KiboCUBE Academy

Lecture 09

Introduction to CubeSat Communication System

Teikyo University

Department of Aerospace Engineering

Lecturer Dr. Yoshihiro Tsuruda

This lecture is NOT specifically about KiboCUBE and covers GENERAL engineering topics of space development and utilization for CubeSats. The specific information and requirements for applying to KiboCUBE can be found at: https://www.unoosa.org/oosa/en/ourwork/psa/hsti/kibocube.html







Lecturer Introduction





© The University of Tokyo / NESTA, 2014 (from Left) **UNIFORM-1, HODOYOSHI-3, HODOYOSHI-4** Fight Model Picture before shipping, April 2014

Yoshihiro Tsuruda, Ph.D. Amatuer First-Class Radio Operator, JA6XMK First-Class Technical Radio Operator for On-The-Ground Services

Position:

- 2010 Ph.D. Degree in Kyushu University
- 2010 Project Researcher, Kyushu University QSAT-EOS Project
- 2011 Project Researcher, Tokyo University UNIFORM-1 & Hodoyoshi-3/4 Project, TRICOM Project
- 2017 Project Lecturer, Tokyo University AQT-D/RWASAT-1 Project, MicroDragon Project
- 2020 Lecturer, Teikyo University TeikyoSat-4 Project

Research Topics:

Micro/Nano/Pico-Satellite System Design and Electrical Components Design, Ground Station Development



- 1. Introduction
- 2. Antenna
- 3. Radio Hardware (TX/RX)
- 4. Signal Processing and Software Functions
- 5. Tele-Comm. Subsystem Integration
- 6. Conclusion

3







KiboCUBE Academy

1.1. Introduction of Tele-Comm. for Micro/Nano/Pico-satellite (CubeSat)

- Tele-Comm: Tele-Communication subsystem
 RF (Radio Frequency) is the most powerful and mature option to get information or send command from/to long distance
- Tele-Comm subsystem's responsibility
 - = very long-term activity

(Start with Tele-Comm, finish with Tele-Comm)

- □ Satellite project start from frequency coordination, and radio station license coordination
- □ Satellite project finish with sending **RF transmit** termination command
- □ In the case of University project (entry project), amateur station (UHF/VHF) is most popular
- Experimental station (S-band/X-band) is tend to be used for matured project (like Earth Observation, Science Mission)



1.2. Tele-Comm. Architecture Examples for Micro/Nano/Pico-satellite (CubeSat)

Architecture: the style and design of a building or buildings \rightarrow the style and design of a system

- The number of components
- The kind of components
- The connections between components (signal flow and its properties)



1.3. Key Concepts of Tele-Comm. for Micro/Nano/Pico-satellite (CubeSat)

□ Tele-Communication Objectives

- □ To confirm of the **satellite survival** (Beacon or CW Morse)
- To get telemetry (digital packet data of housekeeping of the satellite (like temperature, voltage, current, status, and attitude parameters), CCSDS in professional/commercial mission, AX.25 in amateur mission)
 X analog signal (like voice or song) is also one of options in amateur mission
- To control the satellite (change operational mode, start mission camera shooting, registering the mission schedule)
- To get the distance information between the satellite to ground

(ranging, especially for deep space mission)

LEO case: GPS/GNSS-based positioning is popular



1.3. Key Concepts of Tele-Comm. for Micro/Nano/Pico-satellite (CubeSat)

Primary Design Process for Tele-Comm.

- ❑ Step 1: Identify Requirements
 → Orbit, data amount and update period
- J Step 2: Select Frequency → Amateur or experimental or commercial
- ❑ Step 3: Select and Design Hardware (Antenna, Transmitter, Receiver, → Power, gain, sensitivity, G/T
- □ Step 4: Select Data Protocol
- □ Step 5: Identify Link Budget (Margin)

Iteration is needed until all interface conditions and requirements are satisfied

Step	Information Required
1. Identify	Mission type and orbit (LEO or GEO),
Requirements	Data amount and update frequency
2. Select Frequency	Type: Amateur or experimental or commercial, bandwidth, modulation
3. Select and Design	Antenna spec.
Hardware	TX/RX spec.
4. Select	Data packet format
Data Protocol	Error correction method
5. Identify Link Budget	Link margin



1.3. Key Concepts of Tele-Comm. for Micro/Nano/Pico-satellite (CubeSat)

Consideration priority for **reasonably reliable** Tele-Comm. design and implementation

- 1.Keep a **simple** configuration and a **simple** operation plan (**single** task)
- 2.Select devices with **low power** consumption
 - and that are **demonstrated on-orbit**, as much as possible
- 3.The CPU (or equivalent control unit / digital controller) of the Tele-Comm. component should be able to **recover from hang-ups by a power-on reset**
 - (especially, for tolerance against TID/SEU/SEL)
- 4.Failure recovery options and redundant configurations
- 5. Pursuit of **communication performance (Bitrate)**
- 6.Advanced functions (Autonomous control, parallel tasks)

After upper level design and verification has been thoroughly considered, move on the next step



1.3. Key Concepts of Tele-Comm. for Micro/Nano/Pico-satellite (CubeSat)

All key design features and constraints depend on which frequency is used

Frequency allocation is the most important and the first action to kickoff satellite projects

Lower frequency (VHF/UHF) is well-demonstrated, Higher frequency (5 ~ 20 GHz) is difficult to use (especially for first time developer like university mission)

❑ New possible option is Specified Low Power Radio Station

 (like LoRa: a kind of LPWA (Low Power Wide Area) protocol)
 → several CubeSats have already demonstrated this on-orbit
 (In the future, new space data link options may be defined based on these state-of-the-art
 communication technologies)









KiboCUBE Academy

2.1. Introduction of Antenna

How to emit RF to space / correct RF from space

- Antenna is a key component for satellite operation (No antenna, no mission data from space)
- Antenna design is based on studying the electromagnetic field, there are many kinds of antenna design
 (All current-flowing conductors can emit RF potentially = micro-dipole model)
 Tiny copper line on PCB is always emitting RF!
- □ Antenna performance is affected by...
 - □ **Length** (corresponding to target frequency)
 - □ **Material** (permittivity)
 - □ Shape (rod, plate, circle, ring, cubic, corn,,,,)
 - **Combination** of multiple antennas (arraying)
 - Relationship to GND plane (distance and GND plane area))



2.2. Major Characteristics of Antenna

Antenna Gain : dBi [dB is logarithmic scale] (i stands for Isotropic antenna)

- □ Antenna Pattern : Wide (broad) or narrow (beam) HPBW (Half Power Beam Width) [deg]
- □ **EIRP** (Effective Isotropic Radiated Power)
- **G/T** (Antenna gain-to-noise-temperature)
- □ **VSWR** (Voltage Standing Wave Ratio)
- **Polarization** : Linear or circular (RHCP, LHCP)
- Deployable or non-deployable :

Most Critical Decision for CubeSat Mission Success

- Major reason of "CubeSat Dead on Arrival" status might be antenna deployment failure
- In the non-deployed state of the antenna, the performance of RF transmitting is few expected to nominal condition



2.3. Design and Testing of Typical Antenna

Example : Typical monopole and dipole for CubeSat

- Material: Thin ribbon steel tend to be selected for antenna element (to store inside envelope of CubeSat by bending)
- Adjustable parameters are its length, width, thickness (length is most important factor for performance)
- One side is free end, the power feed point and mechanical interfaces are located on the other side
- The center wire of the coaxial cable is attached to this power feed point of the antenna element by soldering or spot welding
 - Monopole: the outer shield of coaxial cable is attached to GND
 - Dipole: the outer shield of coaxial cable is attached to the other paired antenna element
- ☐ Linear polarization: Electric field oscillating direction is kept in the same plane



2.3. Design and Testing of Typical Antenna

Example : Turnstile Antenna for CubeSat

- Combination of 4 antenna elements
 (= a set of 2 dipole antennas = cross-dipole)
- □ Single element's properties are same as dipole case
- □ Input RF power is fed to 4 antenna elements by changing the phase every 90 degrees (0, +90, +180, +270)
- Omnidirectional radiation pattern
 Circular polarization: Electric field rotates in the direction of propagation
 - **RHCP:** Right Hand Circular Polarization
 - □ LHCP: Left Hand Circular Polarization

Circular polarization is much more robust for space telecommunication than linear polarization

GS antenna's RHCP or LHCP must be match the satellite's condition



2.3. Design and Testing of Typical Antenna

Example : Patch Antenna for CubeSat

- Plate style
 - (good for keeping within the envelope requirement)
- PCB and copper plate sizes depends on frequency (In the case of over 1.2GHz-band, the entire antenna size is suitable for typical 1U CubeSat Surface [10x10cm])
- Omnidirectional radiation pattern Circular polarization: Electric field rotates in the direction of propagation depending on the position of a feed point and a

depending on the position of a **feed point** and a **paired notch**

- **RHCP:** Right Hand Circular Polarization
- LHCP: Left Hand Circular Polarization

Circular polarization is much more robust for space telecommunication than linear polarization

GS Antenna's RHCP or LHCP must be match the satellite's condition





2.3. Design and Testing of Typical Antenna

Impedance Matching & Antenna Tuning

- Impedance: the degree of signal flow (alternative current) easy or not
 ex) 50[Ω] or 75 [Ω] : standard RF component
- Signal reflection happens at different impedance points
- For effective signal transmission, this reflection should be as small as possible
- ❑ VSWR (Voltage Standing Wave Ratio) At no reflection case → VSRW = 1.0 At realistic case : VSWR < 2.0 (especially for transmit antenna)
- ❑ VSWR tuning can be made by cutting the antenna element or by adjusting circuit constant (discreet parts: R or C or L) with measuring by a network analyzer



2.4. Examples of Antenna Products

CubeSat Kit Example

Deployable antenna integrated structure

- Frequency and data rate are standardized for typical amateur (VHF/UHF) CubeSat missions
- Users do not need to consider detailed design in the case of the combination of recommended TX/RX and Antenna
- □ Example (right Images)
 - □ ISISPASE: CubeSat Antenna System for 1U/3U
 - GomSpace: NanoCom ANT430
- □ If TX/RX or antenna were to be newly developed, users should verify the combined performance of antennas and TX/RX hardware

Structure + Solar Panel + Antenna = Standard Kit

(If user's interest is NOT focused on these developments, this kind of standard CubeSat Kit is very useful to reduce the project work load)









KiboCUBE Academy

3.1. Introduction of Radio Hardware (TX/RX)

Radio Hardware

□ Fundamental configuration (major functions)

- □ RF power amplify
- □ Frequency conversion (upconvert/downconvert)
- Modulation / demodulation (D/A conversion or A/D conversion)
- Baseband signal processing

Make or Buy

- □ Make: for mature projects
 - □ Wide knowledge is required (digital, analog and RF)
 - hard to develop for first project of a
 - University
 - □ Higher frequency, higher difficulty
 - Amateur radio operators (HAM)'s support
 - Alternative option: SDR (Software Defined Radio) supported by FPGA
- **Buy**: CubeSat kit often includes packaged TX/RX



3.2. Major Characteristics of Radio Hardware (TX/RX)

TX (Transmitter)

- Transmit power [W]
- □ Frequency stability [Hz/degC] or [ppm]
- **Bitrate** [bps]
- Modulation
- **D** Power efficiency [%]
 - (= transmit power / power consumption of module)

RX (Receiver)

- Sensitivity [dBuV] or [dBm]
- □ RSSI (received signal strength intensity) [dBm/V]
- Modulation

Common

- □ Operational temperature range [degC]
- □ Operational voltage [V] ex) +5.0V or +3.3V
- Power consumption [W]
- Dimensions [mm]
- Mass [g]

	Items	Example Product Values
ТХ	Frequency Range	435 ~ 438 [MHz]
ТХ	Transmit Power	CW: 0.1 [W], FM: 0.8[W]
ТХ	Frequency Stability	± 2.5 [ppm] (-30 ~ 60 [degC])
ТХ	Modulation	AFSK, GMSK
RX	Frequency Range	145 ~ 146 [MHz]
RX	Sensitivity	-13 [dBu/V] (-120 [dBm])
	Power Consumption	0.2 ~ 0.5 [W] (@Standby/CW) 2.5 [W] (@FM Transmitting)
	Dimensions	80 x 50 x 12 [mm]
	Mass	50 [g]



3.3. Design and Testing of Radio Hardware (TX/RX)

Key Design Points of RF Components

- □ Circuit **impedance** consideration:
 - Dimensions and shape of signal line pattern on PCB should be designed with consideration for properties corresponding to wavelength of RF Signal
- □ Shielding to improve S/N (signal-to-noise Ratio)
- □ Connector selection corresponding to the frequency
- Signal line **filtering** and power line **decoupling** to reduce noise effect from/to system

Testing Key Points of RF Components

- □ TX frequency stability (corresponding to operational temp)
- □ TX power stability (corresponding to operational temp)
- □ RX frequency stability (corresponding to operational temp)
- □ RX sensitivity stability (corresponding to operational temp)
- □ TX/RX combined operation (TX's effect to RX: EMC/EMI)



Each parts' **distance**, **position**, and **pattern shape** are important to realize suitable RF circuit performance



TX/RX Hardware Cross-section View Compartment of each function (digital, analog, rf front-end) to minimize electromagnetic interference each other





3.4. Examples of Radio Hardware (TX/RX)

Tele-Comm. Hardware Products Example

- Japanese WebShop example <u>makesat.com/Communication</u> © 2016 Infosteller <u>space-for-space.com/Communication</u> @2021 Space BD
- There are various products for different frequencies and communication rates, the user should select a suitable product for their CubeSat mission
- If the user's interest is NOT focused on these developments, this kind of standard CubeSat kit is very useful to reduce the project work load











KiboCUBE Academy

4.1. Introduction of Signal Processing and Software Functions



25

4.2. Major Characteristics of Signal Processing and Software Functions

Modulation Methods

- Modulation: Data conversion of digital to analog signals
 Demodulation: Data conversion analog to digital signals
- □ CW: Continuous wave = no modulation
 (just ON/OFF carrier RF) → simple, legacy, but robust
- FM + FSK or AFSK (Audio Frequency Shift Keying)
 is popular for university missions to communicate digital data
 based on the protocol (AX.25 in amateur)
 1200bps is nominal usage
- GMSK 9600bps or more high speed
- To achieve higher bitrate communication, advanced modulation methods must be considered

Term	Description
AM	Amplitude Modulation
FM	Frequency Modulation
РМ	Phase Modulation
PQM	Phase Quadrature Modulation
РСМ	Pulsed-Code Modulation
РРМ	Pulse Position Modulation
ASK	Amplitude Shift Keying
FSK	Frequency Shift Keying
PSK	Phase Shift Keying
BPSK	Bi-Phase Shift Keying
QPSK	Quadrature Phase Shift Keying
GMSK	Gaussian Filtered Minimum Shift Keying



4.3. Design and Testing of Signal Processing

Digital Signal Packet Design

Open Systems Interconnection model (OSI model)

- Developed by the International Organization for Standardization (ISO) to classify and clarify the roles of the many protocols used in computer networks
- Defines communication functions (communication protocols) in seven layers
- A conceptual model for understanding the overall architecture of internet communications

Digital packet design of TeleCom. for CubeSats is also based on this OSI model concept

Layer Name	Description
7. Application	Specific services (Ex. E-mail, HTTP, FTP)
6. Presentation	Data presentation style (Ex. ASCII Code)
5. Session	Starting and terminating management, Reconnection management
4. Transport	End-to-end communication management (error correction, retransmission control)
3. Network	Decides which physical path the data will take (rooting)
2. Data link	Defines the format of data on the network
1.Physical	Physical (electromagnetic) signal connection (Ex. wired: RS-232, 10BASE-T, wireless: wifi)



4.3. Design and Testing of Signal Processing

Data Packet Protocol Example: Amateur AX.25

- A protocol originally derived from layer 2 of the X.25 protocol suite and designed for use by amateur radio operators in 1984
- AX.25 v2.0 and later occupies the data link layer, the second layer of the OSI model (Reference: AX.25 Version 2.0, By WB4JFI)

Data Packet Protocol Example: CSP

- **CSP:** CubeSat Space Protocol
- Developed by a group of students from Aalborg University in 2008, and further developed for the AAUSAT3 CubeSat mission (2013 Launch) (Reference: <u>GitHub, GomSpace/Space Inventor</u>)

Data Packet Protocol Example: CCSDS

- CCSDS: Consultative Committee for Space Data Systems, Space Packet Protocol
- **Error correction** by RS(Reed-Solomon) Code
- Most popular for commercial space mission (Reference: <u>Space Packet Protocol (CCSDS 133.0-B-2)</u>

For detailed descriptions, refer to official documents

If you want many ground stations to receive your CubeSat's telemetry, you need to consider these protocols

AX.25 (V.2.0)					<pre>※FCS (Frame Check Sequence) = Error Detection Method</pre>					
FI (0x	Flag Address (0x7E) (CallSign)		С	Control Info (User Data		ata)	FCS	Flag (0x7E)		
1 B	yte	14 or 28 I	Bytes	1 or	2 Bytes	Max. 256 E	Bytes	2 Bytes	1 Byte	
CSP (V.2.0)										
	Heade				6 Byte					
 Priority (2 bits) Destination (14 bits) Source (14 bits) Destination Port (6 bits) Source Port (6 bits) 			its)	 Reserved (2 bits) HMAC (1 bit) XTEA (1 bit) RDP (1 bit) CRC (1 bit) 		Ма	User Data Max. 65,535 Bytes			
CCSDS Space Packet				CCSDS			RS Code (Reed-Solomon) ※Error Correction			
		M_PD Heade	DU He er 2 I	Header 2 Bytes	Max. 209 Bytes					
Sync Maker 4 Bytes		Header	2 Bytes		CCSDS Packet 215 Bytes				rtes	
		4 Bytes	VCDL	VCDU Data 217 Bytes			Method			
		Virtual Channel Data Unit (VCDU) 223 Bytes						JZ DYIES		

28







KiboCUBE Academy

5.1. End-to-End Test of Tele-Comm.

GS Antenna Facilities Implementation Example

(Case: Teikyo University, Utsunomiya Campus)

- To perform end-to-end tests of Tele-Comm. subsystem, using actual GS facilities is very important
 - □ Before license : Wired test by coaxial cable between satellite system and GS facilities
 - □ After license : RF link test by flight model antenna.

To emulate loss caused by long distance from satellite, insert the **ATT**s (**attenuators**) to make the value equivalent to the actual operation





5.1. End-to-End Test of Tele-Comm.

GS Antenna Facilities Implementation Example (Case: Teikyo University, Utsunomiya Campus)

GS Antenna Facility Development Consideration Points

Location (skyline measurement) UHF/VHF antenna: few obstacles because of tall tower support

Parabola antenna: see the measurement result \rightarrow

□ Surrounding RF noise source

thorough survey is needed before starting construction

□ Roof load capacity

Reinforcement structure of roof may be required (depend on the local law or regulation of building)

Lighting rod

If the candidate place to build a new antenna tower is outside the valid range of the existing lighting rod, a new lighting rod may be required

(depends on the local law or regulation of building)



Parabola Antenna Facility was established in 2018 under Yagi Antenna existing condition, therefore we checked and evaluated the **skyline** (elevation lower limit) by camera image attached on parabola frame



TeikyoU GS Antenna

Φ2.0m Parabola Antenna (S/C/X-band Downlink) 2 stack **Yagi-Antenna** (L-band Uplink)



5.1. End-to-End Test of Tele-Comm.

Long-Distance Communication Test

- Example case: Teikyo University
 - (Utsunomiya, Tochigi, Japan)
 - ~10km distance from Mt. Haguro (just North direction from the Teikyo University Utsunomiya Campus)
 - □ Almost Az = ~0 deg, EL = ~2 deg
 - Roofed viewing deck is useful setting facilities including antenna panel, satellite components, and portable battery
- Telemetry / command check by using actual GS antenna
- Also useful to perform GS antenna pointing accuracy check
- This place is popular and famous for local amateur radio people of Tochigi



5.1. End-to-End Test of Tele-Comm.

Tele-Comm. Performance Tests Example

- **EMC** (Electromagnetic Compatibility)
 - Identifying internal noise source
 - Especially from EPS (large current root, high-frequency switching regulator)
 - □ System frequency management is important (CPU Clock, switching frequency of regulator)
 - If the noise effects are too bad to realize normal communication, take action such as adding electromagnetic shielding (Al or Cu sheet metal spot casing) to the noise source
- **EMI** (Electromagnetic Interference)
 - □ Effect satellite from inside to outside (equipment on launcher or ISS)
 - Requirements described launcher's ICD (Interface control Documents) The CubeSat project must comply with the frequency and EIRP requirements
 - If the EMI effects are too bad to comply the requirement, the CubeSat project should consider changing the TX hardware or redesigning of TX antenna
 - This may be avoided by conforming to a cold launch with 3 inhibits (depends on the launch condition)

Operational Test

- To confirm the operational capability of downlink duty or power mode or bitrate as entire system level including power budget feasibility
- If the bad condition or trouble was found through the operational test, consider adjustable parameters (duty ratio of TX, power mode)



5.2. Key Concepts and Checkpoints of Tele-Comm. Integration for Mission Success

Table. Checkpoints of Tele-Comm. Design and Implementation Summary

#	Category	Торіс	
Tele-Comm01	Antenna	Concept Monopole Dipole Turnstile Patch	
Tele-Comm02	RF Hardware	 Concept Design Configuration Make or Buy Decision 	
Tele-Comm03	Modulation	Concept Modulation Methods	
Tele-Comm04	Protocol Data Processing	 □ Concept □ OSI Model □ AX.25 □ CubeSat Data Protocol □ CCSDS Space Data Packet 	
Tele-Comm05	GS	 Antenna Location Long Distance Communication Test Wired Test RF Link Test 	
Tele-Comm06	End-to-End	EMC EMI Operational Test	









6. Conclusion

KiboCUBE Academy

6. Conclusion

- Tele-Comm is a very long-term activity (satellite projects start from frequency allocation, then satellite projects finish with sending the RF transmitting termination command)
- Antennas are the most critical item for successful communication of a CubeSat. Select a suitable design for system architecture and project maturity
- As for RF Hardware, Make-or-Buy-decision is important Select a suitable product for requirements (bitrate and protocol) and project maturity
- Tele-Comm subsystem end-to-end tests
 - **EMC** must be confirmed to realize better communication (In the worst case, the communication quality is not as good as the link analysis)
 - **EMI** must be confirmed to comply with the launcher's or ISS's requirements (This may be avoided by conforming to a cold launch with 3 inhibits)
 - □**Operational tests & long distance tests** are useful to find the potential problems (For mission success, CubeSat projects should perform these tests as long as possible)





Thank you very much.

[Disclaimer]

The views and opinions expressed in this presentation are those of the authors and do not necessarily reflect those of the United Nations.

KiboCUBE Academy