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# International GNSS Service (IGS): Orbit Dynamics, Modeling and Timing IGS Advances and Future Applications

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#### Orbits and clocks: defining the reference frame





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# Photons and Radiation Pressure



**Radiation Pressure** 

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#### The Maths.....

$$E = mc^2$$

# For Photons:

 $\rho = E/c$ 

# Momentum = Energy/speed of light



#### **Radiation Pressure**

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#### Solar radiation pressure



#### **Thermal Re-radiation forces**

#### How big are these forces? What effect do they have?

Resultant force from Thermal emissions

Resultant force from solar photons



We develop a detailed structural computer model of the spacecraft



Optical and thermal properties

Spacecraft model represented in the SV body frame



An array of flux origin points is created Dependent on the incoming flux direction

#### Rays are generated from each of these points



# The intersection between each ray and the spacecraft is computed



The material properties of the surface are known at each of these points



The surface normal is calculated at each of these points for planar or **curved** surfaces



New rays are generated for each reflection and the intersection and reflection step repeated: Secondary intersections



The acceleration data for all rays (primary and secondary) is collected:

Resultant acceleration computed for that radiation source direction

# Process is repeated for other incoming flux directions

















## Thermal modelling:

Anisotropic thermal emission from spacecraft results in a net acceleration



# Multilayer Insulation (MLI)

- Pixel array algorithm determines Energy balance: insolation of MLI Incoming radiation (W absorbed • 'Effective emissivity' ( $\mathcal{E}_{eff}$ )  $(\mathbf{H})$ Thermally parameter governs heat stabilised Emitted transfer to bus To s/c bus,  $T_{\rm sc}$ radiation • MLI blackened,  $\alpha$ =0.94  $\Rightarrow$  large thermal force
  - $T_{MLI}^{4} = \frac{\alpha W \cos \theta + \varepsilon_{eff} \sigma T_{sc}^{4}}{\sigma (\varepsilon_{MLI} + \varepsilon_{eff})}$



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### **Solar Panel Thermal Analysis**

- Steady state and transient models (during eclipse) developed to yield temperatures and forces
- Input data : thicknesses and conductivities of panel composite layers, surface emissivities and absorptivities, power draw
- Model verification by comparison with telemetered surface temperatures





#### **Antenna Thrust**

- Recoil force on satellite due to transmitted signals
- Systematic and observable effect
- Requires knowledge of power transmission of satellites







Planetary Radiation Pressure (PRP) models using space based observations of emission and reflectance.



\*Earth textures courtesy of NASA Blue Marble: Next Generation. Earth radiation data courtesy of CERES and MODIS.



How well do these ideas work? Do they make any difference?: Predicting a GPS satellite orbit over a 173,000 km trajectory



Along-track orbit prediction errors over 12 hours for one GPS IIR satellite with different photon-based force models







#### IGS experiments, analysis, standards

- This research leads to IGS standards applied by all analysis groups within the organization. Ideas are tested by large-scale data processing experiments spanning many years and huge networks of data. The recent REPRO2 exercise re-computed orbit, clock, station positions and earth orientation parameters using over twenty years of daily data from the entire network. Such operations give insight into system and planet scale behaviors.
- Earth radiation forcing and antenna thrust methods changed IGS orbit accuracy from 5 cm to 2.5 cm (radial)



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# What do we need to do this, and to push the envelope of what is possible:

- SV mass and mass history
- Structural details (primarily surface geometry)
- Material types (absorptivity, reflectivity, specularity)
- Specific thermal information (MLI characteristics, power draw, thermal emissions, solar panel construction)
- Satellite attitude (both eclipsing and non-eclipsing, nonnominal attitude, yaw flips, noon day and midnight turns)
- Satellite phase centre (phase centre offset, phase centre variations)
- Laser retro-reflector array position, corner cube phase centre



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#### Why strive for cm accuracy orbits and clocks?



Velocity map courtesy of Mike Helflin, NASA JPL



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#### Satellite Altimetry: determining the satellite position by GPS



#### Global sea level rise measured by satellite altimetry





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## Brief philosophical musings.....

- GNSS has been a world-changing technological advance for humanity
- It will feature in history fire, language, the wheel, farming, electricity, the steam engine, radio, the Internet, e-mail, space technology, satellite navigation.....
- Much as Faraday could not have foreseen the Internet, it is difficult to predict what will be feasible in the future
- What is clear is that the work of scientists, engineers, policy makers and commerical companies drives forward what is possible – it is a privilege for all of us to be involved in this great endeavour





## Conclusions

- <u>IGS research in modelling satellite orbit dynamics is</u> pushing the frontiers of what is possible
- A <u>central problem</u> is dealing with *radiation pressure*
- We have powerful, proven tools ready to apply to GNSS
- The <u>IGS seeks the support of the ICG</u> to gain access to SV structural data for all GNSS
- To paraphrase Richard Feynmann there is room at the top and the IGS is poised to help make it happen