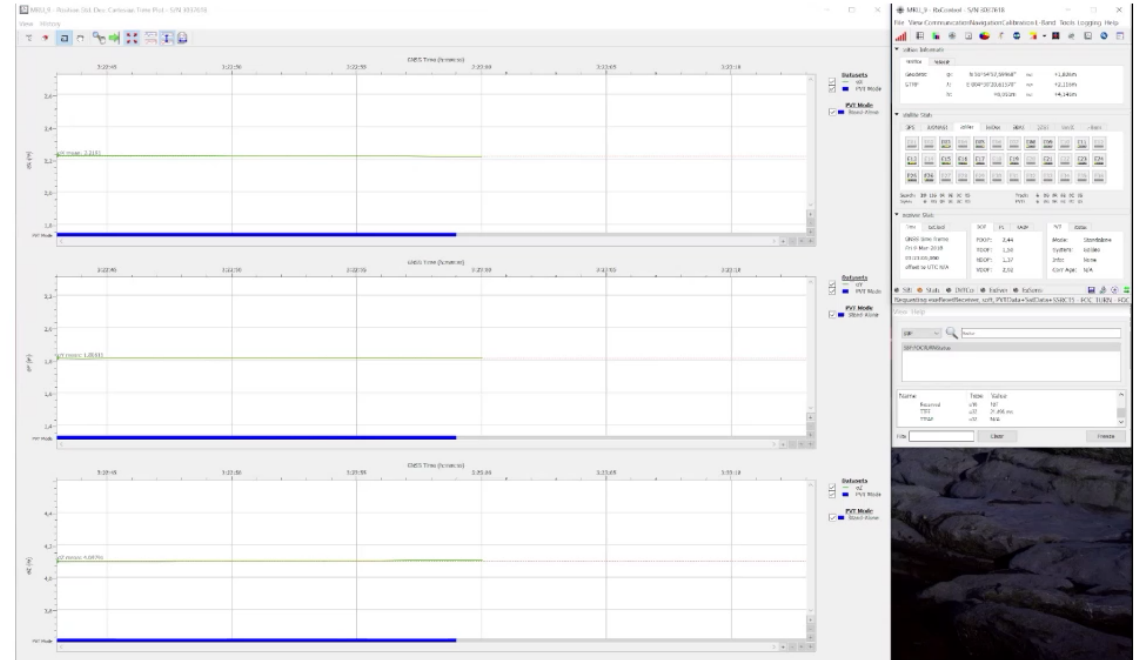
Position error (x,y,z)
with I/NAV Legacy

Step 1: Rx Reset



Position error (x,y,x) with
I/NAV+ Improvements

Step 1: Rx Reset



- I/NAV Improvement – User Benefits
- **I/NAV Improvement – Technical Content**
 - Reduced Clock and Ephemeris Data
 - Reed Solomon coded Clock and Ephemeris Data
 - Secondary Synchronization Pattern
- User Exploitation and Performance
- Deployment Status

- I/NAV improvement provides benefits along 3 axis
 - **Improved Time to First Fix** - fast retrieval of navigation data
 - **Improved data demodulation robustness** – demodulate nav data also in challenging environment
 - **Time ambiguity resolution** – for users with coarse time information (+/- 3 sec) to access Galileo system time without need to demodulate it from navigation message
- Implementation solution selected
 1. **Reduced Clock and Ephemeris** (redCED)
 2. **Reed Solomon** (RS)
 3. **Secondary Synchronisation Pattern** (SSP)
- Careful consideration of users
 - Improvements transparent to legacy users
 - Participating (new) users will take benefit



Public SIS-ICD contains I/NAV impr. spec since 01/2021

Basic Idea: Reduce the number of information required to generate a position fix (at cost of accuracy)

I/NAV Clock and Ephemeris Parameter

- are spread over 4 I/NAV words (I/NAV-1, -2, -3, -4)
- all 4 I/NAV words need to be received (with same IODnav) to establish a position fix

Parameter	Definition	Bits	Scale factor	Unit
M ₀	Mean anomaly at reference time	32*	2 ⁻³¹	semi-circle**
Δn	Mean motion difference from computed value	16*	2 ⁻⁴³	semi-circle/s**
e	Eccentricity	32	2 ⁻³³	dimensionless
A ^{1/2}	Square root of the semi-major axis	32	2 ⁻¹⁹	m ^{1/2}
OMEGA ₀	Longitude of ascending node of orbital plane at weekly epoch***	32*	2 ⁻³¹	semi-circle**
i ₀	Orbit inclination angle at reference time	32*	2 ⁻³¹	semi-circle**
OMEGA	Argument of perigee	32*	2 ⁻³¹	semi-circle**
OMEGADOT	Rate of change of right ascension	24*	2 ⁻⁴³	semi-circle/s**
IDOT	Rate of change of inclination angle	14*	2 ⁻⁴³	semi-circle/s**
C _{uc}	Amplitude of the cosine harmonic correction term to the argument of latitude	16*	2 ⁻²⁹	rad
C _{us}	Amplitude of the sine harmonic correction term to the argument of latitude	16*	2 ⁻²⁹	rad
C _{rc}	Amplitude of the cosine harmonic correction term to the orbit radius	16*	2 ⁻⁵	m
C _{rs}	Amplitude of the sine harmonic correction term to the orbit radius	16*	2 ⁻⁵	m
C _{ic}	Amplitude of the cosine harmonic correction term to the angle of inclination	16*	2 ⁻²⁹	rad
C _{is}	Amplitude of the sine harmonic correction term to the angle of inclination	16*	2 ⁻²⁹	rad
t _{0e}	Ephemeris reference time	14	60	s
Total ephemeris bits		356		

Table 7-2: Ephemeris parameters

Parameter	Definition	Bits	Scale factor	Unit
t _{0e}	Clock correction data reference Time of Week	14	60	s
a ₀	SV clock bias correction coefficient	31*	2 ⁻³⁴	s
a ₀₁	SV clock drift correction coefficient	21*	2 ⁻⁴⁶	s/s
a ₀₂	SV clock drift rate correction coefficient	6*	2 ⁻⁵⁹	s/s ²
Clock Correction Parameters		72		

Table 7-5: Clock correction parameters

Reduced Clock and Ephemeris Parameter

- Idea: reduce number of CED parameters and bit allocation to squeeze all information into a single I/NAV word (**122 bits**)
- redCED
 - are derived **on-board** from full CED
 - provide degraded accuracy compared to CED
- User exploitation:
 - First position fix with redCED

Parameter	Number of bits	Scale factor	Reference value	Unit
ΔA _{red}	5*	2 ⁸	29600000	meter
e _{red}	13*	2 ⁻²²	0	dimensionless
e _{svred}	13*	2 ⁻²²	0	dimensionless
Δi _{red}	17*	2 ⁻²²	56/180	semi-circle**
Ω _{red}	23*	2 ⁻²²	0	semi-circle**
λ _{red}	23*	2 ⁻²²	0	semi-circle**
a _{0red}	22*	2 ⁻²⁶	0	s
a _{01red}	6*	2 ⁻³⁵	0	s/s
Total bits	122			

428 bits



122 bits

Basic Idea: Introduce additional redundancy in navigation message to correct for lost or corrupted data

Introduction of Reed-Solomon Clock and Ephemeris Data (RS CED) to the I/NAV message (E1-B)

- RS coding provides
 - Correction of residual errors **AND** recovery of erased information
- 4 different RS CED words are generated **on-board** (obtained from CED), per sub-frame 2 RS CED words are broadcast
- Any set of four different error free received (RS) CED words recovers the CED

• Examples:



CED recoverable:
all CED words available



CED *not* recoverable:
CED 4 is missing
doublets are not useful



CED recoverable:
CED 2 & 3 can be recovered by
using **joker words** RS CED 1 & 2



CED *not* recoverable:
four *different* words are
required



CED recoverable:
four *different* words are
available

Secondary Synchronization Pattern

Basic Idea: Introduce defined bit pattern in msg that can be detected by correlation (no need to demodulate message)

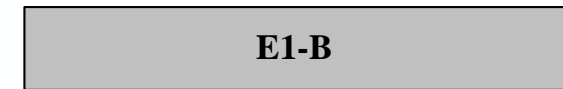
Introduction of Secondary Synchronization Pattern (SSP) into the I/NAV message (E1-B) supports reconstruction of the GST, without the need to demodulate the navigation message

	SSP1	SSP2	SSP3
Plain SSP configurations	00000100	00101011	00101111

- SSP replaces spare bits on E1-B
- Required level of coarse synchronisation
- Ambiguous Time Of Week (TOW) information can be retrieved

- SSP1 detected → TOW modulo 6s = 1s
- SSP2 detected → TOW modulo 6s = 3s
- SSP3 detected → TOW modulo 6s = 5s

Enables fast GST recovery (modulo 6 seconds) already at symbol level

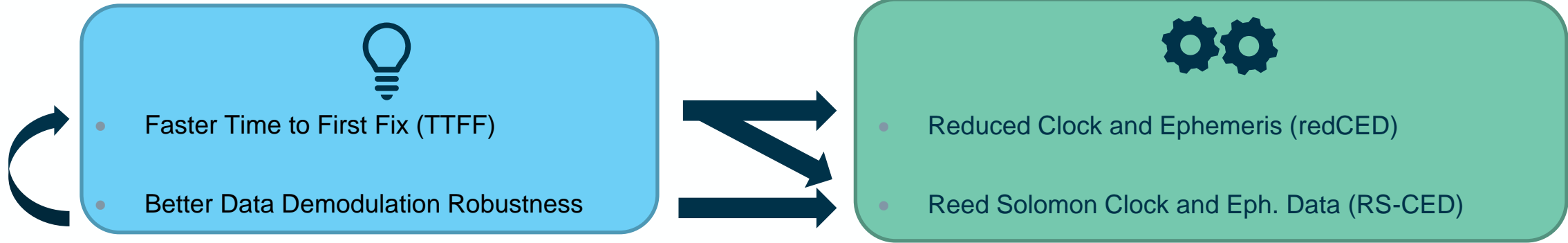


Sync	Symbols for Word i (2/2)	Total
10	240	250

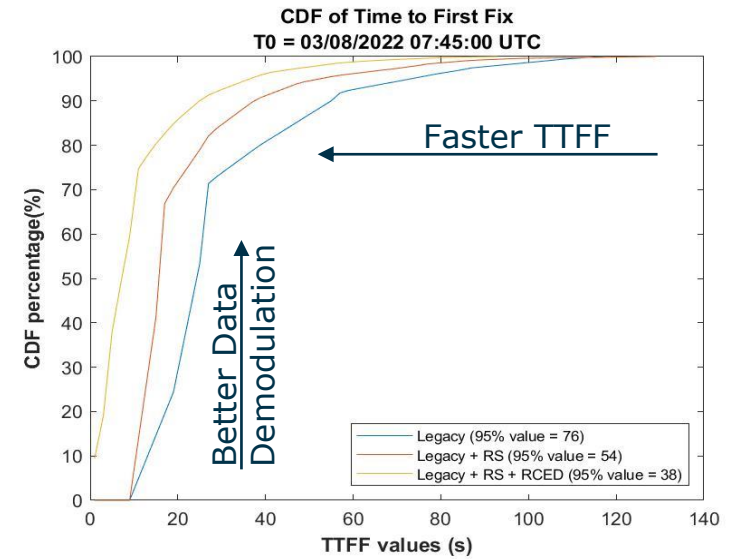
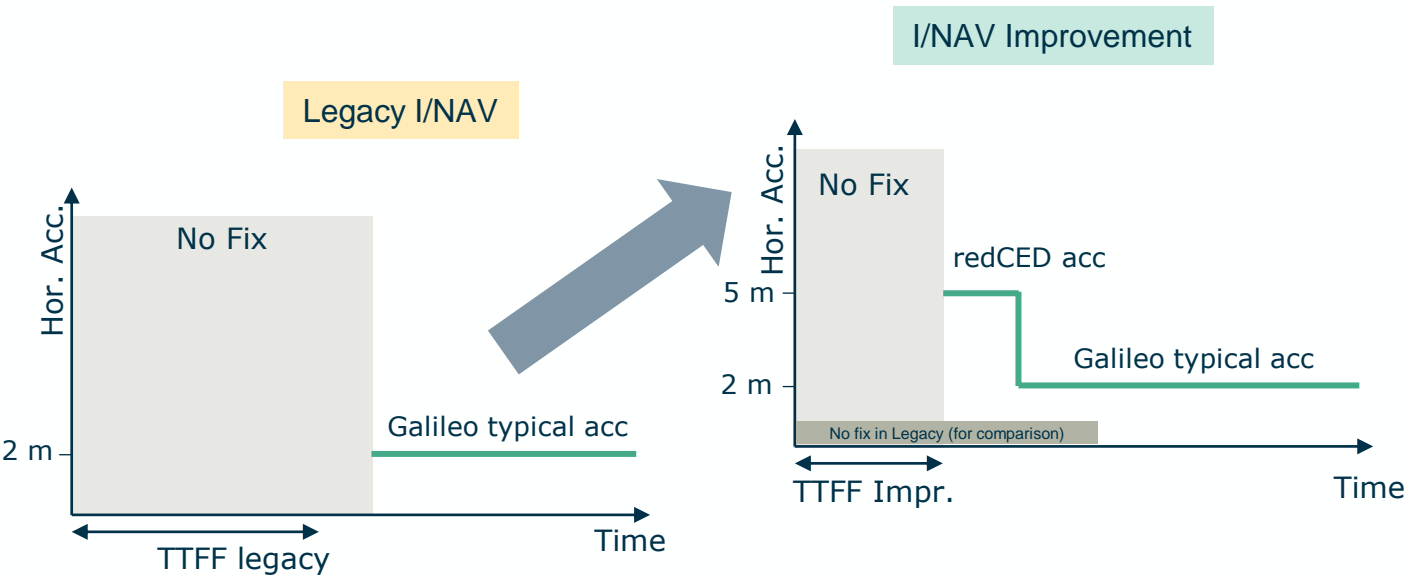
Even/Odd=1	Type=0	Data i (2/2)	EDBS	SAR	Spare	CRC i	SSP	Tail	Total
1	1	16	40	22	2	24	8	6	120

- I/NAV Improvement – User Benefits
- I/NAV Improvement – Technical Content
 - Reduced Clock and Ephemeris Data
 - Reed Solomon coded Clock and Ephemeris Data
 - Secondary Synchronization Pattern
- **User Exploitation and Performance**
- Deployment Status

RS-CED/redCED - User Exploitation and Performance (I)



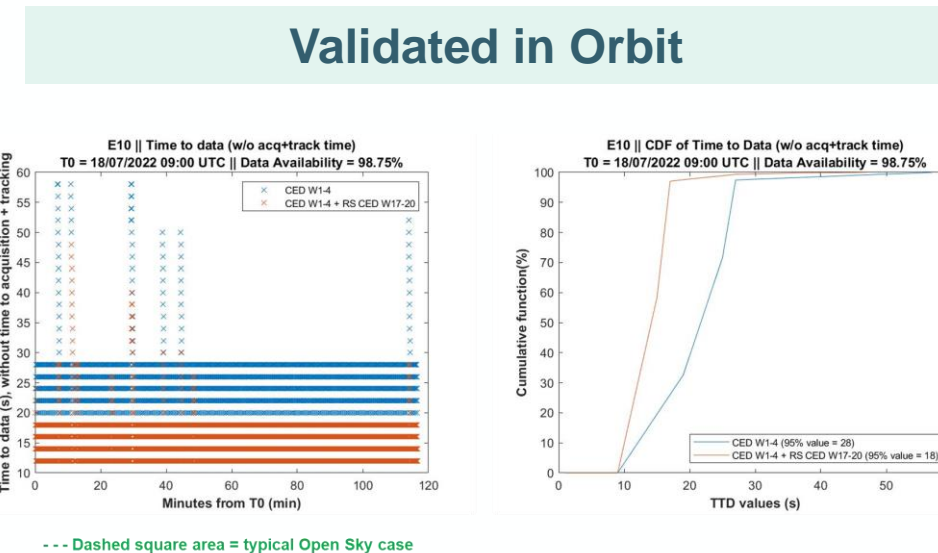
- **Key** performance parameter from **user perspective is TTFF** (also as consequence of better data demodulation robustness)
- User performance improvement validated during I/NAV IOV in Urban Environment



Reed Solomon Clock and Ephemeris Data (RS-CED) and redCED for improved Time to Data

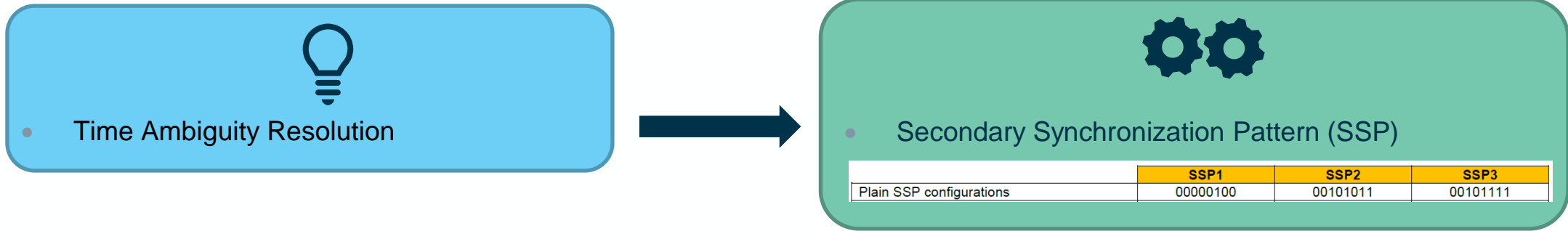
- Time to Data (TTD) – performance parameter on single link (different than Time to First Fix)
- Performance characterization in Open Sky Environment (makes it independent from local environment and deterministic)
- Time to Data for coarse accuracy (as per redCED performance characterization) and full accuracy:

Parameter	Performance as measured (not a commitment)	Percentile
TTD [sec] for <u>coarse accuracy</u> with I/NAV improvement (redCED and RS-CED use) in Open Sky	16*	95%
TTD [sec] for <u>full accuracy</u> with I/NAV improvement (RS-CED use) in Open Sky	22*	95%
TTD [sec] for <u>full accuracy</u> w/o I/NAV improvement in Open Sky	32*	95%

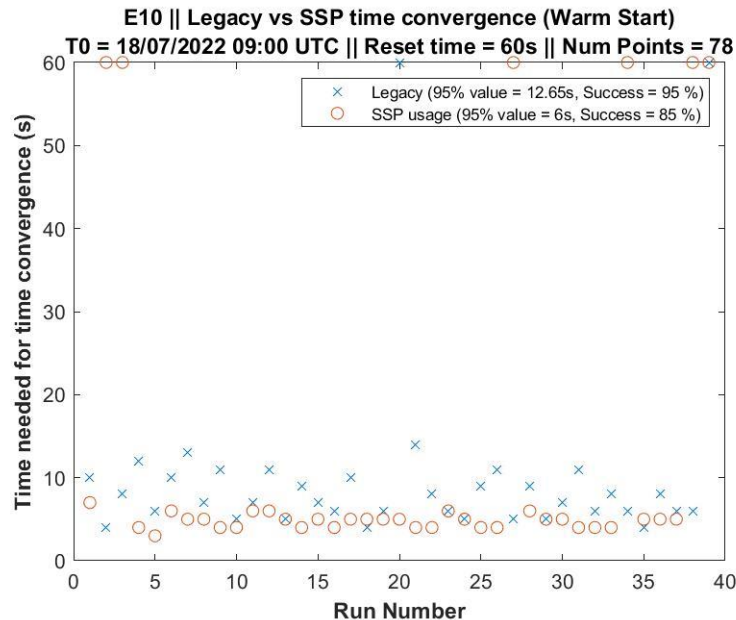


*: as per M. Paonni et al, Improving the Performance of Galileo E1-OS by Optimizing the I/NAV Navigation Message, ION 2019

SSP - User Exploitation and Performance (I)



- **SSP** provide to users that already have coarse time information (+/- 1.5 sec) the means to resolve remaining time ambiguity and give access to the Galileo System Time. No need to demodulate time information from the navigation message.
- User performance improvement validated during I/NAV IOV in Urban Environment



Access to Galileo System Time is key for connected mass market users!



I/NAV Improvement provides 2-times faster access to Galileo system time for users that have coarse time information in Urban Environment!

Secondary Synchronisation Pattern (SSP) for Time (GST) Dissemination

- Time to Time (TTT) – performance parameter on a single link for the provision of time (GST) information
 - Note: SSP provides time ambiguity resolution within interval of +/- 3 sec

Parameter	Performance as measured (not a commitment)	Percentile
TTT [sec] for <u>time ambiguity</u> (with SSP) in Open Sky	6	95%
TTT [sec] for <u>absolute time</u> (with legacy I/NAV) in Open Sky	13	95%



Validated in Orbit

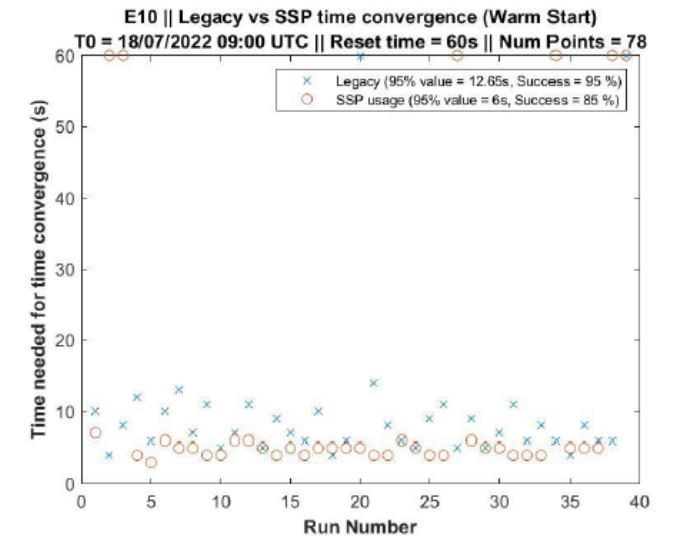


Figure 4-1: Time convergence Legacy vs SSP, Phase 1 - 18/07/2022

- I/NAV Improvement – User Benefits
- I/NAV Improvement – Technical Content
 - Reduced Clock and Ephemeris Data
 - Reed Solomon coded Clock and Ephemeris Data
 - Secondary Synchronization Pattern
- User Exploitation and Performance
- **Deployment Status**

I/NAV Improvement Deployment Status

I/NAV improvement is calculated on-board the satellite

- No change at Ground Mission Segment
- New software for signal generation unit available and under upload

Current Status

- 2 satellites (GSAT0223 and GSAT0224) already provide I/NAV improvement
- Remaining Galileo FOC satellites follow soon

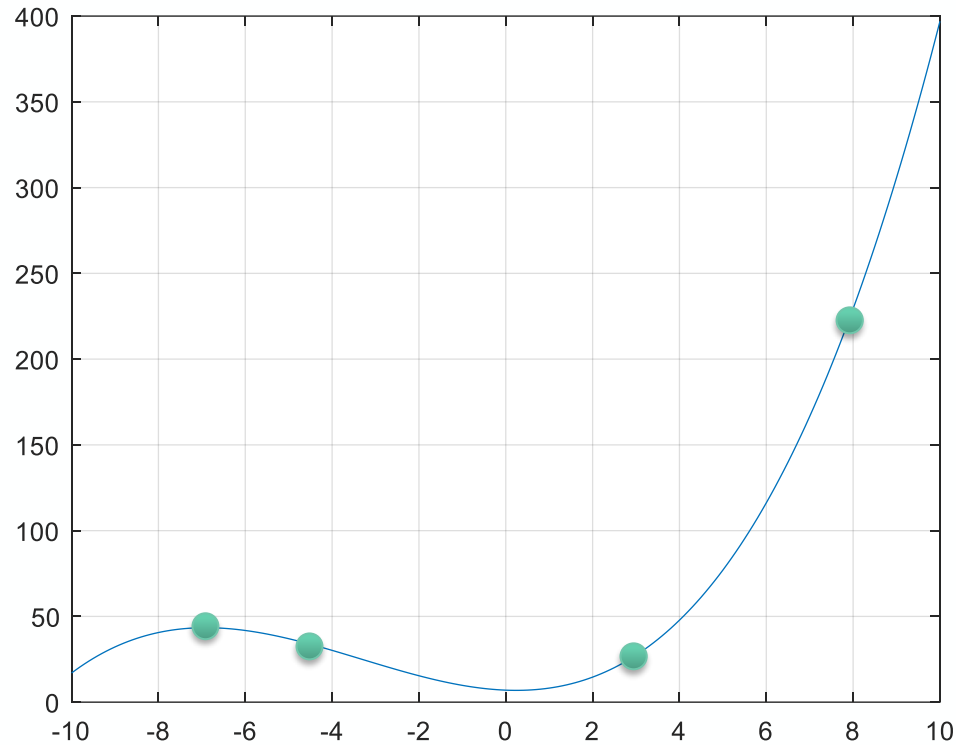


I/NAV improvement is there – get ready to use it!

BACKUP MATERIAL

Basic Idea: Introduce additional redundancy in navigation message to correct for lost or corrupted data

Analogy for illustration

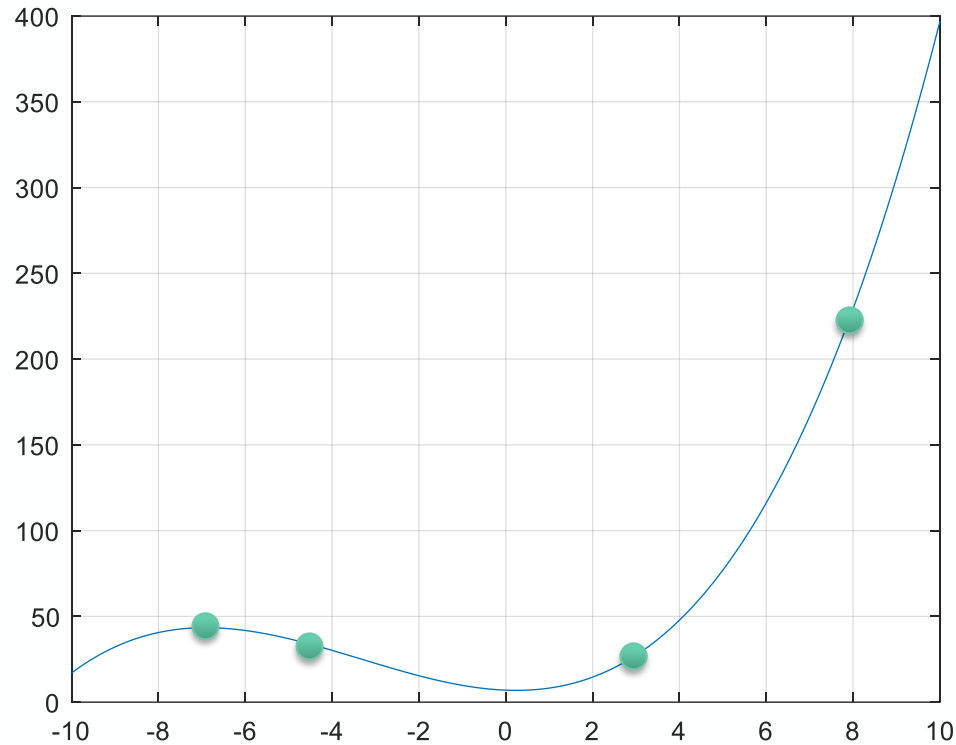


Polynomial of degree 3 ($y = ax^3 + bx^2 + cx + d$)

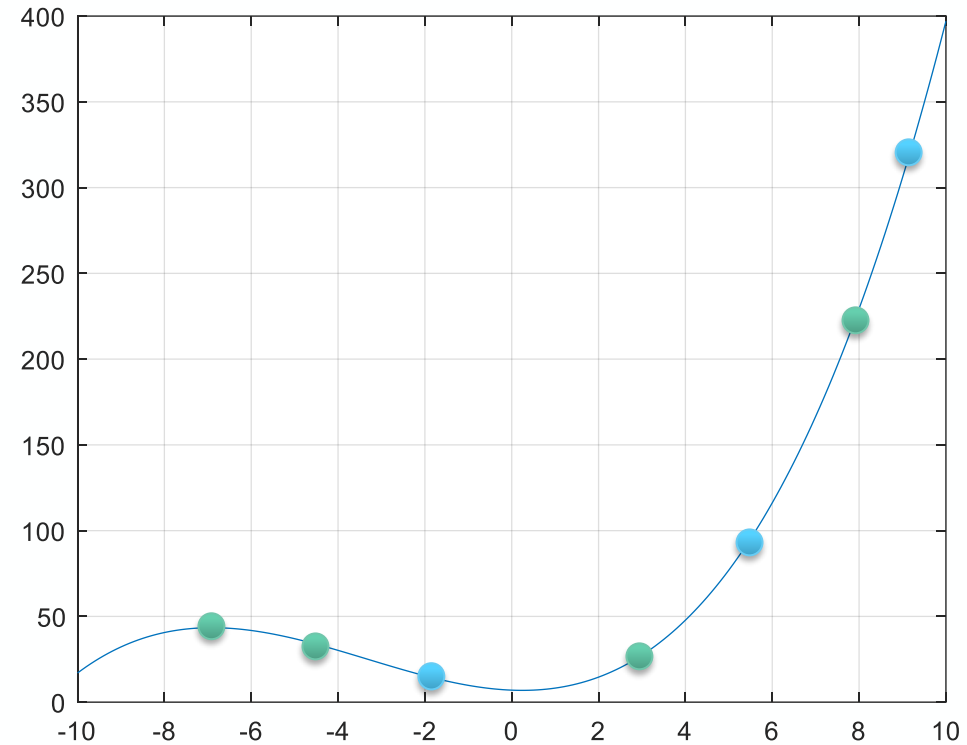
ESA UNCLASSIFIED - Releasable to the Public

Basic Idea: Introduce additional redundancy in navigation message to correct for lost or corrupted data

Analogy for illustration



Polynomial of degree 3 ($y = ax^3 + bx^2 + cx + d$)
 ESA UNCLASSIFIED - Releasable to the Public

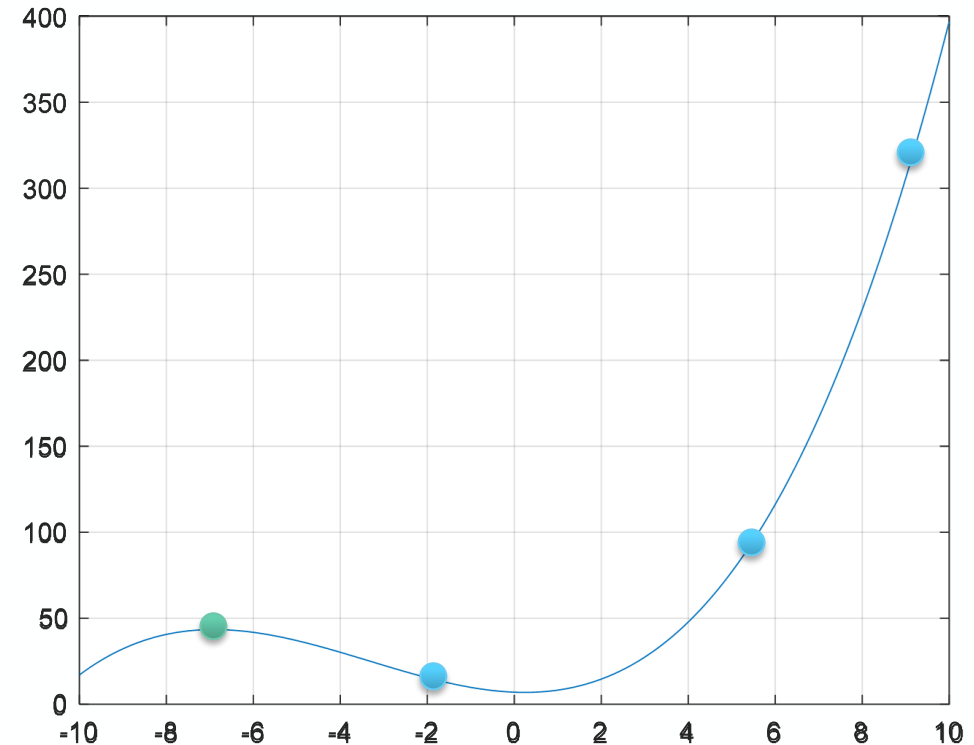
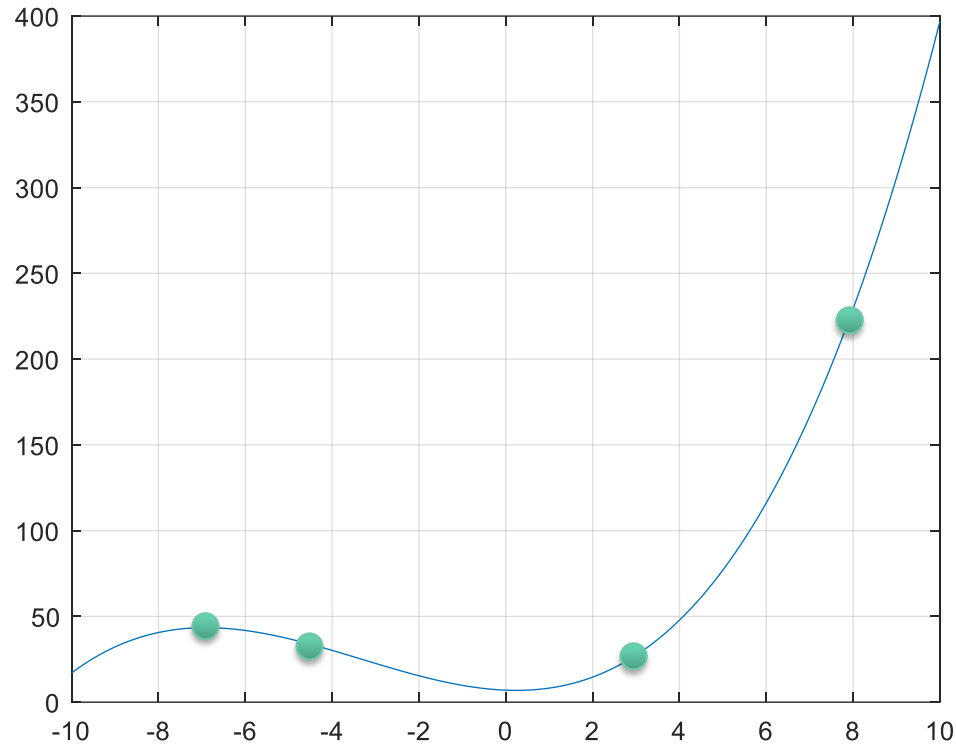


Introduce additional (redundant) grid points

Reed Solomon Coding of the Clock and Ephemeris Data

Basic Idea: Introduce additional redundancy in navigation message to correct for lost or corrupted data

Analogy for illustration



Polynomial of degree 3 ($y = ax^3 + bx^2 + cx + d$)

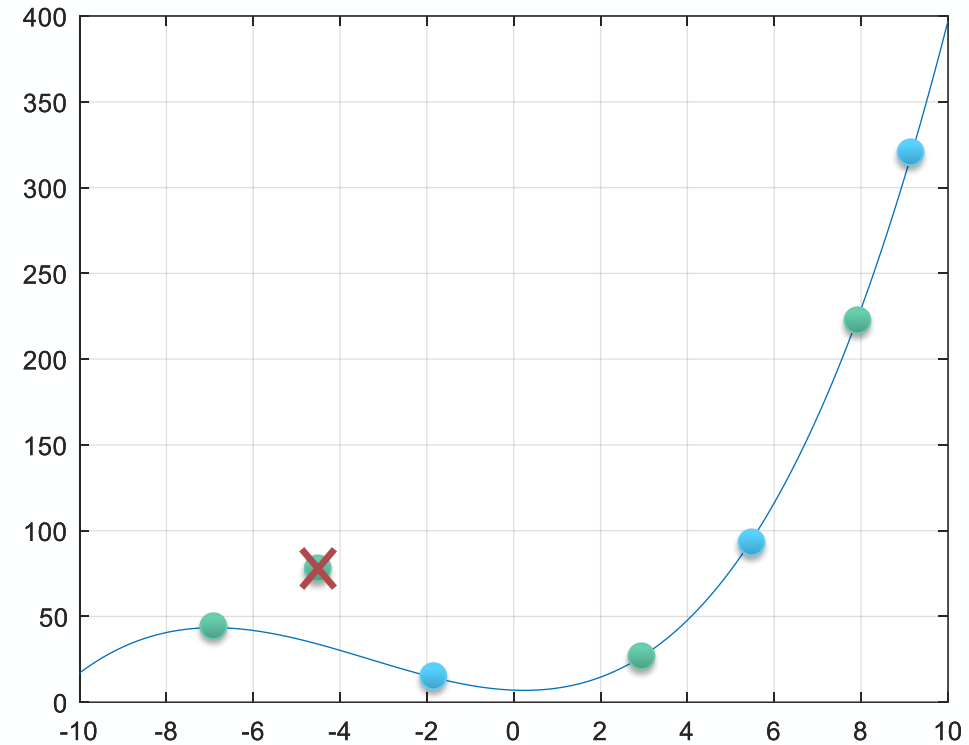
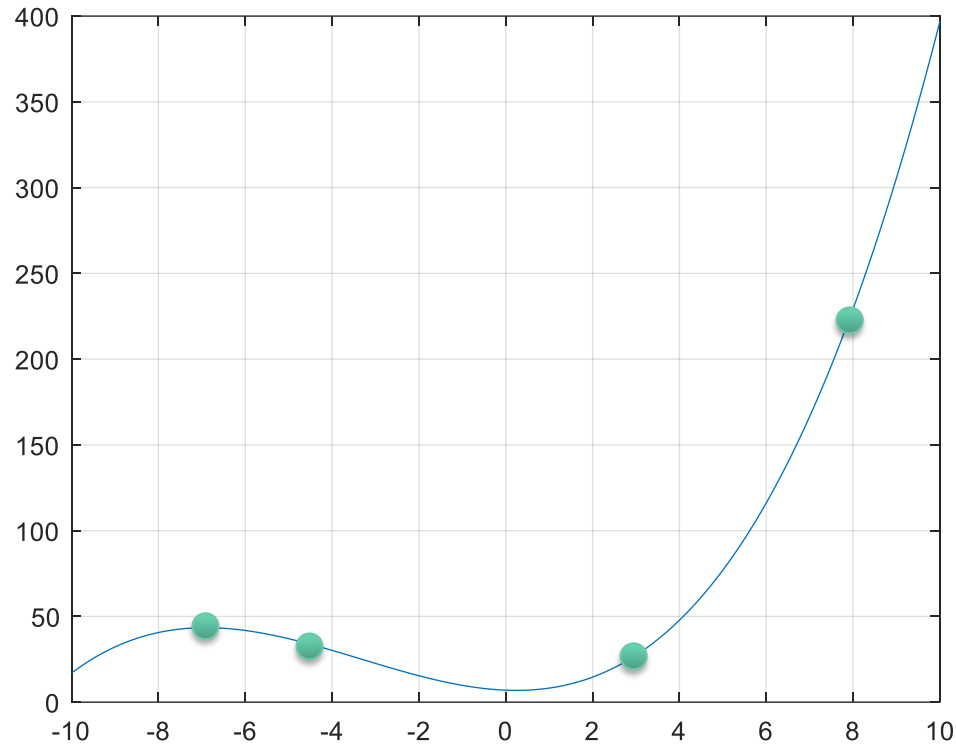
ESA UNCLASSIFIED - Releasable to the Public

Loss of data (erasure correction)

Reed Solomon Coding of the Clock and Ephemeris Data

Basic Idea: Introduce additional redundancy in navigation message to correct for lost or corrupted data

Analogy for illustration



Polynomial of degree 3 ($y = ax^3 + bx^2 + cx + d$)

ESA UNCLASSIFIED - Releasable to the Public

Error Correction