



In-orbit Testing Results of CENTISPACE LEO Augmentation Experimental Satellites

Yang Long

Beijing Future Navigation Technology Co. Ltd

Madrid Oct 2023



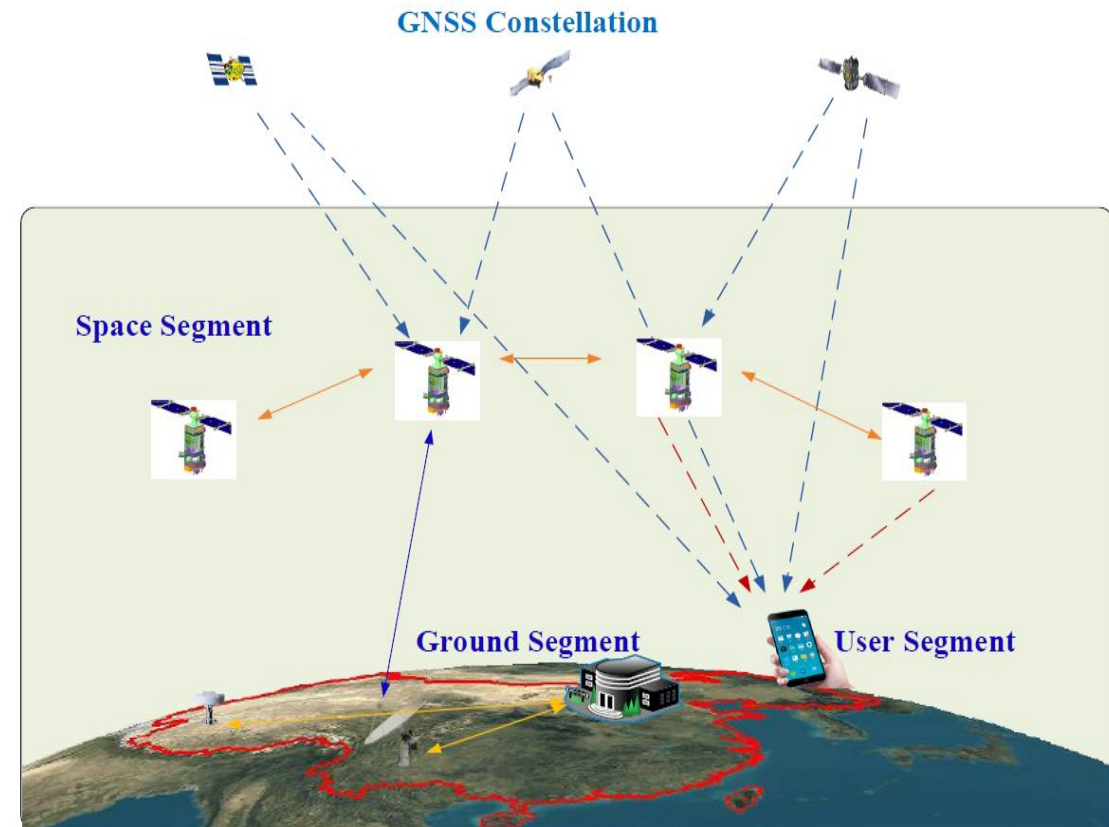
Outline

1. CENTISPACE program overview
2. In-orbit Testing Results
3. Next steps

1. CENTISPACE program overview

◆ System Description

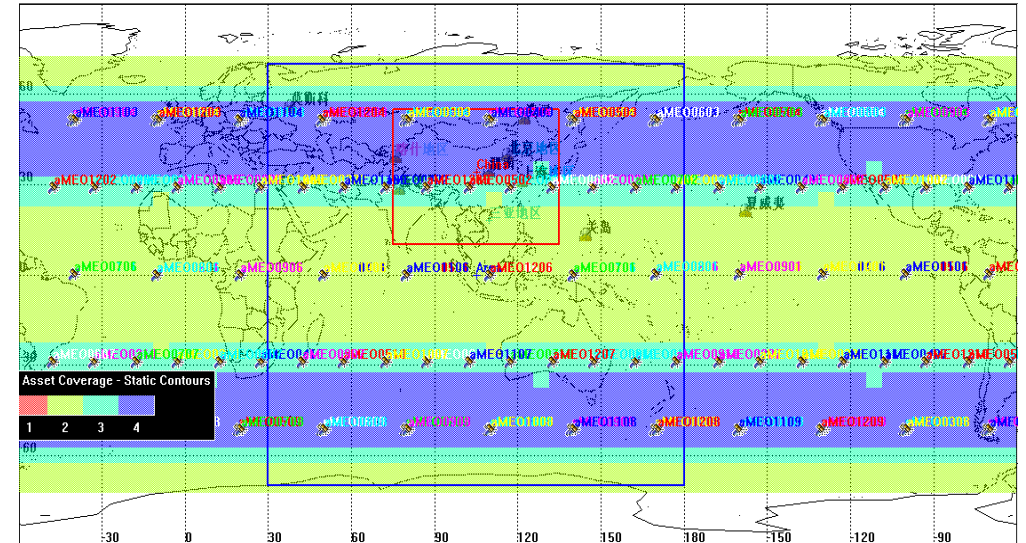
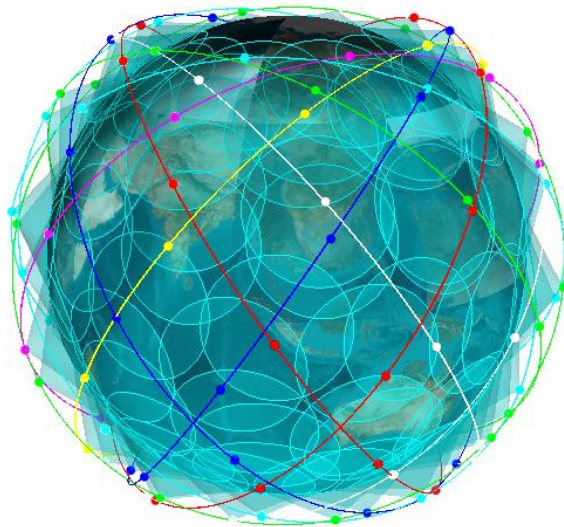
- Commercial LEO argumentation navigation system
- High Accuracy Service, Integrity augmentation Service, and GNSS monitoring Service
- Global System
- Space Segment, Ground Segment and User Segment



1. CENTISPACE program overview

◆ Sub-constellation I

- ★ Constellation: WALKER120/12/1
- ★ Orbit altitude: 975km
- ★ Inclination: 55°



| Lat, Lon | Visible Sats | Lat, Lon | Visible Sats |
|-----------|--------------|-----------|--------------|
| (0°, 0°) | 2-6 | (40°, 0°) | 5-7 |
| (5°, 0°) | 2-5 | (45°, 0°) | 5-7 |
| (10°, 0°) | 3-4 | (50°, 0°) | 5-7 |
| (15°, 0°) | 3-5 | (55°, 0°) | 4-7 |
| (20°, 0°) | 3-4 | (60°, 0°) | 4-6 |
| (25°, 0°) | 3-5 | (65°, 0°) | 2-5 |
| (30°, 0°) | 4-5 | (70°, 0°) | 2-3 |
| (35°, 0°) | 5-6 | | |

More than 2 coverages are between 70°N and 70°S

1. CENTISPACE program overview

◆ Sub-constellation II

- ★ Constellation: WALKER 30/3/1
- ★ Orbit altitude: 1100km
- ★ Inclination: 87.4°
- ★ Expand coverage in polar regions

◆ Sub-constellation III

- ★ Constellation: WALKER 40/4/1
- ★ Orbit altitude: 1100km
- ★ Inclination: 30.0°
- ★ Expand coverage in low latitude regions

1. CENTISPACE program overview

◆ Ground Segment

- **Master Station:** manage and control of the entire system equipment; process monitoring data observed by satellite or ground GNSS receiver; calculate and generate precise ephemeris and satellite clock correction data
- **TT&C Station:** tracking and controlling satellites
- **Monitoring Station:** generate GNSS observation data

1. CENTISPACE program overview

◆ Our ITU Filings

| ID number (SNS) | adm | ORG or Geo.area | Satellite name | Earth station | long_nom | Date of receipt | ssn_ref | ssn_no | WIC/IFIC (ifc.mdb) | WIC/IFIC date |
|---------------------------|----------------|--------------------|------------------------------|----------------|----------------|-----------------|----------------|----------------|----------------------|------------------|
| <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | <u>up down</u> | |
| 118520162 | CHN | | CENTISPACE-1 | | N-GSO | 06.07.2018 | API/C | 488 | 2878 | 04.09.2018 |
| 118520162 | CHN | | CENTISPACE-1 | | N-GSO | 06.07.2018 | CR/C | 4801 | 2882 | 30.10.2018 |
| 118520283 | CHN | | CENTISPACE-2 | | N-GSO | 11.09.2018 | API/C | 539 | 2881 | 16.10.2018 |
| 118520283 | CHN | | CENTISPACE-2 | | N-GSO | 11.09.2018 | CR/C | 4847 | 2886 | 08.01.2019 |
| 120545323 | CHN | | CENTISPACE-3 | | N-GSO | 29.12.2020 | API/A | 12741 | 2942 | 23.03.2021 |
| 120520264 | CHN | | CENTISPACE-3 | | N-GSO | 29.12.2020 | CR/C | 5516 | 2953 | 24.08.2021 |
| 122545286 | CHN | | CENTISPACE-4 | | N-GSO | 24.11.2022 | API/A | 13236 | 2991 | 07.03.2023 |

1. CENTISPACE program overview

◆ RF Characteristics of Ranging Signal

| Parameters | GENTISPACE | |
|---------------------------------------|-------------|-------------|
| | CL1 | CL5 |
| Modulation Type | BPSK | BPSK |
| Frequency Band (MHz) | 1569 - 1581 | 1170 - 1182 |
| Data rate(bps) | 1000 | 1000 |
| Chip Rate(Mcps) | 2.046 | 2.046 |
| User Received Power (Typical, dBW) | -157.0 | -157.0 |



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1. CENTISPACE program overview
- 2. In-orbit Testing Results**
3. Next steps

2. In-orbit Testing Results

◆ Space Segment

- 5 experimental satellites in orbit.
- Conducted some effective experiments on satellite platforms and payloads.
- The experiment is still ongoing.



CENTISPACE-S1
(Launch Date: 29 Sep 2018)



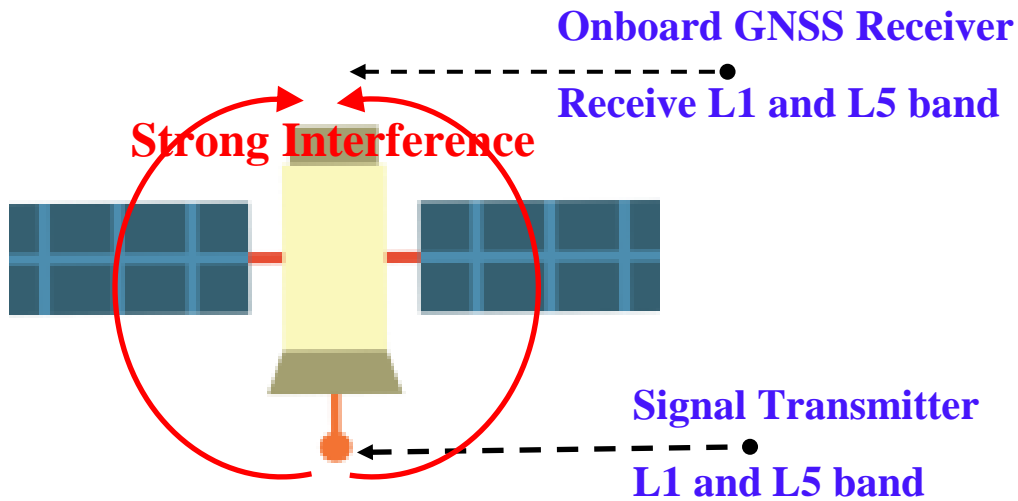
CENTISPACE-S3 / S4
(Launch Date: 6 Sep 2022)



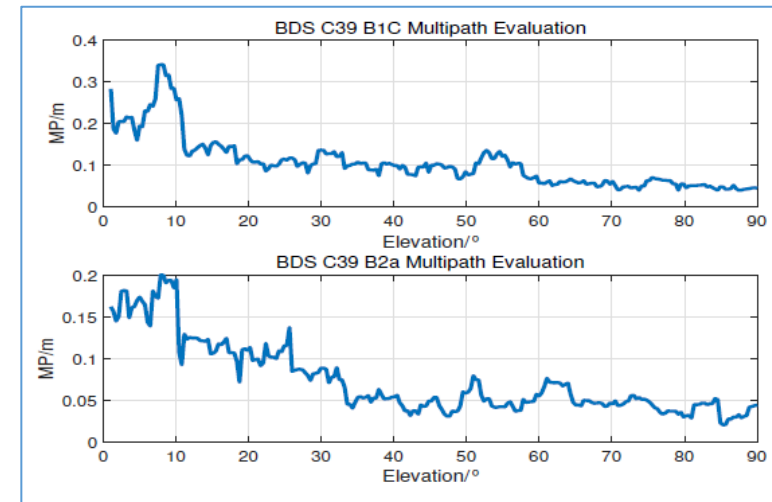
CENTISPACE-S5 / S6
(Launch Date: 7 Oct 2022)

2. In-orbit Testing Results

◆ Onboard GNSS Observation Quality Test



Multipath of Onboard GNSS Receiver

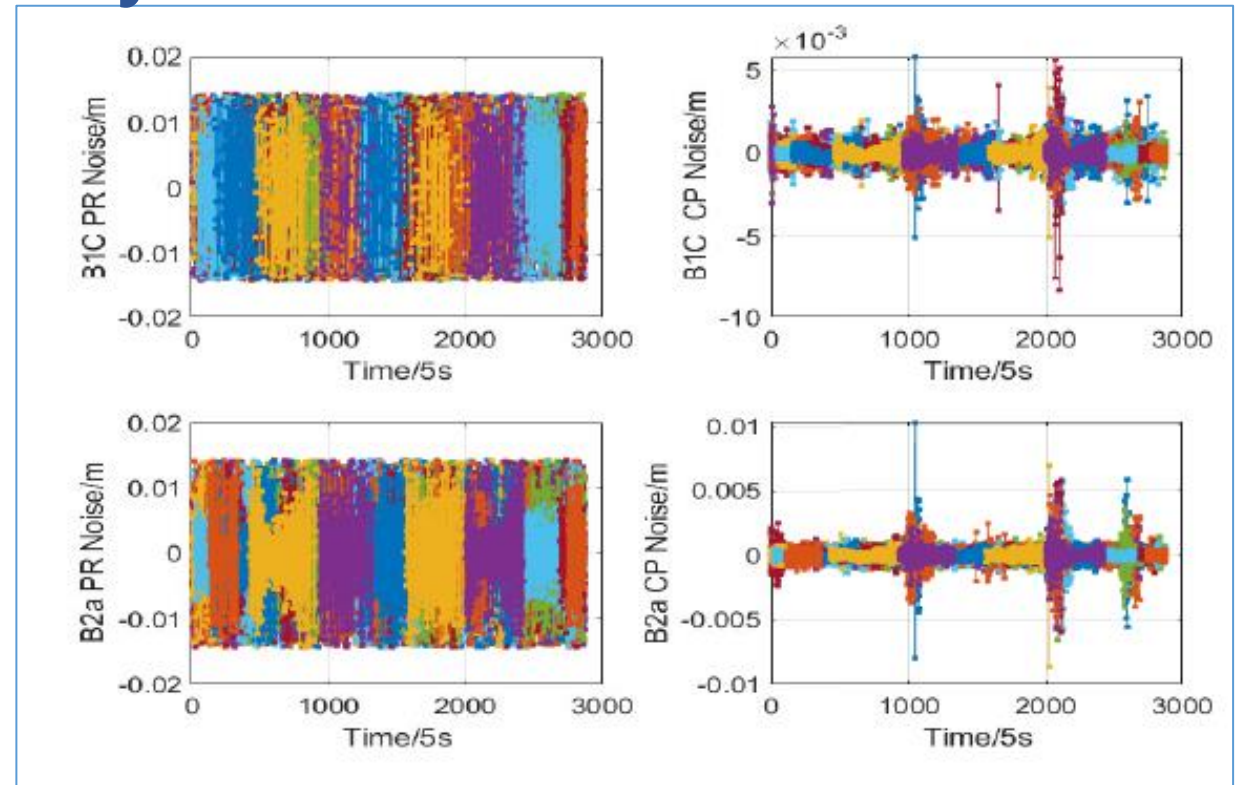


- The multipath statistical results of all the BDS satellites are $<0.48\text{m}$ for B1C signal and $<0.33\text{m}$ for B2a signal respectively, which are all less than 0.5m and are comparable with the multipath of a ground tracking stations.

2. In-orbit Testing Results

◆ Onboard GNSS Observation Quality Test

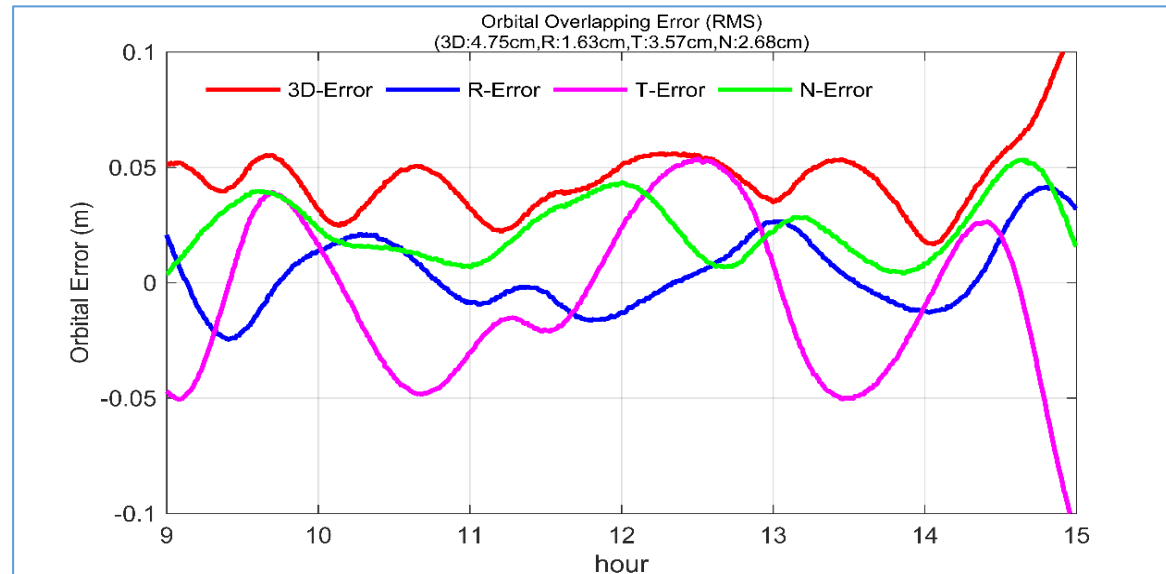
- The measurement value of pseudo range noise for BDS signals: **<8cm**.
- The measurement value of carrier phase noise for BDS signals: **<2mm**.



**BDS Measurement Noise
Observed by Space-borne GNSS Receiver**

2. In-orbit Testing Results

◆ Precise Orbit Determination



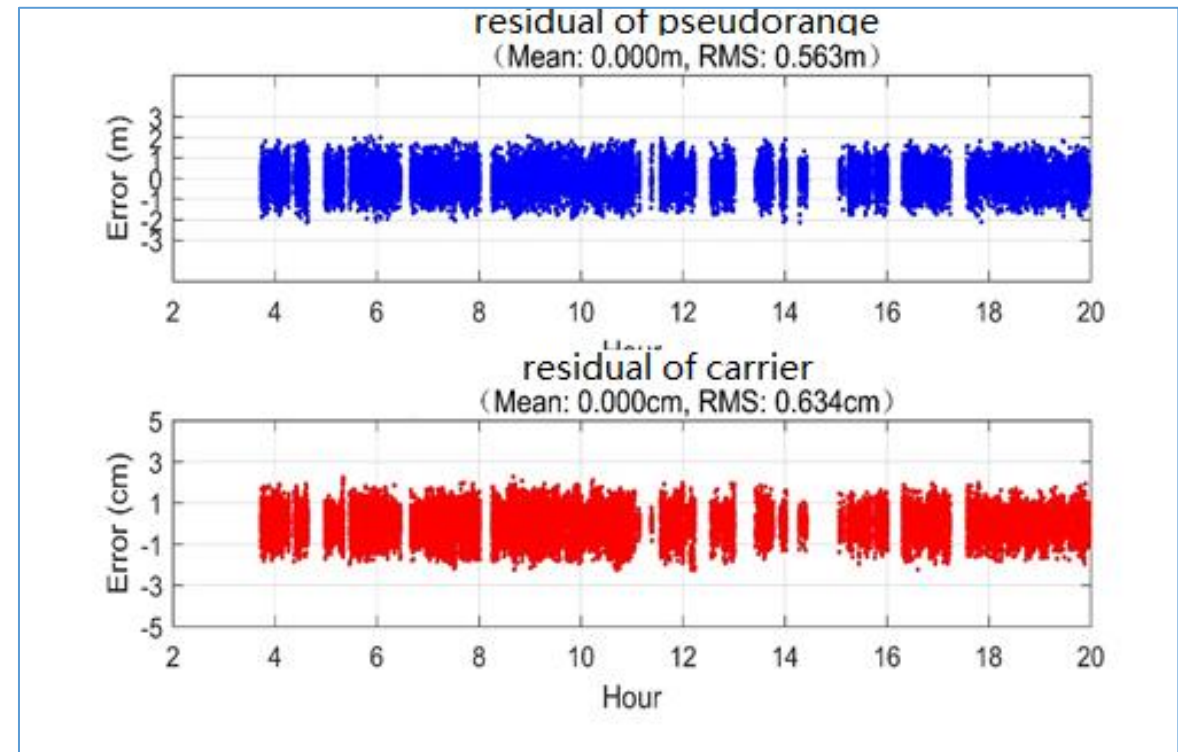
Difference of Subsequent Orbit Arc Overlap

- Precise orbit determination error with BDS signals: **<5cm**.

2. In-orbit Testing Results

◆ Precise Orbit Determination

- LEO satellites orbit determination residual of pseudo range: **0.563m (RMS)** .
- LEO satellites orbit determination residual of carrier phase: **0.634cm (RMS)** .



LEO precision orbit determination ranging residual
(Based on BDS observation data)

2. In-orbit Testing Results

◆ Ranging Signal Test Setup



High Gain Antenna



Static Test Receiver



Kinematic Test Vehicle

2. In-orbit Testing Results

Ranging Signal Quality

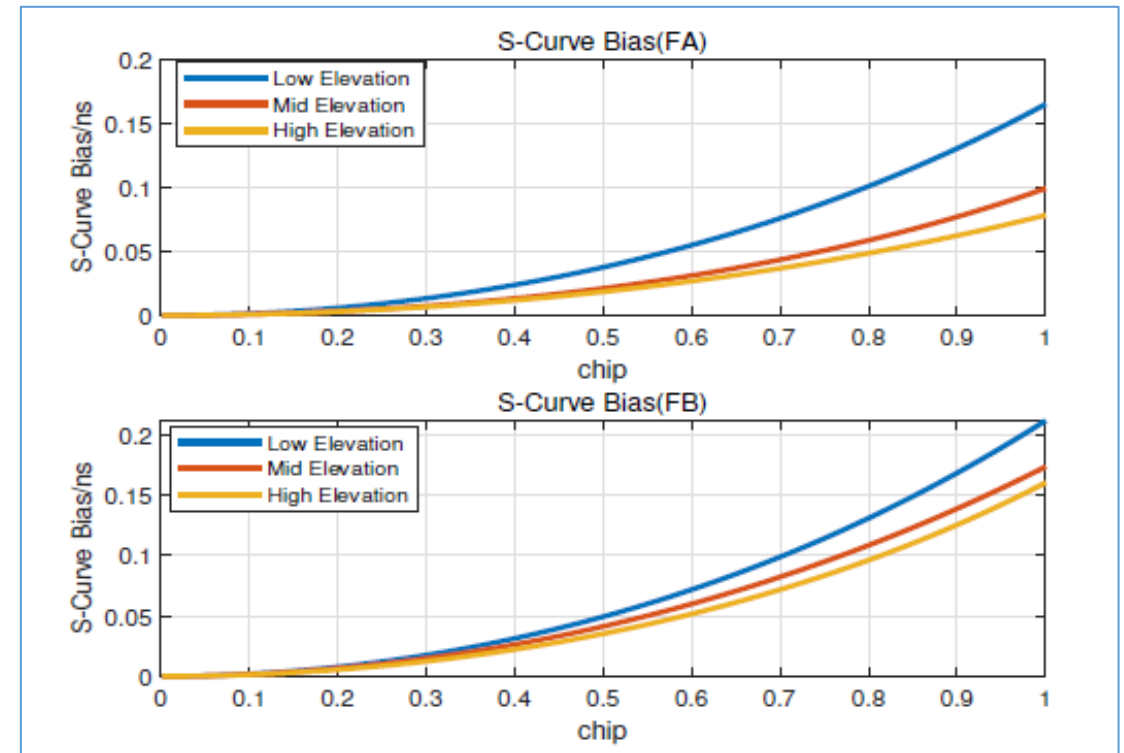
I/Q Orthogonality

| Signal | High Elevation | Mid Elevation | Low Elevation |
|--------------|----------------|---------------|---------------|
| FA (1577MHz) | 0.38° | 0.37° | 1.69° |
| FB (1174MHz) | 0.32° | 0.35° | 1.45° |

Correlation Loss

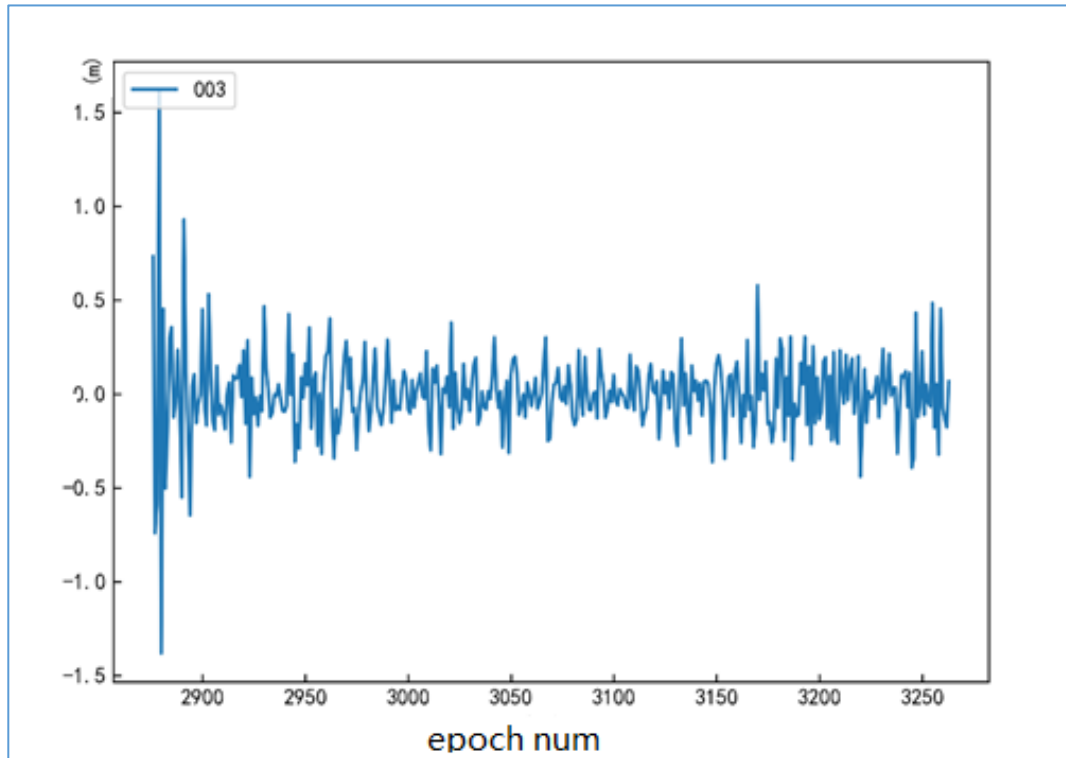
| Signal | | High Elevation | Mid Elevation | Low Elevation |
|--------|---|----------------|---------------|---------------|
| FA | I | 0.25dB | 0.26dB | 0.34dB |
| | Q | 0.23dB | 0.29dB | 0.31dB |
| FB | I | 0.17dB | 0.19dB | 0.29dB |
| | Q | 0.20dB | 0.22dB | 0.25dB |

S-Curve Bias

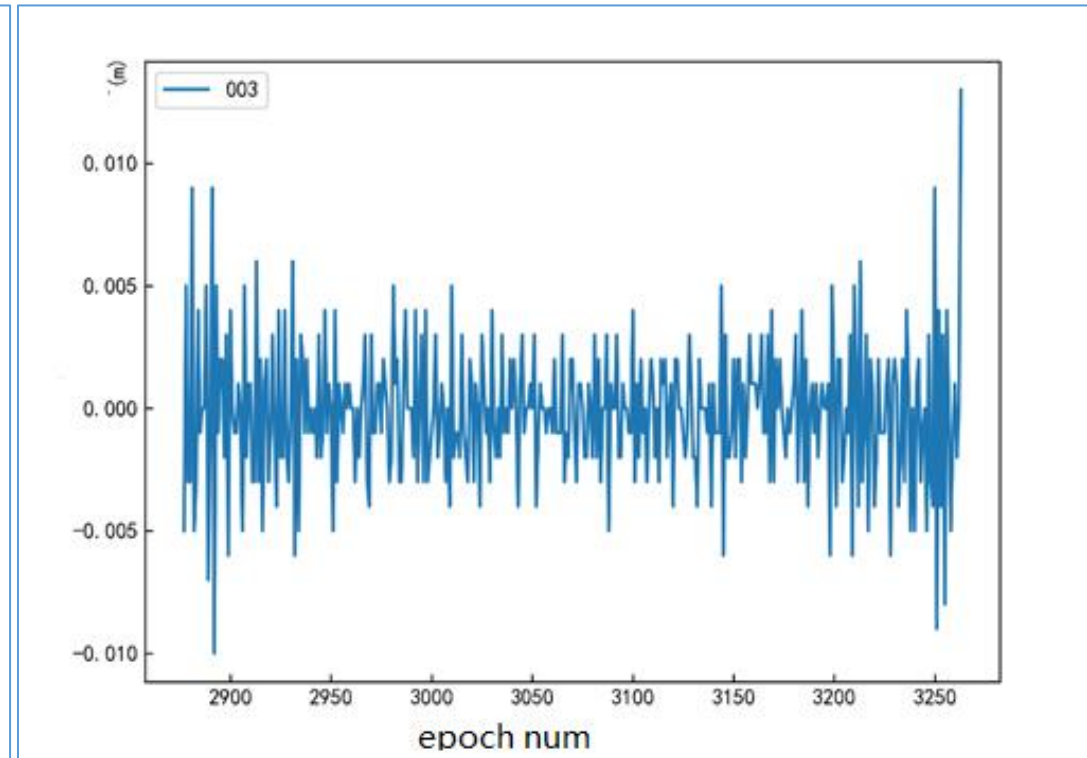


2. In-orbit Testing Results

◆ Ranging Signal Quality



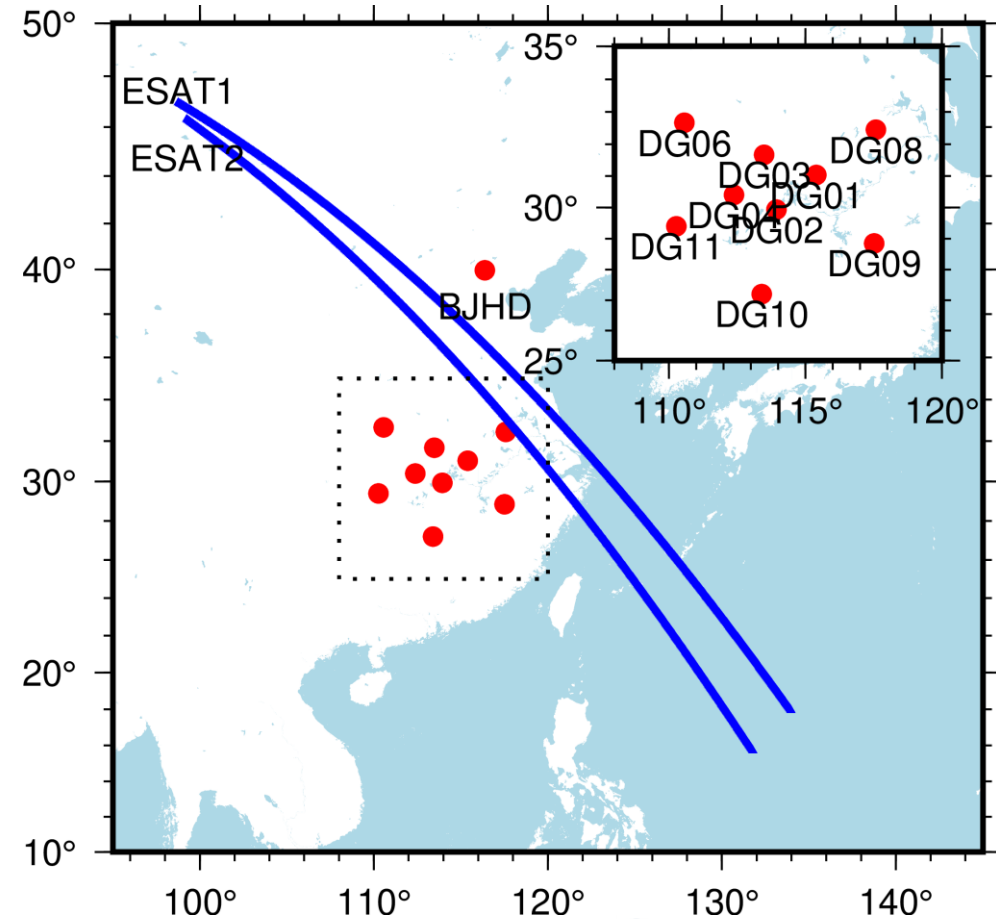
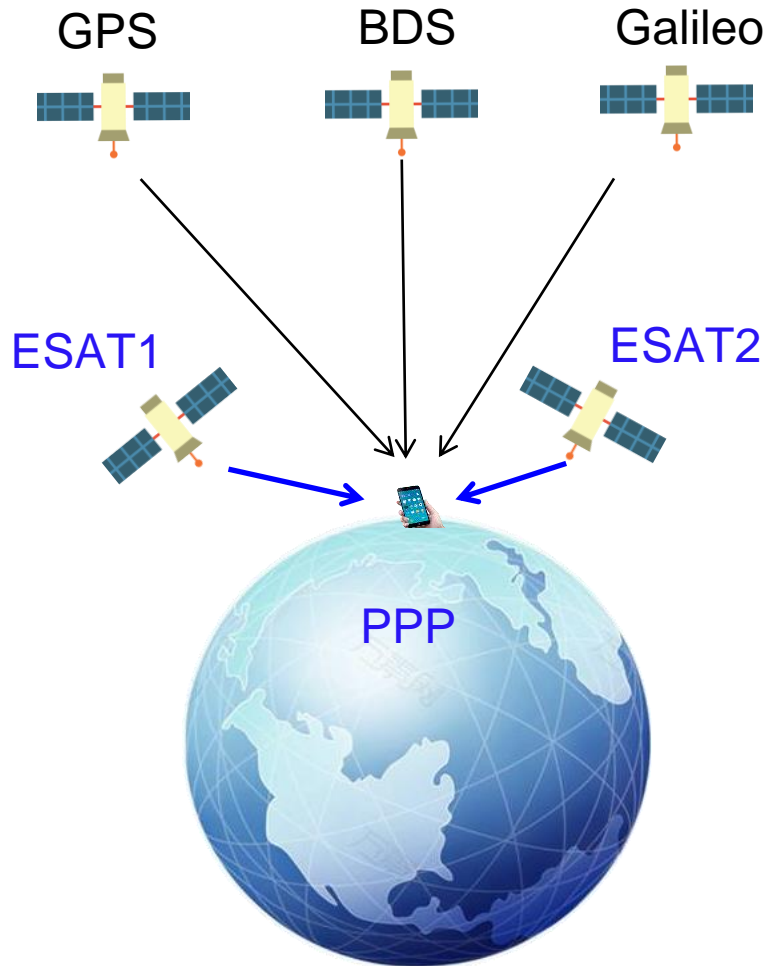
DD of Pseudorange GF Combination
(RMS: 3.2cm)



DD of Phase GF Combination
(RMS: 0.4mm)

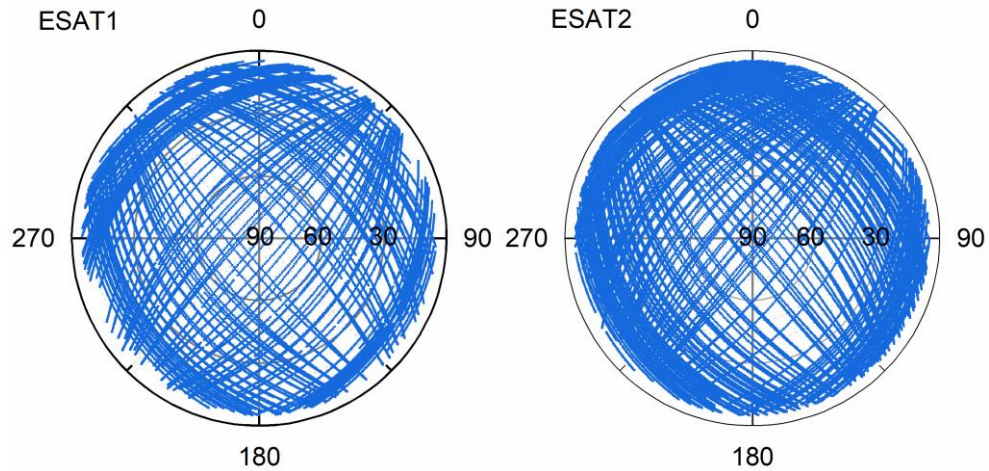
2. In-orbit Testing Results

◆ Two LEOs Augmentation PPP Experiment

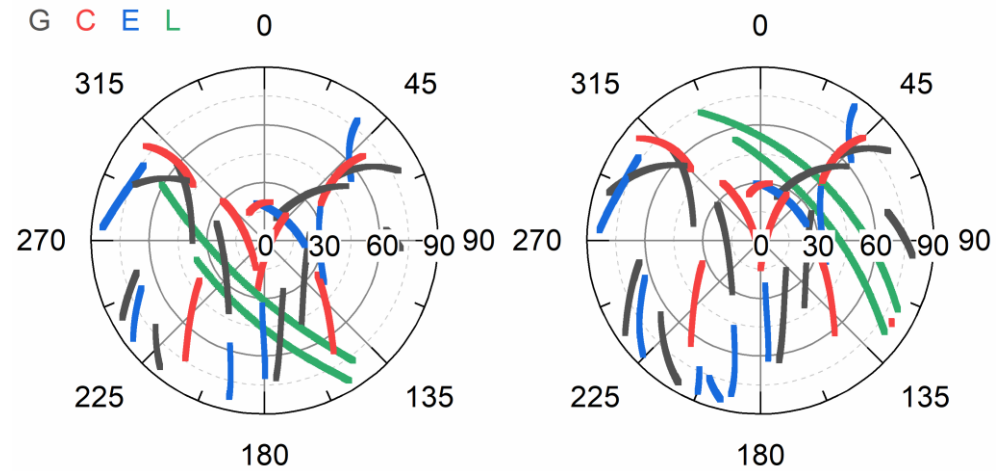


2. In-orbit Testing Results

◆ Two LEOs Augmentation PPP Experiment



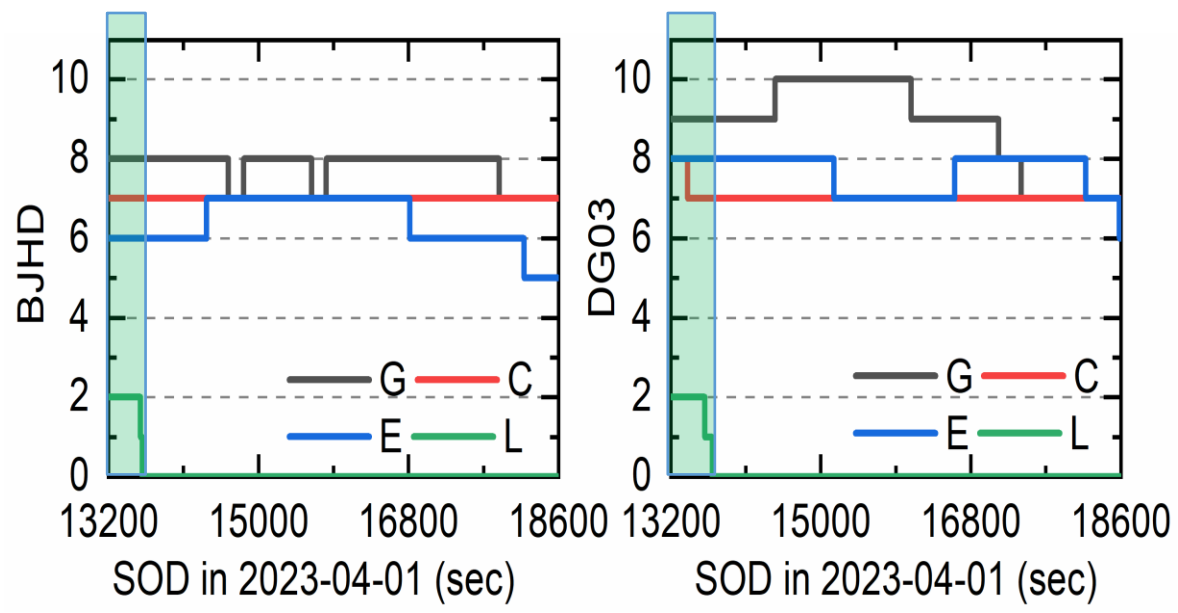
Tracks Plot of Two LEOs in 20 Days



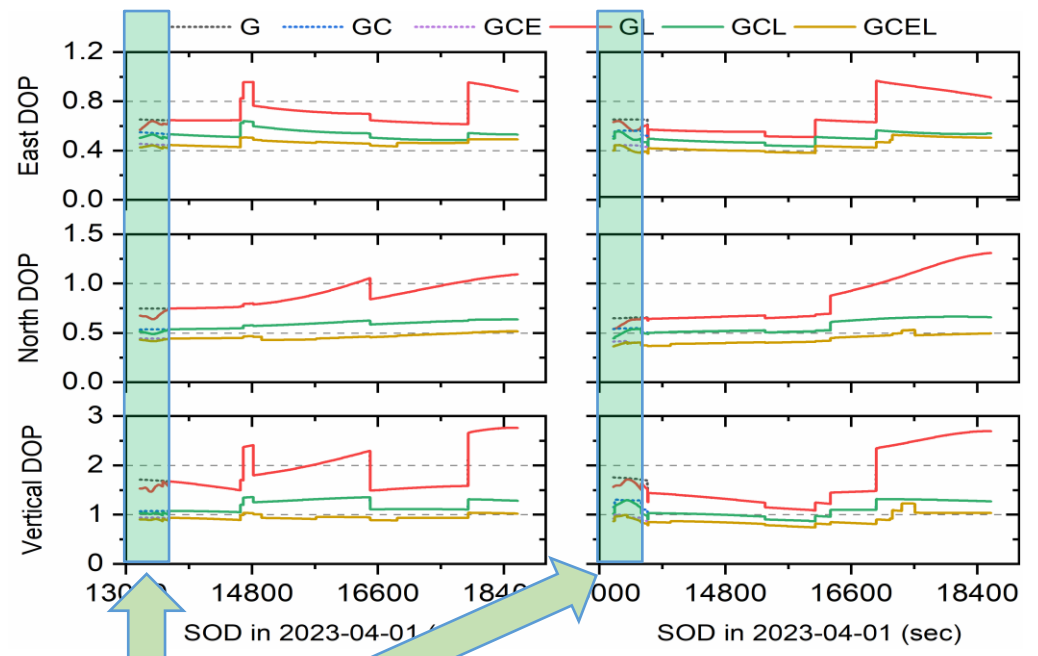
Tracks Plot of 90 Minutes

2. In-orbit Testing Results

Two LEOs Augmentation PPP Experiment



Number of Satellites

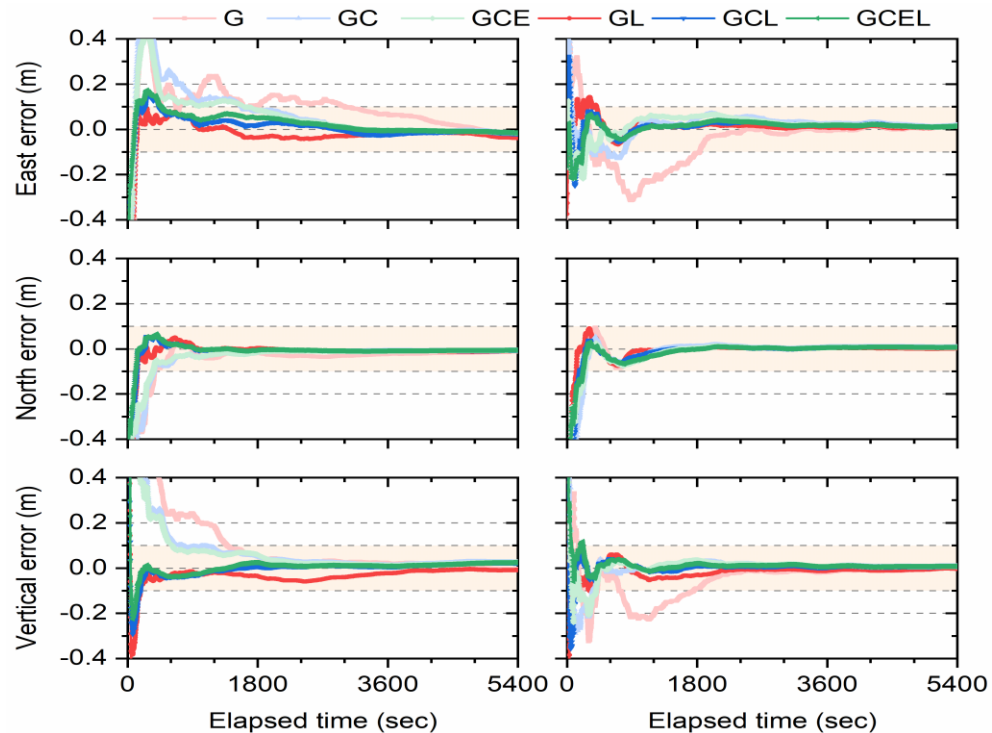


Rapid
Variation
of DOPs

DOPs of 2 Sites

2. In-orbit Testing Results

Two LEOs Augmentation PPP Experiment



Statistical Data of Two LEOs Augmentation PPP
(based on 14 kinds of constellation combinations and 42 sets of PPP results)

| GNSS system number | W/O LEO | | With LEO | |
|--------------------|------------------------|-------------------|------------------------|-------------------|
| | Convergence time (min) | 3D precision (cm) | Convergence time (min) | 3D precision (cm) |
| 1 | 32.7 | 6.1 | 16.7 | 5.2 |
| 2 | 17.9 | 4.6 | 8.9 | 3.9 |
| 3 | 14.2 | 4.6 | 5.7 | 3.8 |



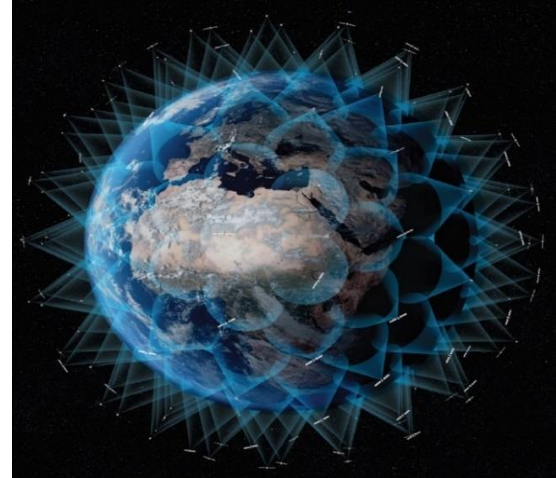
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3. Next steps

◆ Next Steps

- Complete the construction of whole constellation
- Provide high accuracy navigation services globally





THANK YOU FOR YOUR ATTENTION !

Email: yangl@centispace.com