

ICG WG-B Achievements on Interoperable GNSS Space Service Volume (SSV)

November, 2016
Sochi, Russian Federation



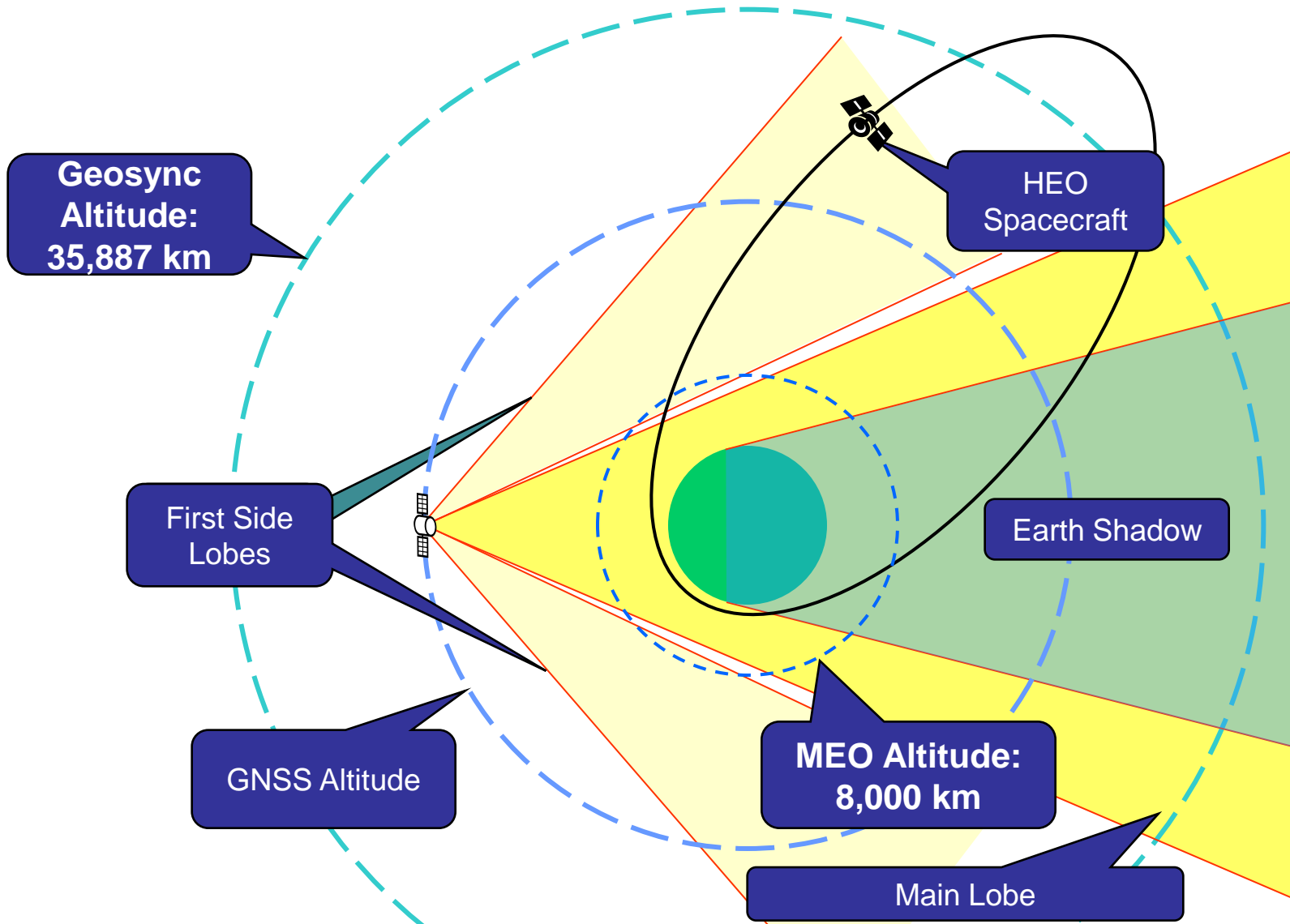
International Committee on
Global Navigation Satellite Systems

ICG WG-B Action Group on SSV

- Action group on SSV was formed within WG-B in order to
 - Establish an Interoperable GNSS Space Service Volume (SSV)
 - Promote the relevance of SSV for users and to the service providers
 - Identify SSV support of every service provider for the benefit of users and receiver manufacturers
 - Harmonize and deepen the mutual understanding on SSV
 - Establish an SSV booklet
- Work of the Action Group is supported by all GNSS service providers
- Special thank to all Space Agencies very actively supporting the work through simulations and SSV Booklet preparation which is close to completion



Reception Geometry for GNSS Signals in Space



The Interoperable GNSS SSV

- Benefits to User - Performance Aspects
 - Nearly continuous on-board Position, Velocity and Time (PVT) capability
 - Interoperable GNSS SSV allows development of new Precise Orbit Determination concepts with high accuracy
- Benefits to User – Operational Aspects
 - New operations concepts with reduced Ground interaction
 - Increase of on-board autonomy
 - Increase of robustness of spacecraft navigation and operations resilience
- Benefits to User - Technology Aspects
 - Enabler for new mission and service concepts
 - Development of GNSS Receiver core technology, applicable for a variety of missions
 - Capability to provide satellite orbit, attitude and time with one sensor

Space Service Volume Template

- Action group agreed on common template to specify the SSV support of every system.
- Template includes
 - For every Open Service signal minimum user received power at max. off-boresight angle at GEO altitude
 - Ranging accuracy at max off-boresight angle at GEO altitude
 - Availability for 1 and 4 signals incl. max. outage period
- Template information can be easily scaled for particular user missions and receiver characteristics

SSV Template of all Service Providers

- All Service Providers filled the SSV template (Details see backup slides)

GPS

Parameters	Value	
User Range Error	0.8 meters	
Signal Center Frequency		
L1 C/A, L1C	1575.42 MHz	
L2C	1227.60 MHz	
L5 (I5 or Q5)	1176.45 MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L1 C/A	-184.0 dBW	23.5 deg
L1C	-182.5 dBW	23.5 deg
L2C	-183.0 dBW	26 deg
L5 (I5 or Q5)	-182.0 dBW	26 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
L1	100%	>97%
L2, L5	100%	100%
Upper Space Service Volume (HEO/GEO)	At least 1 signal	4 or more signals
L1	>80%	>1%
L2, L5	>92%	>6.5%

GLONASS

Parameters	Value	
User Range Error	1.4 meters	
Signal Center Frequency		
L1	1605.375 MHz	
L2	1248.625 MHz	
L3	1201 MHz	
Minimum Received Civilian Signal Power (GEO)	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L1	-185 dBW	20 deg
L2	-184.4 dBW	28 deg
L3	-184 dBW	28 deg
Signal availability		
MEO at 8000 km	At least 1 signal	4 or more signals
L1	59.1%	64%
L2, L3	100%	66%
Upper Space Service Volume (HEO/GEO)	At least 1 signal	4 or more signals
L1	70%	2.7%
L2, L3	100%	29%

SSV Template of all Service Providers

- All Service Providers filled the SSV template (Details see backup slides)

Galileo

Parameters	Typical Characteristics of Nominal GSAT02xx Satellites	
User Range Error	1.1 meters	
Signal Center Frequency		
E1B/C	1575.42 MHz	
E6B/C	1278.75 MHz	
E5b	1206.45 MHz	
E5 AltBOC	1191.795 MHz	
E5a	1176.45 MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
E1B/C	-182.5 dBW	20.5 deg
E6B/C	-182.5 dBW	21.5 deg
E5b	-182.5 dBW	22.5 deg
E5 AltBOC	-182.5 dBW	23.5 deg
E5a	-182.5 dBW	23.5 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
E1B/C	100%	99%
E6B/C	100%	100%
E5b	100%	100%
E5a or E5 AltBOC	100%	100%
Upper Space Service Volume (GEO/HEO)	At least 1 signal	4 or more signals
E1B/C	≥ 64%	0%
E6B/C	≥ 72%	0%
E5b	≥ 80%	0%
E5a or E5 AltBOC	≥ 86%	0%

BDS

Parameters	Value	
User Range Error	2.5 meters	
Signal Center Frequency		
B1	1575.42 MHz	
B2	1191.795MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
B1 (MEO)	-184.2 dBW	25 deg
B1 (GEO/IGSO)	-185.9 dBW	19 deg
B2 (MEO)	-182.8 dBW	28 deg
B2(GEO/IGSO)	-184.4 dBW	22 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
B1	100%	100%
B2	100%	100%
Upper Space Service Volume (GEO/HEO)	At least 1 signal	4 or more signals
B1	97.4%	24.1%
B2	99.9%	45.4%

SSV Template of all Service Providers

- All Service Providers filled the SSV template (Details see backup slides)

QZSS

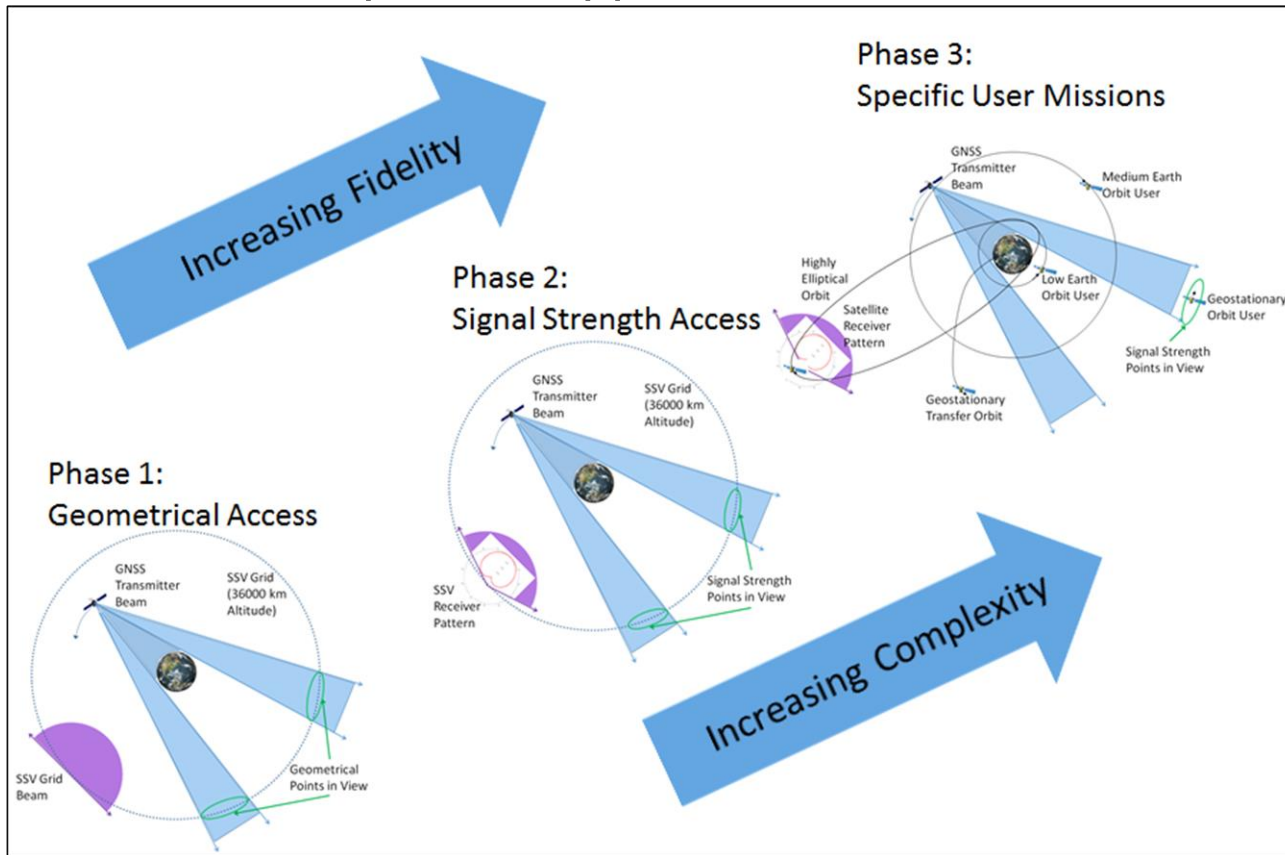
Parameters	Value	
User Range Error	2.6 meters (95%)	
Signal Center Frequency		
L1 C/A	1575.42 MHz	
L1C	1575.42 MHz	
L2 C	1227.60 MHz	
L5 (I5 or Q5)	1176.42 MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L1 C/A	-185.3 dBW	22 deg
L1C	-185.3 dBW	22 deg
L2 C	-188.7 dBW	24 deg
L5 (I5 or Q5)	-180.7 dBW	24 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
L1	100%	N/A
L2, L5	100%	N/A
Upper Space Service Volume (HEO/HEO)	At least 1 signal	4 or more signals
L1	≥ 54%	N/A
L2, L5	≥ 54%	N/A

IRNSS

Parameters	Value	
User Range Error	2.11 meters	
Signal Center Frequency		
L5	1176.42 MHz	
Minimum Received Civilian Signal Power, in dBW	0 dBi RCP antenna at GEO	Reference Off-Boresite Angle
L5	-184.54	16 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
L5	98.00%	51.40%
Upper Space Service Volume (HEO/HEO)	At least 1 signal	4 or more signals
L5	36.9%	0.6%

SSV Simulations – 3 Phases

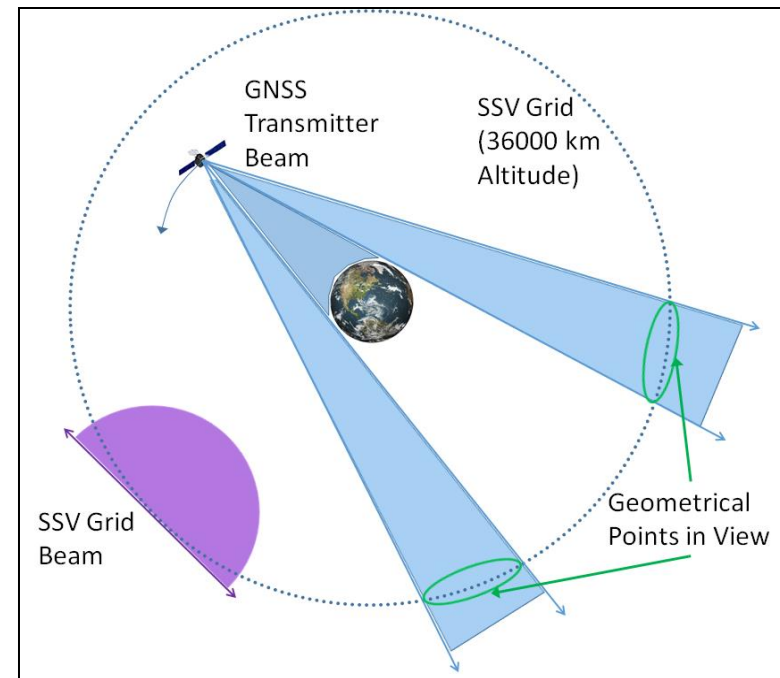
- SSV simulations conducted by the group involving 5 independent simulation tools in a phased approach



Phase 1 Simulation – Objectives and Setup

- Objectives
 - Ensure unambiguous interpretation of SSV template parameters
 - Identify and agree on details for performance evaluation metrics
 - Alignment of simulation tools and identification of error sources

- Setup
 - Availability purely geometry based, no assumptions on user antenna
 - All systems with nominal constellations
 - Signals: L1 and L5
 - User locations: Equal area-based grid at GEO and MEO (8000 km) altitude



Phase 1 Simulation Results - GEO

L1	BDS	Galileo	GLONASS	GPS	IRNSS	QZSS	All
1 Satellite System Availability (%)	97.4	78.5	59.1	90.5	-----	26.7	99.9
4 Satellite System Availability (%)	24.1	1.2	0.5	4.8	-----	0.8	94.4
1 Satellite Maximum Outage (minutes)	45	98	134	111	-----	(SD)	39
4 Satellite Maximum Outage (minutes)	(SD)	(SD)	(SD)	(SD)	-----	(SD)	97

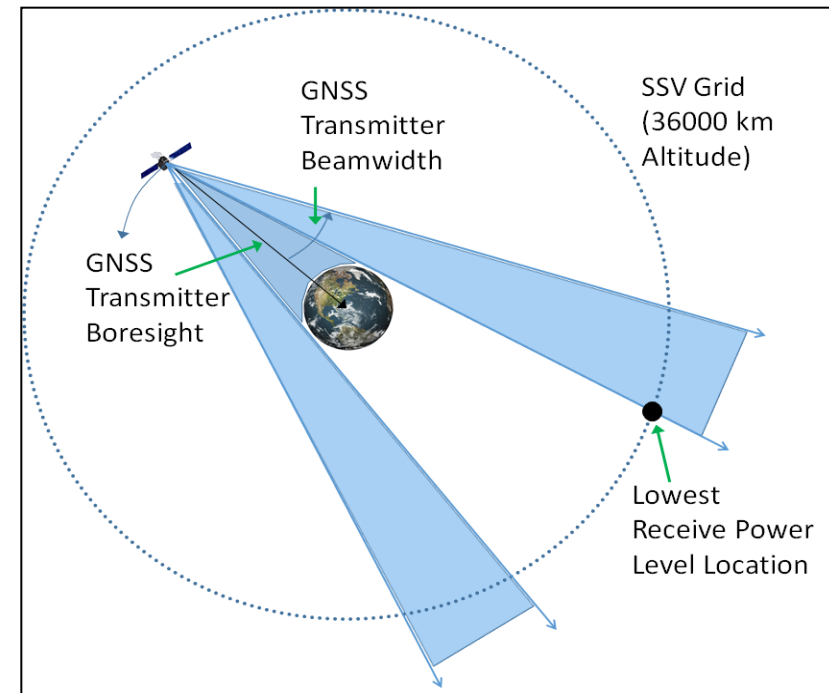
- No single constellation can provide high availability of 4 ranging signals on its own, but all constellations together can do so!

(SD) Scenario Duration

Phase 2 Simulation – Objectives and Setup

- Objectives
 - Introduce user antenna gain characteristics and receiver operational thresholds for availability evaluation
 - Bridging towards Phase 3 while not affecting the validated orbit propagation of Phase 1

- Setup changes wrt. Phase 1
 - Introduction of antenna gain figures
 - Availability evaluation at different receiver operational thresholds
 - Focus on GEO altitude users only

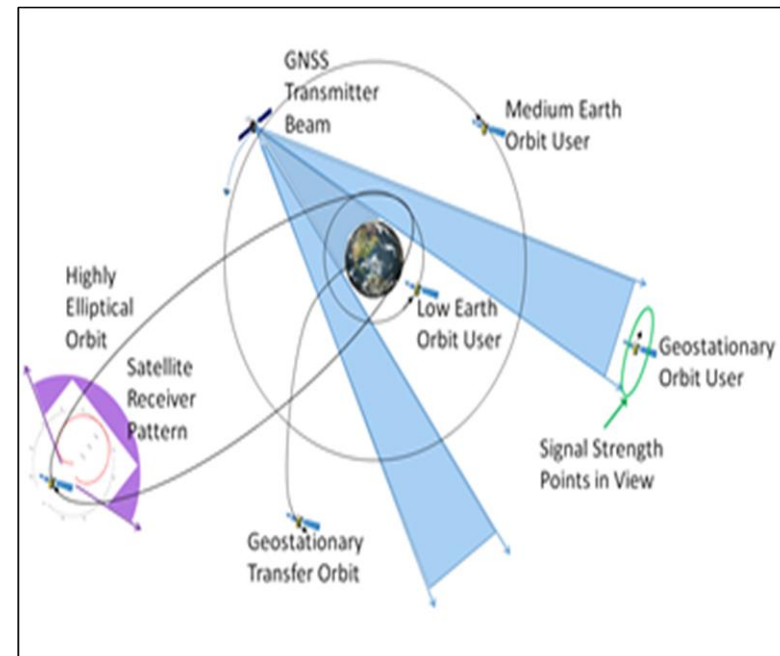


Phase 2 Simulation Results- GEO - L1

	BDS	Galileo	GLONASS	GPS	QZSS	All
20 dBHz						
1+ Availability	69%	78%	0%	90%	0%	99%
Max 1+ Outage	70	98	(SD)	111	(SD)	49 min
4+ Availability	0%	1%	0%	4%	0%	62%
Max 4+ Outage	(SD)	(SD)	(SD)	(SD)	(SD)	223 min
15 dBHz						
1+ Availability	97%	78%	59%	90%	26%	99%
Max 1+ Outage	45	98	134	111	(SD)	39 min
4+ Availability	24%	1%	0%	4%	0%	94%
Max 4+ Outage	(SD)	(SD)	(SD)	(SD)	(SD)	97 min

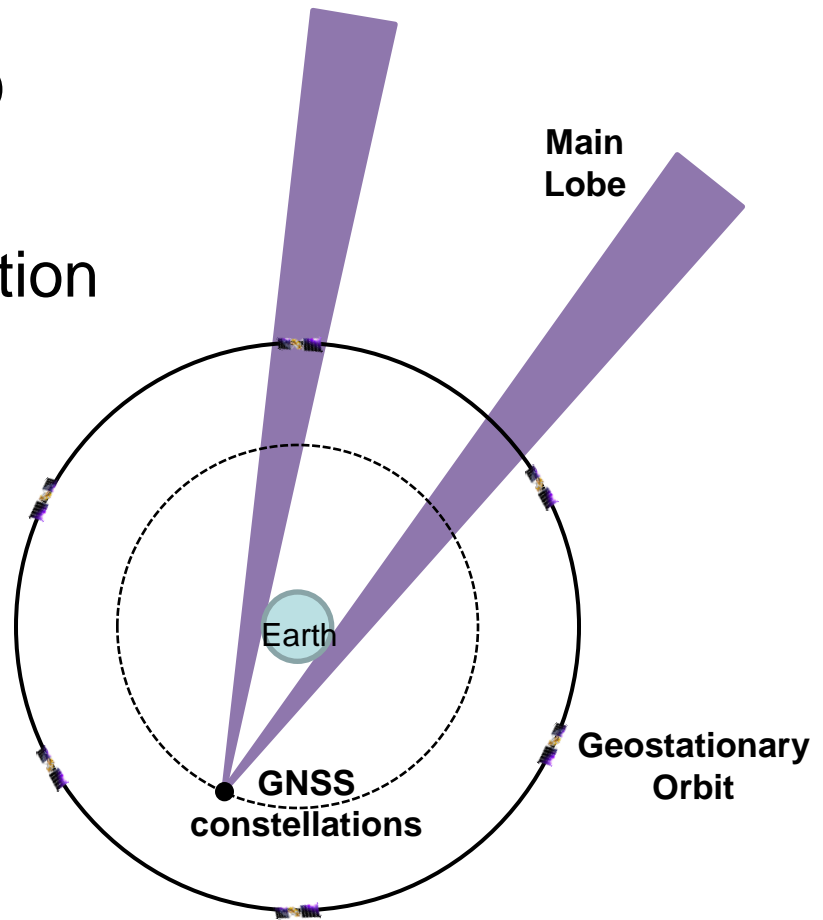
Phase 3 Simulation – Objectives and Setup

- Objectives
 - Simulation of availability and outage periods for realistic, representative missions
- Setup changes wrt. Phase 2
 - User trajectory instead of user grid
 - GEO
 - Scientific HEO
 - Lunar Mission
 - Real user antenna gain patterns
 - Consideration of satellite attitude and antenna location
- Phase 3 scenarios are defined and will be simulated in the very near future



Phase 3 GEO Mission Setup

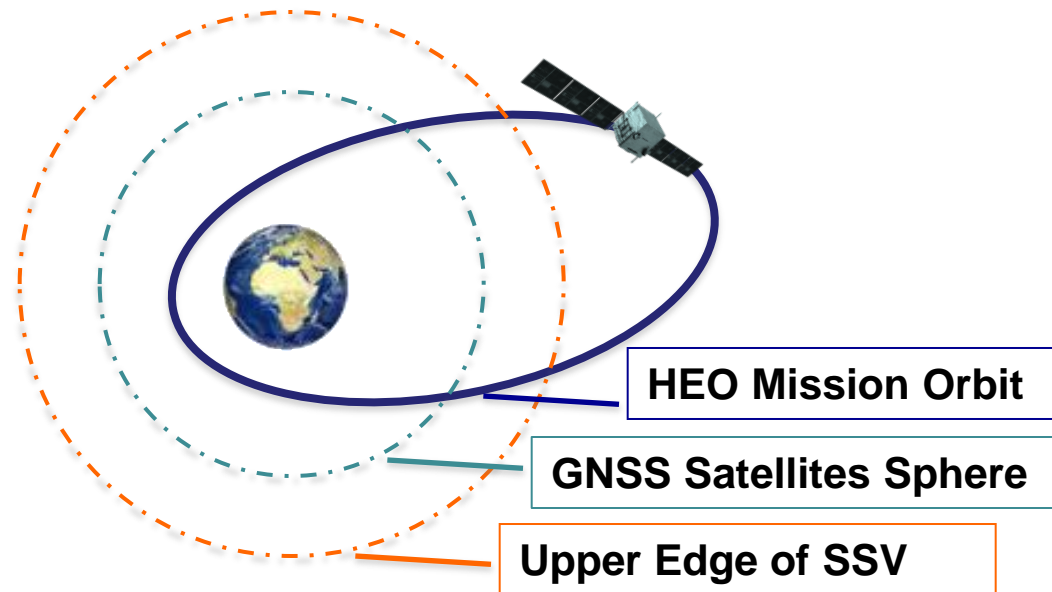
- GEO user will be placed in six different locations in GEO
- Satellite is Nadir pointing
- Antenna is also in Nadir direction



Phase 3 Scientific HEO Mission Setup

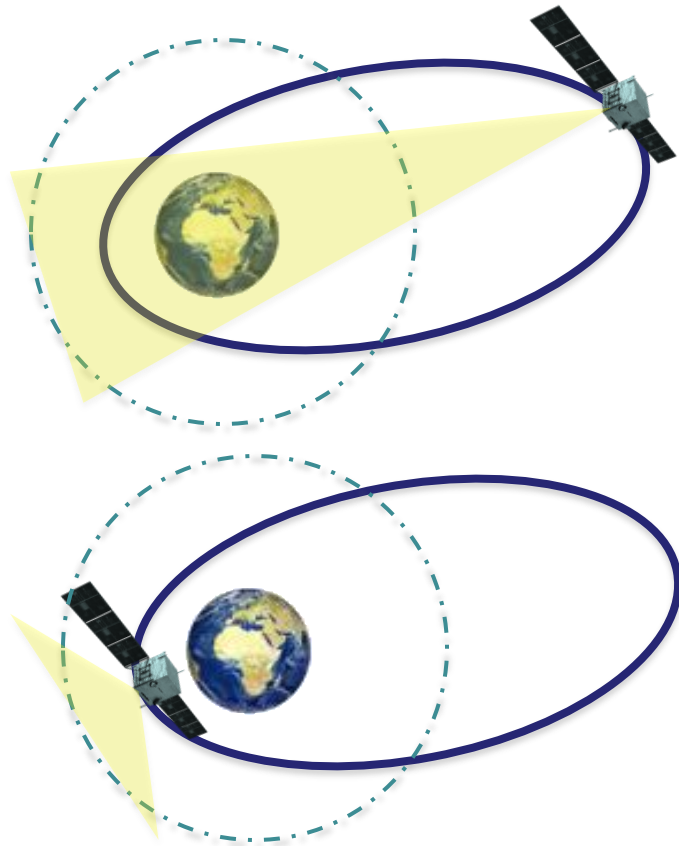
Mission: SSV-HV1

Perigee	500 km
Apogee	65000 km
Inclination	63.4°



Phase 3 Scientific HEO Mission Setup

Mission: SSV-HV1



- GNSS antennas: NADIR and Zenith facing
- During the apogee period: Reception through NADIR pointing antenna
- During the perigee period: Reception through Zenith pointing antenna

Phase 3 Lunar Mission Setup: Exploration Mission 1 (EM-1)

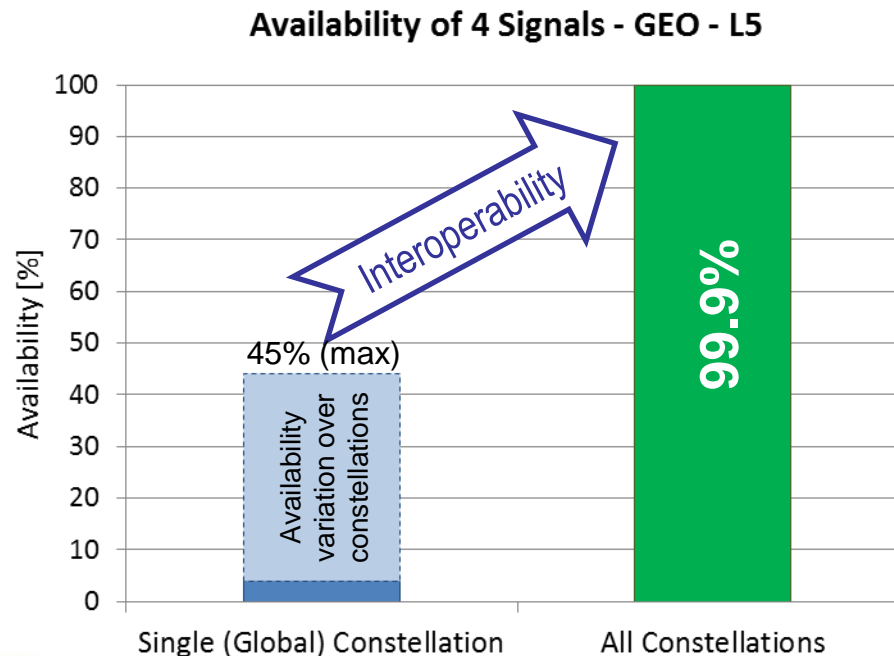
Basic Info:

Mission	Exploration Mission 1 (EM-1)
Description	Free-return lunar trajectory with optional lunar orbit phase
Perigee	391 km
Apogee	378 021 km
Inclination	28.5°
Duration	10 d
Attitude profile	TBD (simplification: nadir-pointing)
Receive antenna	High-gain



Findings

- High altitude space users particularly benefit from an interoperable GNSS SSV as
 - No single constellation can provide high availability
 - Interoperable GNSS SSV can ensure high availability



Findings

- Work of the SSV Action group
 - Demonstrates the benefits and importance of the Interoperable GNSS SSV
 - Generates awareness on the relevance of GNSS interoperability
 - Fosters the cooperation between Service Providers and Space Agencies on SSV
 - Establishes a commonly shared unambiguous interpretation of the SSV template and its parameters
 - Provides relevant material for space receiver manufacturers

WG-B Sessions During ICG11

ICG WG-B sessions on SSV during ICG11 in room
“Camelia”

- Tuesday, 08/11, 14:00 – 18:00
- Wednesday, 09/11, 11:20 – 15:40

BACKUP MATERIAL



SSV Booklet

- SSV booklet under finalisation by the Action Group covers
 - SSV general aspects incl. benefits for users
 - Single point of entry regarding SSV support from every service provider (beneficial for users and receiver developers) incl. statements on future plans
 - Simulation scenarios and performance evaluation underlining the benefits of interoperability of GNSS for space users
 - Conclusions and next steps
- SSV booklet will be made available on ICG website and interested service provider websites

Space Service Volume Characteristics

Medium Earth Orbit (MEO)

- 3,000-8000 km
- Four GNSS signals typically available; One-meter orbit accuracies
- Wide range of received GNSS signal strength
- GNSS signals received from NADIR and Zenith direction
- Signals over the limb of the Earth become increasingly important

High Earth Orbit/Geosynchronous Earth Orbit (HEO/GEO)

- 8,000-36,000 km
- Nearly all GNSS signals received over the limb of the Earth
- Users will experience periods when no satellites are available
- User will highly benefit from interoperable GNSS SSV for availability
- Will require specially designed high sensitivity receivers
- Properly designed receiver should be capable of tens to hundreds of meters accuracy with performance depending upon GNSS signal availability, receiver sensitivity and clock stability

GPS Support to SSV

Parameters	Value	
User Range Error	0.8 meters	
Signal Center Frequency		
L1 C/A, L1C	1575.42 MHz	
L2C	1227.60 MHz	
L5 (I5 or Q5)	1176.45 MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L1 C/A	-184.0 dBW	23.5 deg
L1C	-182.5 dBW	23.5 deg
L2C	-183.0 dBW	26 deg
L5 (I5 or Q5)	-182.0 dBW	26 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
L1	100%	>97%
L2, L5	100%	100%
Upper Space Service Volume (HEO/GEO)	At least 1 signal	4 or more signals
L1	>80%	>1%
L2, L5	>92%	>6.5%

U.S. Support to Ensure GNSS Interoperability for Space Users

- Performing additional flight experiments above the constellation to characterize signals in cis-Lunar space
- Developing new weak signal GPS/GNSS receivers for spacecraft in cis-Lunar space through government technology developments (e.g. NASA Goddard Navigator, NavCube) and commercial procurements
- Working with the GPS Directorate and DoD community to formally document GPS requirements and antenna patterns for space users
- Developing missions and systems to utilize GNSS signals in the SSV (e.g. MMS, GOES, Orion)
- Supporting ICG WG-B SSV initiative through sustained technical guidance, SSV booklet development & leadership of SSV analysis
- Encouraging international coordination with other GNSS constellations (e.g, Galileo, GLONASS, BDS) to **specify** interoperable SSV capabilities

GLONASS Support to SSV

Parameters	Value	
User Range Error	1.4 meters	
Signal Center Frequency		
L1	1605.375 MHz	
L2	1248.625 MHz	
L3	1201 MHz	
Minimum Received Civilian Signal Power (GEO)	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L1	-185 dBW	20 deg
L2	-184.4 dBW	28 deg
L3	-184 dBW	28 deg
Signal availability		
MEO at 8000 km	At least 1 signal	4 or more signals
L1	59.1%	64%
L2, L3	100%	66%
Upper Space Service Volume (HEO/GEO)	At least 1 signal	4 or more signals
L1	70%	2.7%
L2, L3	100%	29%

Galileo Support to SSV

Parameters	Typical Characteristics of Nominal GSAT02xx Satellites	
User Range Error	1.1 meters	
Signal Center Frequency		
E1B/C	1575.42 MHz	
E6B/C	1278.75 MHz	
E5b	1206.45 MHz	
E5 AltBOC	1191.795 MHz	
E5a	1176.45 MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
E1B/C	-182.5 dBW	20.5 deg
E6B/C	-182.5 dBW	21.5 deg
E5b	-182.5 dBW	22.5 deg
E5 AltBOC	-182.5 dBW	23.5 deg
E5a	-182.5 dBW	23.5 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
E1B/C	100%	99%
E6B/C	100%	100%
E5b	100%	100%
E5a or E5 AltBOC	100%	100%
Upper Space Service Volume (GEO/HEO)	At least 1 signal	4 or more signals
E1B/C	≥ 64%	0%
E6B/C	≥ 72%	0%
E5b	≥ 80%	0%
E5a or E5 AltBOC	≥ 86%	0%

Galileo SSV

- Galileo SSV Characteristics are provided for Galileo FOC (GSAT02xx) satellites
- The Galileo SSV parameters presented above are derived from on-ground and in-orbit measurement campaigns
- The Galileo SSV parameters do not represent a commitment from the European Commission to comply with such characteristics for the already launched or future satellites.
- Official information related to SSV characteristics of Galileo will be published through the Galileo OS Service Definition Document in the future

BDS Support to SSV

Parameters	Value	
User Range Error	2.5 meters	
Signal Center Frequency		
B1	1575.42 MHz	
B2	1191.795MHz	
Minimum Received Civilian Signal Power	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
B1 (MEO)	-184.2 dBW	25 deg
B1 (GEO/IGSO)	-185.9 dBW	19 deg
B2 (MEO)	-182.8 dBW	28 deg
B2(GEO/IGSO)	-184.4 dBW	22 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
B1	100%	100%
B2	100%	100%
Upper Space Service Volume (GEO/HEO)	At least 1 signal	4 or more signals
B1	97.4%	24.1%
B2	99.9%	45.4%

BDS Support to SSV

- SSV parameters obtained from simulation and test results of new-generation BDS satellites deployed in 2015
- URE value is from BDS_OS_PS1.0. Current system accuracy is better than published standard. And it will be further improved with the construction of global BDS
- BDS is taking actions in SSV performance characterization and specification
- Official information related to SSV characteristics will be published in the future through BDS Standard Document

QZSS Support to SSV

Parameters		Value	
User Range Error		2.6 meters (95%)	
Signal Center Frequency			
	L1 C/A	1575.42 MHz	
	L1C	1575.42 MHz	
	L2 C	1227.60 MHz	
	L5 (I5 or Q5)	1176.42 MHz	
Minimum Received Civilian Signal Power		0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
	L1 C/A	-185.3 dBW	22 deg
	L1C	-185.3 dBW	22 deg
	L2 C	-188.7 dBW	24 deg
	L5 (I5 or Q5)	-180.7 dBW	24 deg
Signal Availability			
Lower Space Service Volume (MEO)		At least 1 signal	4 or more signals
	L1	100%	N/A
	L2, L5	100%	N/A
Upper Space Service Volume (GEO/HEO)		At least 1 signal	4 or more signals
	L1	≥ 54%	N/A
	L2, L5	≥ 54%	N/A

IRNSS Support to SSV

Parameters	Value	
User Range Error	2.11 meters	
Signal Center Frequency		
L5	1176.42 MHz	
Minimum Received Civilian Signal Power, in dBW	0 dBi RCP antenna at GEO	Reference Off-Boresight Angle
L5	-184.54	16 deg
Signal Availability		
Lower Space Service Volume (MEO)	At least 1 signal	4 or more signals
L5	98.00%	51.40%
Upper Space Service Volume (HEO/GEO)	At least 1 signal	4 or more signals
L5	36.9%	0.6%

IRNSS Support to SSV

- The IRNSS is an ISRO initiative to build an independent satellite navigation system to provide PVT solutions to users over the Indian region and a region extending 1500km around India.
- All IRNSS satellites are pointing (Nadir) towards 83°E and 5°N on Earth.
- IRNSS SSV parameters do not represent a commitment. Official information related to SSV will be published in the future through the IRNSS Service Volume Document

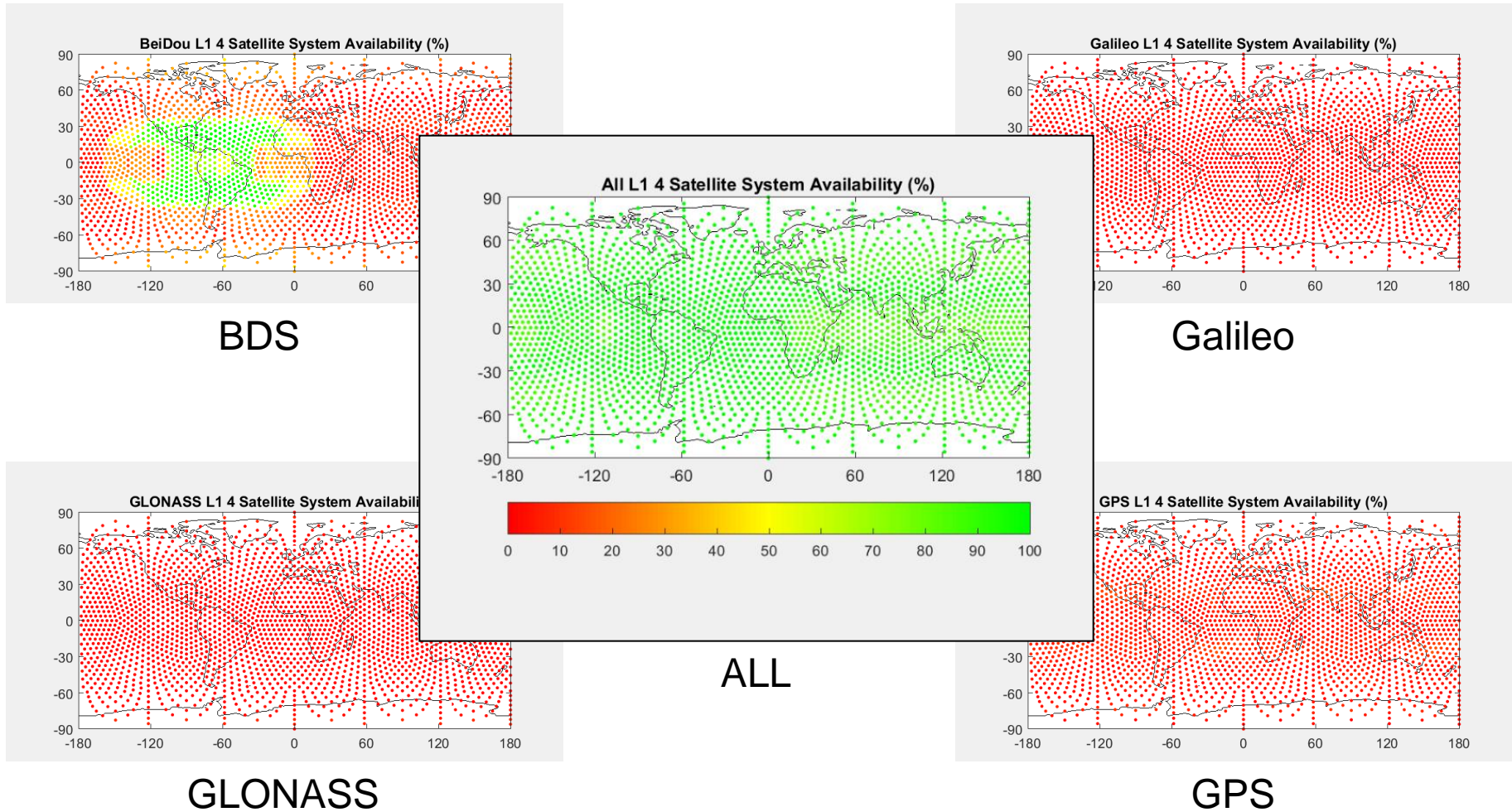
Phase 1 Simulation Results - GEO

L1	BDS	Galileo	GLONASS	GPS	IRNSS	QZSS	All
1 Satellite System Availability (%)	97.4	78.5	59.1	90.5	-----	26.7	99.9
4 Satellite System Availability (%)	24.1	1.2	0.5	4.8	-----	0.8	94.4
1 Satellite Maximum Outage (minutes)	45	98	134	111	-----	(SD)	39
4 Satellite Maximum Outage (minutes)	(SD)	(SD)	(SD)	(SD)	-----	(SD)	97

L5	BDS	Galileo	GLONASS	GPS	IRNSS	QZSS	All
1 Satellite System Availability (%)	99.9	93.4	98.3	96.9	36.9	30.5	100
4 Satellite System Availability (%)	45.4	4.2	14.5	15.6	0.6	1.5	99.9
1 Satellite Maximum Outage (minutes)	7	55	35	77	(SD)	(SD)	0
4 Satellite Maximum Outage (minutes)	644	(SD)	2252	1180	(SD)	(SD)	35

(SD) Scenario Duration

Phase 1 Simulation Results - GEO



L1 4-signal availability: individual and combined

Phase 1 Simulation Results - MEO

			BDS	Galileo	GLONASS	GPS	IRNSS	QZSS	All
L1, Omni-Directional Antenna									
1	Satellite System	Availability (%)	100	99.9	94.0	100	----	99.6	100
4	Satellite System	Availability (%)	100	95.0	41.4	99.6	----	79.4	100
1	Satellite System	Maximum Outage (minutes)	0	11	60	0	----	197	0
4	Satellite System	Maximum Outage (minutes)	0	60	345	45	----	Max SD	0
L5, Omni-Directional Antenna									
1	Satellite System	Availability (%)	100	100	100	100	98.0	99.6	100
4	Satellite System	Availability (%)	100	100	100	99.9	51.4	79.4	100
1	Satellite System	Maximum Outage (minutes)	0	0	0	0	348	197	0
4	Satellite System	Maximum Outage (minutes)	0	0	0	16	Max SD	Max SD	0

(SD) Scenario Duration

Phase 2 Simulation Results – GEO - L5

	BDS	Galileo	GLONASS	GPS	IRNSS	QZSS	All
25 dBHz							
1+ Availability	0%	0%	0%	0%	0%	0%	0%
4+ Availability	0%	0%	0%	0%	0%	0%	0%
20 dBHz							
1+ Availability	99%	93%	98%	96%	1%	30%	100%
Max 1+ Outage	7	55	35	77	(SD)	(SD)	0 min
4+ Availability	32%	4%	14%	15%	0%	1%	99%
Max 4+ Outage	644	(SD)	2252	1180	(SD)	(SD)	35 min
15 dBHz							
1+ Availability	99%	78%	98%	96%	36%	30%	100%
Max 1+ Outage	7	55	35	77	(SD)	(SD)	0 min
4+ Availability	45%	1%	14%	15%	0%	1%	99%
Max 4+ Outage	644	(SD)	2252	1180	(SD)	(SD)	35 min

Phase 3 GEO Mission Setup

Characteristic	Details
Spacecraft Orbit	<ul style="list-style-type: none"> GEO Users will be located at six locations, separated by 60 deg Format for Orbit representation: Kepler state vector - CCSDS format
Spacecraft Attitude	<ul style="list-style-type: none"> Nadir pointing spacecraft Attitude Format for Attitude representation: quaternion - CCSDS-AEM format
Receiver Antenna Pattern	High-gain
Receiver Antenna Orientation	Nadir pointing antenna
Receiver Threshold	20 dBHz

Phase 3 Lunar Mission Setup: Exploration Mission 1 (EM-1)

Characteristic	Details
Spacecraft Orbit	Outbound ephemeris in CCSDS-OEM v1 format. Optional orbit phase + return.
Spacecraft Attitude	Attitude ephemeris in CCSDS-AEM quaternion format.
Receiver Antenna Pattern	High-gain
Receiver Antenna Orientation	TBD
Receiver Threshold	TBD